



TECHNICAL NOTE

D-1393

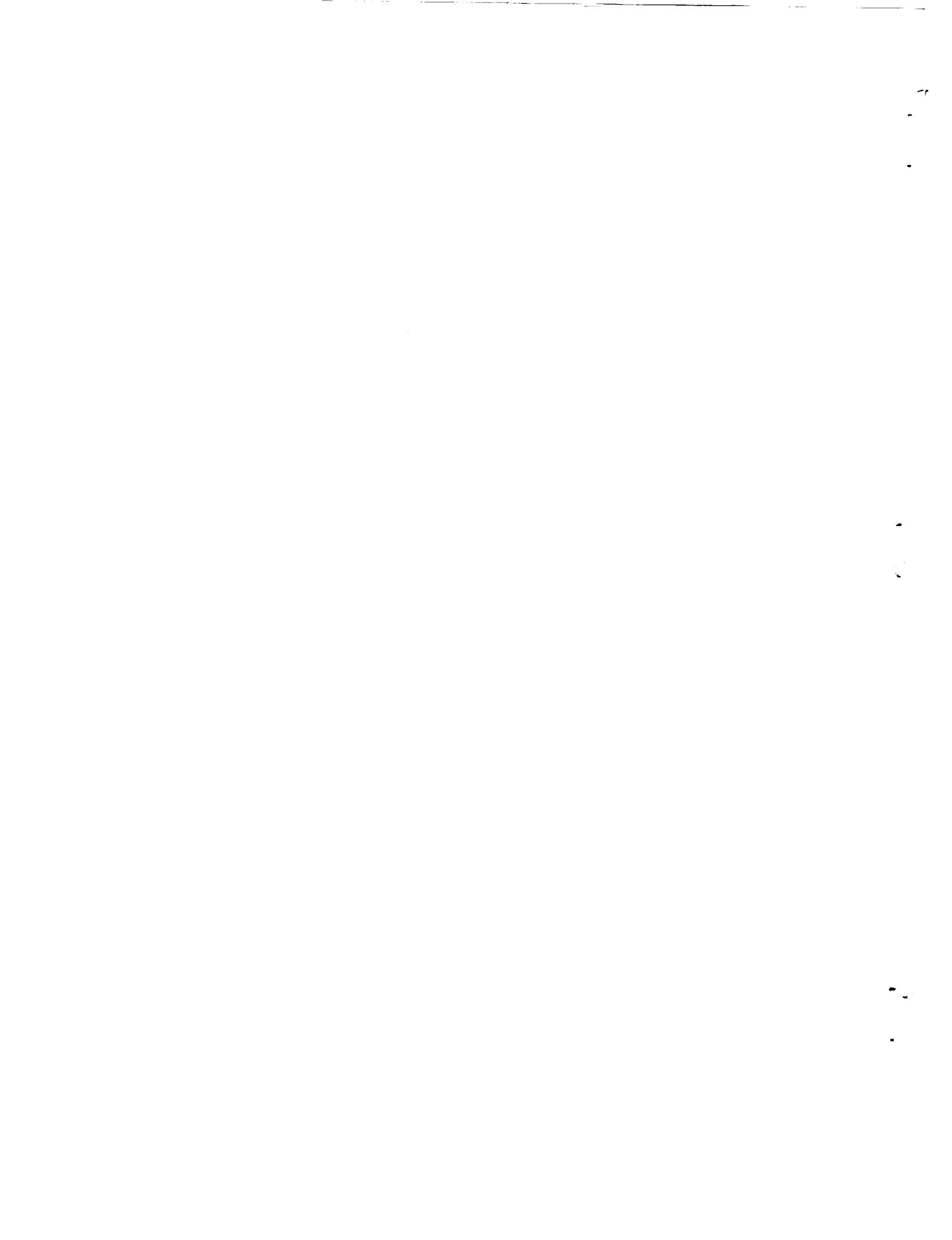
A TABULATION OF WIND-TUNNEL PRESSURE DATA AND
SECTION AERODYNAMIC CHARACTERISTICS AT MACH NUMBERS
OF 1.61 AND 2.01 FOR A REFLEX CAMBERED WING AND A
CAMBERED AND TWISTED WING HAVING
THE SAME SWEPT PLANFORM

By Emma Jean Landrum

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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
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THE SAME SWEPT PLANFORM

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SUMMARY

The pressure, section normal-force, and section pitching-moment coefficients for a cambered and twisted wing and a reflex cambered wing both having the same swept planform are tabulated. Both wings had an NACA 65A005 thickness distribution, 50° sweepback of the quarter chord, a taper ratio of 0.20, and an aspect ratio of 3.5. The cambered and twisted wing had at each spanwise station an $a = 0$ mean line modified to have a maximum height of 4-percent chord and a linear spanwise twist variation with 6° of washout at the tip. The reflex cambered wing had a 1-wave-length sinusoidal mean line with a leading-edge angle of attack of -6°. The wings were tested at Mach numbers of 1.61 and 2.01 with fixed transition at Reynolds numbers of 3.6×10^6 and 3.1×10^6 , respectively, through an angle-of-attack range from -20° to 20°. For convenience, previously published data for a flat wing, a cambered wing, and a linearly twisted wing of the same planform as the present wings have been included.

INTRODUCTION

The prediction of the changes in aerodynamic characteristics of wings when they distort under variable flight loads is of considerable interest in the design of efficient wings for supersonic aircraft. As part of a general investigation at low supersonic speeds of the effects of arbitrary camber and twist, a series of sweptback wings having the same planform but with systematic variations in surface shape has been tested. The tabulated results of a pressure investigation of the separate effects of camber and twist on the aerodynamic characteristics of sweptback wings at Mach numbers of 1.61 and 2.01 are presented in

reference 1, and a limited analysis of some of these results is presented in reference 2. The results of a force study of the same wings are given in reference 3. The section normal-force and section pitching-moment coefficients for the flat and twisted wings of reference 1 are given in reference 4.

The purpose of this report is to present the pressure, section normal-force, and section pitching-moment coefficients obtained at Mach numbers of 1.61 and 2.01 for two additional swept wings of the same planform, one with camber and twist and one with reflex camber. For convenience, some of the previously published data for a flat wing, a cambered wing, and a linearly twisted wing of the same planform as the present wings have been included. No analysis of the data has been made.

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SYMBOLS

$b/2$	semispan
C_p	pressure coefficient
c	wing chord
c_n	section normal-force coefficient
c_m	section pitching-moment coefficient (taken about midchord of wing mean aerodynamic chord)
M	Mach number
x	chordwise distance from wing leading edge
y	spanwise distance from wing root chord
α	angle of attack of wing root, deg

MODELS AND MODEL MOUNTING

Two semispan sweptback wings were tested, one with camber and twist and one with reflex camber. Both wings had an NACA 65A005 thickness distribution, 50° sweepback of the quarter chord, a taper ratio of 0.20, and an aspect ratio of 3.5. The cambered and twisted wing had at each spanwise station an $a = 0$ mean line modified to have a maximum height of 4-percent chord and a linear spanwise twist variation with 6° of

washout at the tip. The reflex cambered wing had a 1-wave-length sinusoidal mean line with a leading-edge angle of attack of -6° . A plan view of the models is shown in figure 1.

Each wing had seven streamwise rows of orifices located at 0.05, 0.20, 0.35, 0.50, 0.70, 0.825, and 0.95 semispan. Orifices were located on both surfaces of the wings. Chordwise orifice locations for the cambered and twisted wing are given in table I in terms of a nominal location. For the reflex cambered wing the chordwise orifice locations are 0.0125, 0.025, 0.050, 0.075, 0.100, 0.150, 0.200, 0.250, 0.300, 0.400, 0.500, 0.600, 0.700, 0.800, and 0.900 local chord at 0.050, 0.200, 0.350, and 0.500 semispan and are 0.025, 0.075, 0.150, 0.250, 0.350, 0.450, 0.550, 0.650, 0.750, and 0.850 local chord at 0.700, 0.825, and 0.950 semispan.

The semispan wings were mounted horizontally in the tunnel from a turntable in a boundary-layer bypass plate which was located vertically in the test section about 10 inches from the tunnel wall.

TESTS AND TEST PROCEDURES

The tests were conducted in the Langley 4- by 4-foot supersonic pressure tunnel at Mach numbers of 1.61 and 2.01. Transition was fixed about 1/2 inch from the wing leading edge by No. 60 carborundum grains.

Angle of attack was changed through a range from -20° to 20° by manual rotation of the turntable on which the models were mounted and was measured by a vernier scale outside the tunnel. The tests were conducted at a tunnel stagnation pressure of 15 pounds per square inch absolute which provided Reynolds numbers of 3.6×10^6 and 3.1×10^6 at Mach numbers of 1.61 and 2.01, respectively.

PRESENTATION OF RESULTS

For the convenience of the user of the data for the cambered and twisted wing and the reflex cambered wing, tables at the same Reynolds number and transition condition from previously published reports for a flat wing, a cambered wing, and a linearly twisted wing have been included in this report. These wings have the same planform as those of the present tests. The cambered wing has the same camber as the cambered and twisted wing and the linearly twisted wing has the same spanwise twist variation. The flat wing is designated wing F in references 1, 3, and 4; the cambered wing is designated wing C in references 1 and 3; the twisted wing is designated wing 1 in references 1, 3, and 4.

The pressure coefficients are presented in tables II to VI and the section normal-force and section pitching-moment coefficients are given in tables VII to XI. Data are presented in tables II and VII for the cambered and twisted wing, in tables III and VIII for the reflex cambered wings, in tables IV (from ref. 1) and IX (from ref. 4) for the flat wing, in tables V (from ref. 1) and X for the cambered wings, and in tables VI (from ref. 1) and XI (from ref. 4) for the linearly twisted wing.

Langley Research Center,
National Aeronautics and Space Administration,
Langley Station, Hampton, Va., June 18, 1962.

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REFERENCES

1. Grant, Frederick C.: A Tabulation of Wind-Tunnel Pressure Data at Mach Numbers of 1.61 and 2.01 for Five Swept Wings of the Same Plan Form but Different Surface Shapes. NACA RM L58D23, 1958.
2. Grant, Frederick C., and Mugler, John P., Jr.: Span Loadings Due to Wing Twist at Transonic and Supersonic Speeds. NACA RM L57D24a, 1957.
3. Landrum, Emma Jean, and Czarnecki, K. R.: Effects at Mach Numbers of 1.61 and 2.01 of Camber and Twist on the Aerodynamic Characteristics of Three Swept Wings Having the Same Planform. NASA TN D-929, 1961.
4. Landrum, Emma Jean: A Tabulation of Section Aerodynamic Characteristics at Mach Numbers of 1.61 and 2.01 for Four Swept Wings Having the Same Planform but Different Surface Shapes. NASA TN D-1244, 1962.

TABLE I-- CHORDWISE ORIFICE LOCATIONS FOR CAMBERED AND TWISTED WING

TABLE II
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING

(a) $M = 1.81$

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING
(a) $M = 1.61$ - Continued

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING
(a) M = 1.61 - Continued

x/c, nominal	Cp at y/b_2 of:														x/c, nominal					
	.05		.20		.35		.50		.70		.825		.95							
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower						
$\alpha = -.04$																				
.0125	.554	-.416	.516	-.361	.508	-.333	.511	-.297	.410	-.296	.464	-.286	.437	-.300	.438	-.265	.0125			
.025	.594	-.417	.558	-.361	.549	-.311	.540	-.296	.530	-.300	.540	-.302	.541	-.258	.025	.050				
.050	.537	-.447	.568	-.358	.559	-.312	.550	-.300	.540	-.304	.528	-.302	.541	-.258	.075	.100				
.100	.502	-.398	.592	-.374	.503	-.338	.504	-.293	.524	-.300	.504	-.288	.518	-.299	.531	-.250	.150			
.150	.476	-.165	.522	-.385	.477	-.319	.474	-.293	.493	-.300	.474	-.288	.485	-.299	.500	.200	.250			
.200	.434	-.030	.477	-.371	.412	-.357	.411	-.314	.437	-.304	.412	-.304	.417	-.233	.250	.300	.350			
.250	.408	-.009	.456	-.348	.403	-.355	.409	-.317	.422	-.290	.402	-.304	.417	-.233	.250	.300	.350			
.300	.388	-.074	.432	-.301	.394	-.349	.395	-.326	.411	-.298	.403	-.309	.420	-.215	.250	.300	.350			
.350	.401	-.022	.402	-.047	.408	-.038	.415	-.028	.421	-.060	.409	-.040	.428	-.199	.450	.400	.500			
.400	.450	-.008	.430	-.056	.401	-.097	.428	-.102	.409	-.309	.411	-.315	.407	-.278	.458	-.189	.550			
.450	.500	-.022	.427	-.087	.405	-.109	.419	-.116	.428	-.132	.413	-.130	.417	-.268	.460	.600	.650			
.500	.550	-.064	.453	-.100	.444	-.126	.421	-.128	.467	-.136	.430	-.121	.454	-.149	.700	.750	.800			
.550	.600	-.084	.470	-.098	.449	-.112	.408	-.129	.448	-.132	.420	-.113	.449	-.149	.750	.800	.850			
.600	.650	-.062	.462	-.085	.466	-.102	.430	-.102	.455	-.132	.427	-.113	.455	-.149	.750	.800	.850			
.650	.700	-.050	.458	-.074	.452	-.095	.445	-.102	.455	-.132	.427	-.113	.455	-.149	.750	.800	.850			
.700	.750	-.042	.450	-.067	.446	-.084	.428	-.102	.448	-.132	.420	-.113	.455	-.149	.750	.800	.850			
.750	.800	-.038	.446	-.060	.442	-.077	.421	-.102	.448	-.132	.420	-.113	.455	-.149	.750	.800	.850			
.800	.850	-.034	.442	-.055	.438	-.072	.418	-.102	.448	-.132	.420	-.113	.455	-.149	.750	.800	.850			
.850	.900	-.030	.438	-.050	.434	-.067	.413	-.102	.448	-.132	.420	-.113	.455	-.149	.750	.800	.850			
.900	.950	-.026	.434	-.045	.430	-.062	.413	-.102	.448	-.132	.420	-.113	.455	-.149	.750	.800	.850			
$\alpha = -.02$																				
.0125	.512	-.398	.478	-.331	.478	-.278	.482	-.255	.425	-.253	.425	-.242	.408	-.258	.412	-.248	.0125			
.025	.399	-.397	.415	-.329	.427	-.274	.376	-.253	.425	-.262	.408	-.258	.412	-.248	.025	.050				
.050	.309	-.413	.316	-.327	.320	-.274	.354	-.254	.300	-.288	.262	.298	-.245	.300	.075	.100				
.075	.253	-.354	.248	-.343	.254	-.278	.258	-.251	.160	-.248	.138	.262	-.246	.150	.200	.250				
.100	.210	-.039	.175	-.354	.175	-.286	.195	-.251	.160	-.248	.138	.262	-.246	.150	.200	.250				
.150	.130	-.036	.112	-.337	.126	-.308	.165	-.270	.160	-.267	.136	.260	-.246	.150	.200	.250				
.200	.094	-.006	.035	-.319	.065	-.318	.061	-.270	.077	-.248	.064	-.262	.069	-.256	.250	.300				
.250	.068	-.032	.011	-.291	.041	-.312	.053	-.274	.041	-.256	.031	-.256	.031	-.227	.300	.400				
.300	.044	-.042	.004	-.167	.015	-.300	.017	-.285	.041	-.256	.016	-.256	.016	-.216	.400	.500				
.350	.015	.008	-.061	.067	-.084	.258	-.076	-.271	.108	-.267	.085	-.251	.062	-.216	.450	.500				
.400	.000	-.042	-.001	-.091	-.037	-.133	-.201	-.139	-.247	-.157	-.271	-.139	-.248	-.103	-.209	.550	.600			
.450	.000	-.058	-.001	-.118	-.011	-.144	-.013	-.156	-.221	-.171	-.258	-.159	-.238	-.141	-.205	.650	.700			
.500	.000	-.095	-.024	-.127	-.017	-.161	-.052	-.174	-.196	-.178	-.285	-.161	-.228	-.178	-.202	.750	.800			
.550	.000	-.110	-.047	-.125	-.022	-.147	-.005	-.167	-.174	-.172	-.228	-.158	-.219	-.185	-.205	.850	.900			
.600	.000	-.123	-.009	-.138	-.010	-.168	-.011	-.183	$\alpha = .00$											
$\alpha = .02$																				
.0125	.471	-.373	.448	-.291	.444	-.233	.444	-.203	.400	-.179	.388	-.204	.391	-.209	.0125					
.025	.360	-.373	.380	-.293	.385	-.230	.340	-.204	.307	-.255	.256	-.207	.256	-.212	.025	.050				
.050	.264	-.365	.266	-.289	.276	-.230	.206	-.204	.207	-.255	.207	-.262	.256	-.212	.075	.100				
.075	.226	-.270	.198	-.305	.206	-.234	.207	-.205	.207	-.255	.207	-.262	.213	-.213	.150	.200				
.100	.147	-.004	.130	-.307	.159	-.243	.152	-.205	.111	-.188	.107	-.208	.138	-.213	.150	.200				
.150	.092	-.005	.058	-.268	.074	-.213	.064	-.211	.074	-.188	.107	-.208	.138	-.213	.150	.200				
.200	.046	-.027	.005	-.257	.016	-.263	.006	-.228	.052	-.191	.024	-.212	.025	-.215	.250	.300				
.250	.025	-.057	.028	-.245	.003	-.250	.010	-.228	.052	-.191	.024	-.212	.025	-.215	.250	.300				
.300	.008	-.003	.048	-.085	.086	-.235	.062	-.222	.081	-.199	.074	-.204	.070	-.214	.300	.400				
.350	.013	.045	-.096	-.087	.120	-.165	-.124	-.196	.141	-.207	.121	-.198	.117	-.210	.400	.500				
.400	.000	-.069	.036	-.123	.065	-.167	-.064	-.179	.179	-.203	.175	-.197	.155	-.201	.550	.600				
.450	.000	-.092	.029	-.149	.044	-.185	.112	-.201	.128	-.212	.184	-.197	.192	-.186	.650	.700				
.500	.000	-.127	.006	-.153	.015	-.194	.061	-.211	.100	-.209	.214	-.203	.181	-.224	.750	.800				
.550	.000	-.135	-.019	-.153	.011	-.184	.022	-.210	.072	-.207	.167	-.203	.173	-.216	.750	.800				
.600	.000	-.123	-.009	-.138	-.010	-.168	-.011	-.183	$\alpha = .02$											
$\alpha = .02$																				
.0125	.429	-.343	.417	-.248	.413	-.180	.415	-.143	.361	-.123	.356	-.142	.364	-.153	.0125					
.025	.323	-.341	.337	-.250	.348	-.178	.309	-.143	.361	-.123	.356	-.142	.364	-.153	.025	.050				
.050	.216	-.300	.219	-.245	.234	-.175	.234	-.145	.209	-.149	.211	-.157	.211	-.157	.075	.100				
.075	.164	-.123	.152	-.262	.159	-.181	.161	-.146	.209	-.149	.211	-.157	.211	-.157	.150	.200				
.100	.122	.023	.093	-.249	.084	-.189	.104	-.146	.206	-.149	.206	-.145	.209	-.161	.150	.200				
.150	.057	.031	.018	-.208	.019	-.206	.012	-.151	.060	-.127	.050	-.145	.099	-.161	.150	.200				
.200	.020	.061	-.042	-.187	-.034	-.201	-.046	-.168	.014	-.131	.027	-.152	.019	-.164	.250	.300				
.250	.003	.082	-.070	-.042	-.068	-.187	-.042	-.163	.014	-.131	.027	-.152	.019	-.164	.250	.300				
.300	-.024	.051	-.088	.179	-.110	-.168	-.103	-.149	-.123	-.136	-.119	-.140	-.105	-.170	.350	.400				
.400	-.056	.068	-.138	.116	-.161	-.074	-.171	-.111	-.168	-.140	-.163	-.131	-.154	-.166	.450	.500				
.450	-.102	.077	-.162	.101	-.211	.132	-.226	-.073	-.233	-.132	-.213	-.128	-.195	-.160	.550	.600				
.500	-.127	.079	-.179	.080	-.226	.130	-.229	-.041	-.251	-.106	-.235	-.125	-.228	-.147	.650	.700				
.550	-.163	.048	-.185	.056	-.226	.077	-.248	-.006	-.256	-.141	-.243	-.125	-.262	-.139	.750	.800				
.600	-.148	.007	-.187	.041	-.216	.067	-.251	.014	-.253	-.093	-.2									

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING

(a) $M = 1.61$ - Continued

x/c , nominal	Cp at $y/\frac{c}{2}$ of:														x/c , nominal		
	+.05		+.20		+.35		+.50		+.70		+.825		+.95				
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower			
$\alpha = .04$																	
.0125	.385	-.311	.377	-.202	.375	-.121	.379	-.080	.375	-.078	.328	-.050	.317	-.070	.337	-.082	.0125
.025	.284	-.302	.293	-.205	.298	-.117	.276	-.078	.328	-.050	.317	-.070	.337	-.082	.025		
.050	.166	-.233	.166	-.200	.179	-.113	.110	-.077	.157	-.078	.161	-.077	.163	-.086	.050		
.075	.115	-.004	.101	-.215	.110	-.119	.110	-.059	.081	-.083	.012	-.056	.006	-.073	.052	-.088	.075
.100	.075	.043	.042	-.190	.042	-.129	.059	-.081	.079	-.079	.073	-.058	.073	-.077	.063	-.093	.100
.150	.021	.061	-.026	-.148	-.020	-.143	-.083	-.083	.012	-.056	.006	-.073	.052	-.088	.200		
.200	-.015	.095	-.100	-.057	-.089	-.135	-.089	-.096	.079	-.079	.073	-.058	.073	-.077	.063	-.093	.250
.250	-.027	.105	-.115	.153	-.103	-.109	-.154	-.055	.167	-.058	.166	-.058	.148	-.096	.350		
.300	-.063	.110	-.134	.193	-.154	-.073	-.154	-.055	.230	-.049	.209	-.043	.196	-.089	.400		
.350	-.096	.131	-.176	.144	-.210	-.104	-.214	-.012	.278	-.039	.252	-.044	.233	-.078	.500		
.400	-.131	.115	-.200	.134	-.234	-.187	-.264	.018	.289	-.009	.274	-.041	.267	-.067	.600		
.500	-.157	.110	-.211	.109	-.269	-.140	-.274	.052	.289	-.009	.274	-.041	.267	-.067	.650		
.600	-.193	.079	-.214	.091	-.271	.103	-.278	.088	.293	-.044	.280	-.038	.288	-.072	.700		
.700	-.217	.040	-.216	.073	-.249	.072	-.281	.084	.292	-.001	.281	-.049	.261	-.091	.800		
.800	-.173	.040	-.216	.073	-.232	.055	-.278	-.029	.292	-.001	.281	-.049	.261	-.091	.850		
.900	-.167	.050	-.194	.062	-.232	.055	-.278	-.029	.292	-.001	.281	-.049	.261	-.091	.900		
.950	-.167	.050	-.194	.062	-.232	.055	-.278	-.029	.292	-.001	.281	-.049	.261	-.091	.950		
$\alpha = .06$																	
.0125	.347	-.266	.344	-.149	.341	-.056	.341	-.006	.290	-.037	.283	-.016	.298	-.002	.0125		
.025	.244	-.253	.262	-.153	.251	-.054	.229	-.004	.126	-.129	.009	.115	-.002	.050			
.050	.124	-.133	.120	-.149	.132	-.049	.061	-.003	.064	-.020	.035	-.024	.015	-.005	.075		
.075	.078	.043	.060	-.157	.061	-.053	.064	-.008	.011	.020	.035	-.024	.015	-.005	.100		
.100	.041	.076	.003	-.122	.001	-.064	.069	-.009	.020	.035	.015	.008	.008	-.005	.150		
.150	-.012	.089	-.062	-.052	-.064	-.069	-.129	-.009	.098	-.033	.108	-.013	.099	-.012	.200		
.200	-.040	.131	-.129	.116	-.131	-.054	-.129	-.009	.108	-.059	.196	-.039	.178	-.004	.300		
.250	-.040	.131	-.165	.217	-.165	-.054	-.052	-.143	.023	.098	.033	-.108	.013	-.012	.350		
.300	-.093	.170	-.168	.213	-.202	-.067	-.196	.063	.201	-.059	.196	-.039	.178	-.004	.400		
.350	-.123	.175	-.206	.180	-.248	-.244	-.253	.107	.255	-.078	.237	-.058	.227	-.032	.450		
.400	-.162	.162	-.234	.174	-.273	.198	-.303	.127	.298	-.082	.278	-.061	.265	-.042	.500		
.500	-.153	.167	-.237	.146	-.291	.157	-.314	.147	.313	-.099	.297	-.067	.298	-.041	.600		
.600	-.177	.148	-.237	.146	-.291	.157	-.314	.157	.318	-.062	.305	-.071	.308	-.028	.700		
.700	-.209	.119	-.240	.126	-.317	.132	-.323	.157	.318	-.062	.305	-.071	.308	-.028	.800		
.800	-.191	.076	-.238	.109	-.288	.105	-.314	.130	.318	-.086	.306	-.066	.281	-.011	.850		
.900	-.187	.090	-.214	.095	-.262	.090	-.304	-.034	.318	-.086	.306	-.066	.281	-.011	.900		
.950	-.187	.090	-.214	.095	-.262	.090	-.304	-.034	.318	-.086	.306	-.066	.281	-.011	.950		
$\alpha = .08$																	
.0125	.298	-.229	.296	-.091	.294	-.020	.298	.035	.239	-.109	.220	-.118	.241	-.115	.0125		
.025	.190	-.217	.211	-.097	.206	-.021	.180	.118	.239	-.109	.220	-.118	.241	-.115	.025		
.050	.072	.001	.065	-.088	.084	-.030	.030	.125	.078	-.071	.108	-.048	.103	-.075	.050		
.075	.031	.080	.009	-.079	.004	-.026	.023	.047	.047	.110	.090	.107	.124	-.048	.107		
.100	-.000	.111	-.047	-.023	-.048	-.015	-.047	.111	.111	.038	.120	-.090	.107	-.107	.100		
.150	-.048	.120	-.112	.110	-.111	-.111	-.108	.173	.137	.155	.190	-.171	.139	-.145	.200		
.200	-.073	.167	-.169	.239	-.180	-.088	-.173	.137	.349	-.179	.334	-.179	.337	-.171	.230		
.250	-.060	.171	-.204	.246	-.193	-.175	-.186	.159	.349	-.187	.337	-.187	.332	-.171	.230		
.300	-.140	.223	-.205	.250	-.239	-.246	-.231	.181	.344	-.244	.343	-.244	.343	-.237	.230		
.350	-.155	.225	-.237	.223	-.289	.270	-.288	.208	.345	-.208	.340	-.208	.340	-.237	.230		
.400	-.184	.213	-.277	.219	-.322	.224	-.335	.211	.345	-.179	.317	-.181	.305	-.157	.250		
.500	-.184	.213	-.269	.187	-.312	.195	-.347	.206	.349	-.187	.337	-.171	.332	-.128	.300		
.600	-.209	.192	-.271	.167	-.343	.172	-.355	.200	.349	-.147	.341	-.160	.336	-.092	.270		
.700	-.234	.159	-.271	.167	-.343	.172	-.355	.200	.349	-.147	.341	-.160	.336	-.092	.270		
.800	-.213	.110	-.266	.148	-.342	.151	-.351	.156	.347	-.139	.341	-.152	.307	-.053	.230		
.900	-.211	.131	-.241	.133	-.295	.134	-.342	-.027	.370	-.184	.370	-.184	.370	-.327	.088	.230	
.950	-.211	.131	-.241	.133	-.295	.134	-.342	-.027	.370	-.184	.370	-.184	.370	-.327	.088	.230	
$\alpha = .10$																	
.0125	.255	-.174	.255	-.022	.254	-.101	.244	.084	.217	-.179	.200	.143	.201	.172	.242	.0125	
.025	.140	-.159	.167	-.035	.155	-.103	.122	.217	.203	-.022	.008	.231	-.023	.276	.050		
.050	.029	.072	.022	-.004	.053	-.113	.116	-.015	.022	-.008	.231	-.023	.276	.075	.100		
.075	-.007	.111	-.033	-.033	-.038	-.114	-.015	.022	.022	-.008	.231	-.023	.276	.100	.100		
.100	-.031	.145	-.087	.110	-.086	.116	-.106	.195	.258	-.137	.284	-.096	.273	-.117	.285		
.150	-.076	.156	-.145	.231	-.149	.178	-.212	.250	.215	-.227	.265	-.208	.326	-.210	.280		
.200	-.102	.213	-.205	.281	-.212	.250	-.227	.256	.215	-.227	.265	-.210	.326	-.217	.280		
.250	-.090	.225	-.232	.284	-.232	.302	-.227	.265	.225	-.227	.265	-.210	.326	-.217	.280		
.300	-.165	.276	-.252	.287	-.271	.309	-.271	.280	.270	-.303	.282	-.294	.266	-.257	.350		
.350	-.178	.271	-.252	.265	-.315	.288	-.321	.286	.321	-.247	.317	-.284	.306	-.248	.400		
.400	-.204	.254	-.306	.264	-.353	.256	-.368	.259	.359	-.212	.351	-.234	.338	-.196	.500		
.500	-.230	.233	-.293	.227	-.358	.225	-.381	.241	.373	-.221	.364	-.209	.344	-.166	.600		
.600	-.211	.131	-.241	.133	-.295	.134	-.342	-.027	.370	-.184	.370	-.184	.370	-.327	.088	.230	
.700	-.253	.196	-.290	.211	-.353	.206	-.391	.225	.378	-.185	.369	-.196	.345	-.130	.700		
.800	-.232	.150	-.284	.192	-.363	.178	-.387	.189	.380	-.174	.370	-.188	.370	-.327	.088	.230	
.900	-.231	.173	-.260	.176	-.349	.167	-.368	-.027	.370	-.184	.370	-.184	.370	-.327	.088	.230	
.950	-.231	.173	-.260	.176	-.349	.167	-.368	-.027	.370	-.184	.370	-.184	.370	-.327	.088	.230	

TABLE II. - Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING

(a) $M = 1.61$ - Continued

x/c , nominal	C_p at y/b of:														x/c , nominal	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 12$																
.0125	.213	-.085	.212	.061	.206	.161	.186	.190	.097	.343	.046	.347	.056	.374	.0125	
.025	.094	-.101	.120	.049	.107	.176	.064	.304	.313	.047	.087	.358	.123	.354	.025	
.050	-.006	.111	-.012	.108	.010	.239	-.076	.267	-.050	-.047	-.087	.358	.123	.354	.050	
.075	-.039	.146	-.091	.167	-.076	.267	-.129	.330	-.047	-.047	-.087	.358	.123	.354	.075	
.100	-.065	.190	-.127	.247	-.149	.294	-.129	.330	-.047	-.047	-.087	.358	.123	.354	.100	
.150	-.107	.203	-.189	.306	-.194	.325	-.106	.349	-.251	-.266	-.245	.387	.152	.333	.204	
.200	-.131	.264	-.238	.325	-.250	.344	-.261	.351	-.251	-.266	-.243	.297	.260	.348	.200	
.250	-.114	.284	-.266	.327	-.267	.351	-.261	.356	-.245	-.251	-.243	.297	.260	.348	.250	
.300	-.184	.325	-.285	.332	-.307	.343	-.306	.354	-.314	-.317	-.317	.345	-.311	.303	.350	
.350	-.203	.320	-.274	.312	-.349	.326	-.351	.334	-.357	-.294	-.349	.326	-.342	.299	.400	
.400	-.224	.304	-.327	.308	-.381	.292	-.389	.301	-.357	-.294	-.349	.326	-.342	.299	.450	
.450	-.224	.304	-.327	.308	-.381	.292	-.389	.301	-.384	-.267	-.375	.272	-.366	.250	.500	
.500	-.250	.278	-.318	.270	-.397	.268	-.408	.286	-.395	-.262	-.392	.254	-.374	.206	.600	
.650	-.250	.278	-.318	.270	-.397	.268	-.408	.286	-.395	-.262	-.392	.254	-.374	.206	.650	
.700	-.269	.240	-.311	.253	-.392	.247	-.414	.256	-.402	-.236	-.397	.239	-.376	.170	.750	
.750	-.247	.197	-.304	.231	-.379	.223	-.412	.231	-.402	-.219	-.397	.227	-.369	.122	.800	
.800	-.247	.197	-.304	.231	-.379	.223	-.412	.231	-.402	-.219	-.397	.227	-.369	.122	.850	
.850	-.247	.223	-.278	.223	-.261	.215	-.286	-.042	-.219	-.397	.227	-.369	.122	.900	.950	
$a = 14$																
.0125	.169	-.012	.155	.141	.144	.287	.085	.393	.027	.441	-.069	.430	-.083	.435	.0125	
.025	.049	-.006	.045	.148	.042	.171	-.029	.390	-.080	-.027	.441	-.069	.430	-.083	.435	
.050	-.039	.168	-.036	.280	-.029	.396	-.129	.396	-.080	-.027	.441	-.069	.430	-.083	.435	
.075	-.074	.201	-.129	.326	-.115	.369	-.163	.415	-.139	-.199	.411	-.210	.408	-.075	.100	
.100	-.096	.242	-.185	.342	-.184	.392	-.163	.420	-.223	.428	-.236	.403	-.278	.404	.150	
.150	-.133	.267	-.232	.363	-.235	.397	-.205	.429	-.291	.417	-.282	.440	-.306	.323	.250	
.200	-.155	.315	-.275	.376	-.289	.405	-.230	.432	-.306	.418	-.295	.360	-.323	.406	.250	
.250	-.146	.342	-.303	.373	-.306	.401	-.260	.432	-.306	.418	-.295	.360	-.323	.406	.300	
.300	-.198	.369	-.316	.379	-.338	.395	-.338	.403	-.352	.400	-.352	.398	-.367	.365	.350	
.350	-.221	.367	-.316	.361	-.377	.372	-.378	.385	-.388	.345	-.380	.385	-.387	.364	.400	
.400	-.245	.353	-.342	.356	-.405	.334	-.411	.348	-.411	.305	-.405	.330	-.407	.303	.500	
.450	-.245	.353	-.342	.356	-.405	.334	-.411	.348	-.411	.305	-.405	.330	-.407	.303	.600	
.500	-.269	.333	-.344	.316	-.419	.314	-.423	.324	-.422	.308	-.418	.313	-.408	.267	.650	
.600	-.262	.290	-.334	.298	-.427	.300	-.432	.301	-.428	.282	-.424	.298	-.411	.222	.700	
.700	-.262	.290	-.334	.298	-.427	.300	-.432	.301	-.428	.282	-.424	.298	-.411	.222	.750	
.750	-.260	.249	-.324	.287	-.415	.275	-.429	.282	-.409	.273	-.424	.285	-.406	.177	.800	
.800	-.262	.265	-.293	.269	-.297	.264	-.306	-.040	-.219	-.357	-.425	.526	-.418	.372	.850	
.850	-.262	.265	-.293	.269	-.297	.264	-.306	-.040	-.219	-.357	-.425	.526	-.418	.372	.900	
.900	-.262	.265	-.293	.269	-.297	.264	-.306	-.040	-.219	-.357	-.425	.526	-.418	.372	.950	
$a = 16$																
.0125	.114	.068	.094	.248	.058	.413	-.023	.510	-.188	-.162	.540	-.218	.516	-.309	.540	.0125
.025	-.008	.134	.002	.288	-.040	.439	-.137	.480	-.239	-.289	.478	-.332	.518	-.075	.050	
.050	-.082	.207	-.078	.385	-.118	.447	-.177	.480	-.224	.490	-.300	.504	-.325	.478	.100	
.075	-.120	.255	-.163	.404	-.159	.456	-.224	.500	-.300	.504	-.325	.478	-.359	.577	.150	
.100	-.133	.292	-.219	.409	-.211	.462	-.224	.500	-.323	.486	-.339	.504	-.365	.425	.200	
.150	-.171	.329	-.278	.421	-.270	.458	-.300	.500	-.339	.488	-.339	.504	-.365	.425	.250	
.200	-.187	.367	-.316	.436	-.321	.464	-.332	.500	-.344	.486	-.344	.504	-.365	.425	.300	
.250	-.182	.399	-.338	.426	-.339	.456	-.344	.488	-.344	.478	-.342	.477	-.401	.478	.350	
.300	-.216	.423	-.350	.431	-.376	.448	-.374	.478	-.392	.477	-.401	.478	-.412	.585	.400	
.350	-.244	.418	-.363	.414	-.407	.435	-.407	.461	-.418	.415	-.416	.465	-.423	.550	.450	
.400	-.268	.404	-.359	.408	-.435	.407	-.434	.415	-.437	.384	-.435	.430	-.423	.451	.500	
.450	-.268	.383	-.368	.383	-.446	.389	-.446	.397	-.444	.390	-.441	.451	-.426	.394	.600	
.500	-.302	.347	-.355	.371	-.454	.369	-.454	.386	-.439	.358	-.431	.354	-.437	.389	.700	
.600	-.302	.321	-.336	.363	-.437	.357	-.436	.369	-.431	.357	-.425	.356	-.418	.372	.800	
.700	-.276	.321	-.336	.363	-.437	.357	-.436	.369	-.431	.357	-.425	.356	-.418	.372	.850	
.750	-.276	.321	-.336	.363	-.437	.357	-.436	.369	-.431	.357	-.425	.356	-.418	.372	.900	
.800	-.276	.321	-.336	.363	-.437	.357	-.436	.369	-.431	.357	-.425	.356	-.418	.372	.950	
$a = 18$																
.0125	.066	.152	.031	.387	-.028	.541	-.151	.641	-.247	.601	-.328	.604	-.387	.623	-.451	.637
.025	-.049	.207	-.056	.410	-.116	.553	-.247	.585	-.327	.540	-.428	.540	-.434	.621	-.451	.537
.050	-.154	.278	-.144	.439	-.190	.555	-.275	.585	-.346	.573	-.437	.559	-.457	.556	-.451	.537
.075	-.154	.278	-.144	.439	-.190	.555	-.275	.585	-.346	.573	-.437	.559	-.457	.556	-.451	.537
.100	-.167	.314	-.253	.485	-.267	.572	-.299	.573	-.377	.577	-.437	.559	-.457	.556	-.451	.537
.150	-.196	.396	-.313	.582	-.308	.554	-.374	.555	-.437	.559	-.437	.559	-.457	.556	-.451	.537
.200	-.208	.426	-.349	.510	-.352	.566	-.374	.555	-.437	.559	-.437	.559	-.457	.556	-.451	.537
.250	-.204	.460	-.373	.510	-.366	.558	-.388	.555	-.437	.559	-.437	.559	-.457	.556	-.451	.537
.300	-.231	.484	-.386	.534	-.398	.543	-.410	.546	-.437	.559	-.437	.559	-.457	.556	-.451	.537
.400	-.261	.496	-.402	.521	-.427	.515	-.439	.525	-.437	.559	-.437	.559	-.457	.556	-.451	.537
.450	-.284	.510	-.384	.506	-.454	.482	-.460	.482	-.457	.559	-.457	.559	-.457	.556	-.451	.537
.500	-.303	.485	-.387	.469	-.467	.451	-.469	.472	-.457	.559	-.457	.559	-.457	.556	-.451	.537
.550	-.303	.485	-.387	.469	-.467	.451	-.469	.472	-.457	.559	-.457	.559	-.457	.556	-.451	.537
.600	-.315	.439	-.377	.447	-.475	.441	-.475	.463	-.457	.559	-.457	.559	-.457	.556	-.451	.537
.700	-.315	.439	-.377	.447	-.475	.441	-.475	.463	-.457							

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING

(a) M = 1.61 - Concluded

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING
(b) $M = 2.01$

x/c , nominal	C_p at $y/\frac{c}{2}$ of:														x/c , nominal	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -20$																
.0125	.828	-.303	.691	-.310	.649	-.312	.597	-.308	.617	-.306	.556	-.286	.542	-.255	.0125	
.025	.820	-.304	.741	-.311	.719	-.317	.664	-.311	.655	-.308	.556	-.286	.542	-.255	.025	
.050	.759	-.314	.618	-.307	.689	-.297	.655	-.308	.617	-.306	.556	-.286	.542	-.255	.050	
.075	.716	-.307	.630	-.305	.664	-.312	.629	-.284	.645	-.302	.559	-.283	.606	-.252	.075	
.100	.683	-.298	.621	-.307	.618	-.312	.629	-.284	.575	-.302	.539	-.281	.539	-.246	.100	
.150	.555	-.294	.575	-.302	.595	-.312	.584	-.312	.575	-.302	.513	-.299	.491	-.275	.150	
.200	.505	-.270	.521	-.300	.533	-.304	.528	-.309	.505	-.295	.475	-.279	.414	-.246	.200	
.250	.464	-.239	.462	-.297	.481	-.306	.491	-.305	.475	-.299	.410	-.273	.439	-.241	.250	
.300	.442	-.254	.292	-.292	.433	-.298	.443	-.294	.414	-.294	.410	-.273	.439	-.231	.300	
.350																
.400	.403	-.187	.345	-.302	.342	-.300	.365	-.299	.345	-.287	.323	-.266	.396	-.221	.400	
.450																
.500	.319	-.188	.295	-.293	.287	-.300	.288	-.291	.305	-.279	.323	-.266	.358	-.214	.500	
.550																
.600	.288	-.185	.251	-.289	.253	-.297	.261	-.273	.286	-.285	.314	-.264	.342	-.212	.600	
.650																
.700	.248	-.190	.217	-.293	.218	-.300	.239	-.266	.279	-.273	.308	-.273	.321	-.209	.700	
.750																
.800	.197	-.179	.203	-.290			.235	-.291	.286	-.280	.317	-.270	.343	-.210	.800	
.850																
.900	.220	-.190	.159	-.282	.221	-.292	.252	-.280								
.950																
$\alpha = -18$																
.0125	.787	-.305	.680	-.318	.637	-.318	.597	-.317	.610	-.309	.555	-.285	.537	-.252	.0125	
.025	.769	-.310	.710	-.323	.690	-.324	.644	-.317	.620	-.317	.563	-.281	.584	-.251	.025	
.050	.705	-.321	.680	-.317	.649	-.305	.623	-.317	.620	-.308	.504	-.280	.506	-.249	.050	
.075	.663	-.317	.638	-.320	.621	-.317	.599	-.317	.538	-.308	.475	-.291	.459	-.243	.100	
.100	.619	-.318	.580	-.317	.565	-.319	.580	-.288	.538	-.308	.504	-.280	.506	-.249	.150	
.150	.505	-.313	.521	-.318	.545	-.323	.539	-.317	.475	-.309	.450	-.276	.459	-.243	.200	
.200	.448	-.275	.468	-.320	.477	-.317	.480	-.320	.475	-.286	.274	-.266	.311	-.217	.250	
.250	.406	-.243	.399	-.317	.430	-.320	.441	-.321	.475	-.292	.235	-.292	.262	-.264	.300	
.300	.383	-.229		-.309	.383	-.315	.394	-.305	.372	-.303	.235	-.288	.263	-.267	.350	
.350																
.400	.344	-.167	.293	-.318	.295	-.317	.318	-.309	.294	-.296	.257	-.286	.274	-.225	.400	
.450																
.500	.274	-.172	.242	-.305	.236	-.316	.239	-.305	.257	-.286	.274	-.266	.311	-.217	.500	
.550																
.600	.251	-.166	.205	-.304	.207	-.313	.214	-.292	.235	-.292	.257	-.271	.268	-.207	.600	
.650																
.700	.203	-.173	.169	-.304	.177	-.311	.188	-.286	.229	-.288	.263	-.267	.286	-.209	.700	
.750																
.800	.154	-.166	.163	-.302		-.309	.192	-.300	.235	-.288	.263	-.267	.286	-.207	.800	
.850																
.900	.177	-.181	.114	-.275	.180	-.304	.204	-.285								
.950																
$\alpha = -16$																
.0125	.750	-.311	.664	-.325	.631	-.319	.592	-.324	.603	-.309	.551	-.281	.537	-.252	.0125	
.025	.714	-.317	.679	-.326	.661	-.323	.620	-.325	.586	-.311	.544	-.275	.559	-.251	.025	
.050	.653	-.326	.630	-.320	.610	-.309	.585	-.326	.586	-.311	.466	-.272	.475	-.248	.050	
.075	.606	-.326	.568	-.322	.573	-.318	.534	-.322	.486	-.326	.495	-.311	.466	-.268	.100	
.100	.541	-.326	.531	-.322	.520	-.319	.534	-.322	.486	-.326	.495	-.311	.466	-.268	.150	
.150	.448	-.262	.468	-.326	.488	-.322	.486	-.326	.429	-.326	.429	-.321	.475	-.248	.200	
.200	.399	-.269	.410	-.326	.424	-.322	.429	-.326	.378	-.309	.403	-.268	.415	-.245	.250	
.250	.358	-.213	.345	-.326	.376	-.322	.391	-.326	.349	-.311	.403	-.268	.415	-.245	.300	
.300	.327	-.189		-.312	.326	-.319	.341	-.311	.326	-.307	.320	-.261	.345	-.238	.350	
.350																
.400	.281	-.150	.241	-.318	.241	-.318	.263	-.314	.242	-.305	.262	-.262	.301	-.231	.400	
.450																
.500	.215	-.163	.188	-.310	.185	-.318	.184	-.311	.204	-.292	.220	-.258	.262	-.218	.500	
.550																
.600	.197	-.156	.148	-.300	.157	-.314	.158	-.300	.185	-.293	.210	-.258	.243	-.213	.600	
.650																
.700	.155	-.166	.131	-.295	.120	-.308	.137	-.294	.175	-.289	.205	-.262	.212	-.208	.700	
.750																
.800	.112	-.157	.124	-.278		-.305	.132	-.302	.178	-.289	.205	-.261	.228	-.206	.800	
.850																
.900	.140	-.170	.078	-.216	.140	-.299	.154	-.283	.127	-.284	.153	-.269	.166	-.208	.900	
.950																
$\alpha = -14$																
.0125	.705	-.334	.637	-.325	.614	-.319	.578	-.318	.591	-.318	.586	-.308	.536	-.284	.519	-.256
.025	.662	-.334	.632	-.325	.622	-.321	.591	-.318	.586	-.308	.536	-.284	.519	-.256	.025	
.050	.596	-.323	.572	-.318	.572	-.310	.546	-.320	.548	-.317	.516	-.278	.525	-.256	.050	
.075	.541	-.323	.530	-.321	.528	-.316	.545	-.327	.548	-.317	.516	-.278	.525	-.256	.075	
.100	.494	-.327	.474	-.322	.467	-.318	.518	-.320	.548	-.317	.516	-.278	.525	-.256	.100	
.150	.392	-.317	.413	-.328	.434	-.322	.539	-.321	.540	-.308	.424	-.275	.441	-.254	.150	
.200	.348	-.238	.355	-.326	.369	-.322	.582	-.321	.578	-.309	.360	-.270	.366	-.251	.200	
.250	.305	-.155	.284	-.324	.318	-.328	.342	-.321	.378	-.309	.310	-.270	.366	-.251	.250	
.300	.276	-.156		-.307	.282	-.325	.286	-.310	.274	-.306	.275	-.265	.290	-.244	.300	
.350																
.400	.231	-.134	.188	-.302	.191	-.319	.214	-.316	.198	-.306	.275	-.265	.290	-.244	.400	
.450																
.500	.172	-.147	.142	-.291	.133	-.312	.134	-.311	.152	-.296	.163	-.261	.204	-.223	.500	
.550																
.600	.153	-.138	.102	-.281	.103	-.304	.115	-.302	.130	-.294	.153	-.261	.183	-.213	.600	
.650																
.700	.113	-.151	.090	-.264	.											

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING
(b) $M = 2.01$ - Continued

x/c , nominal	C_p at y/b of:														x/c , nominal	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -12^\circ$																
.0125	.674	-.308	.618	-.316	.602	-.308	.570	-.312	.576	-.299	.530	-.278	.518	-.250	.0125	
.025	.629	-.300	.599	-.313	.597	-.311	.565	-.312	.576	-.299	.530	-.278	.518	-.250	.025	
.050	.542	-.315	.535	-.311	.535	-.306	.519	-.312	.520	-.295	.495	-.270	.500	-.248	.050	
.075	.486	-.315	.492	-.313	.492	-.310	.440	-.312	.412	-.301	.397	-.265	.416	-.245	.075	
.100	.434	-.315	.431	-.314	.425	-.310	.391	-.311	.412	-.301	.397	-.265	.416	-.245	.100	
.150	.346	-.293	.363	-.319	.387	-.313	.320	-.311	.338	-.298	.326	-.260	.326	-.241	.150	
.200	.302	-.158	.313	-.311	.324	-.318	.330	-.311	.320	-.290	.326	-.260	.326	-.241	.200	
.250	.260	-.108	.242	-.305	.276	-.319	.290	-.312	.338	-.298	.326	-.260	.326	-.241	.250	
.300	.230	-.117	.287	-.238	.315	-.315	.245	-.303	.238	-.297	.244	-.257	.251	-.234	.300	
.350	.189	-.119	.155	-.267	.155	-.302	.170	-.305	.162	-.299	.162	-.256	.210	-.224	.400	
.400	.138	-.123	.102	-.247	.096	-.289	.094	-.297	.110	-.294	.130	-.251	.162	-.214	.500	
.500	.117	-.118	.072	-.235	.071	-.279	.070	-.290	.090	-.283	.115	-.253	.142	-.206	.600	
.600	.072	-.131	.060	-.191	.046	-.270	.047	-.284	.083	-.263	.108	-.258	.108	-.201	.700	
.700	.039	-.127	.050	-.125		-.263	.050	-.275	.087	-.264	.112	-.266	.120	-.205	.800	
.800	.061	-.135	.018	-.120	.052	-.258	.061	-.249							.900	
.900	.030	-.123	.016	-.114	.009	-.229	.013	-.235							.950	
$\alpha = -10^\circ$																
.0125	.639	-.310	.586	-.312	.576	-.304	.554	-.308	.557	-.298	.512	-.284	.509	-.254	.0125	
.025	.577	-.310	.557	-.312	.561	-.304	.524	-.306	.520	-.298	.462	-.278	.468	-.252	.025	
.050	.488	-.318	.485	-.309	.493	-.307	.481	-.307	.480	-.297	.379	-.271	.379	-.249	.050	
.075	.419	-.318	.435	-.312	.447	-.307	.394	-.282	.372	-.294	.352	-.271	.379	-.249	.100	
.100	.375	-.312	.382	-.317	.371	-.307	.345	-.310	.329	-.309	.326	-.266	.326	-.243	.200	
.150	.293	-.261	.314	-.317	.333	-.310	.345	-.310	.329	-.309	.327	-.271	.327	-.243	.300	
.200	.253	-.109	.259	-.305	.266	-.318	.279	-.312	.288	-.295	.282	-.266	.285	-.243	.400	
.250	.211	-.099	.191	-.292	.228	-.318	.237	-.312	.219	-.302	.190	-.294	.201	-.237	.500	
.300	.188	-.099	.180	-.270	.181	-.309	.192	-.302	.180	-.294	.201	-.264	.203	-.237	.600	
.350	.145	-.104	.109	-.241	.107	-.288	.120	-.295	.118	-.298		-.262	.164	-.230	.700	
.400	.095	-.109	.063	-.214	.042	-.267	.050	-.286	.066	-.292	.084	-.262	.119	-.224	.800	
.500	.075	-.104	.033	-.189	.024	-.255	.026	-.277	.048	-.282	.070	-.265	.090	-.218	.900	
.600	.033	-.119	.018	-.130	.006	-.242	.004	-.269	.038	-.261	.060	-.274	.068	-.221	.950	
$\alpha = -8^\circ$																
.0125	.597	-.310	.560	-.310	.552	-.293	.534	-.303	.519	-.290	.499	-.289	.483	-.265	.0125	
.025	.531	-.310	.542	-.310	.533	-.296	.492	-.303	.493	-.293	.425	-.286	.429	-.265	.025	
.050	.423	-.315	.442	-.308	.454	-.300	.453	-.303	.435	-.297	.321	-.283	.341	-.264	.050	
.075	.364	-.309	.382	-.310	.402	-.295	.330	-.297	.347	-.278	.322	-.287	.321	-.264	.100	
.100	.328	-.294	.336	-.314	.330	-.297	.347	-.278	.322	-.294	.321	-.283	.341	-.264	.200	
.150	.250	-.216	.258	-.310	.282	-.303	.294	-.303	.282	-.303	.207	-.287	.235	-.279	.300	
.200	.198	-.077	.201	-.289	.220	-.307	.233	-.308	.202	-.311	.240	-.287	.242	-.262	.400	
.250	.168	-.091	.144	-.268	.178	-.299	.178	-.298	.202	-.311	.240	-.287	.153	-.272	.500	
.300	.144	-.080	.127	-.237	.129	-.286	.144	-.293	.146	-.287			.159	-.256	.600	
.350	.105	-.082	.064	-.204	.064	-.256	.078	-.274	.078	-.285			.127	-.251	.700	
.400	.059	-.088	.018	-.176	.003	-.233	.011	-.261	.024	-.277	.045	-.267	.080	-.240	.800	
.500	.049	-.088	-.005	-.133	-.016	-.217	-.012	-.249	.006	-.267	.026	-.259	.054	-.230	.900	
.600	.003	-.099	-.020	-.088	-.035	-.205	-.035	-.240	.001	-.220	.020	-.251	.025	-.222	.700	
.700	-.024	-.104	-.025	-.086	-.199	-.041	-.220	-.006	-.224	.016	-.243	.026	-.220	.850	-.800	
.800	-.008	-.104	-.046	-.101	-.025	-.194	-.031	-.213							.900	
.900	-.033	-.082	-.070	-.083	-.067	-.154	-.074	-.172							.950	
$\alpha = -6^\circ$																
.0125	.556	-.302	.534	-.298	.531	-.289	.504	-.296	.505	-.270	.488	-.278	.481	-.269	.0125	
.025	.470	-.304	.490	-.295	.501	-.290	.448	-.294	.411	-.276	.402	-.276	.400	-.270	.025	
.050	.372	-.308	.396	-.296	.420	-.298	.415	-.298							.075	
.075	.319	-.300	.338	-.299	.360	-.291	.304	-.277							.100	
.100	.278	-.277	.289	-.304	.296	-.293	.293	-.293							.150	
.150	.198	-.139	.213	-.290	.226	-.303	.234	-.299	.289	-.269	.288	-.275	.312	-.268	.200	
.200	.153	-.056	.155	-.258	.173	-.298	.173	-.305	.202	-.270	.196	-.273	.211	-.269	.350	
.250	.127	-.075	.099	-.226	.142	-.283	.146	-.305	.202	-.270					.300	
.300	.104	-.059	.198	-.088	.268	-.268	.087	-.269							.400	
.350	.063	-.058	.026	-.155	.025	-.228	.026	-.238							.500	
.400	.025	-.068	-.016	-.116	-.025	-.198	-.036	-.219							.600	
.500	.005	-.064	-.040	-.083	-.050	-.183	-.059	-.208							.700	
.600	.030	-.076	-.051	-.066	-.074	-.171	-.080	-.194							.800	
.700	-.051	-.090	-.051	-.069	-.162	-.083	-.177								.900	
.800	-.033	-.082	-.070	-.083	-.067	-.154	-.074	-.172							.950	

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING

(b) $M = 2.01$ - Continued

y/c , nominal	C_p at $y/b/2$ of:														y/c , nominal	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -.04$																
.0125	.524	-.283	.517	-.276	.517	-.262	.506	-.260	.496	-.242	.462	-.261	.467	-.261	.0125	
.025	.437	-.283	.472	-.274	.487	-.262	.442	-.257	.389	-.260	.380	-.258	.379	-.265	.025	
.050	.334	-.288	.367	-.274	.389	-.272	.389	-.260	.380	-.244	.383	-.258	.379	-.265	.050	
.075	.281	-.269	.307	-.277	.322	-.262	.250	-.268	.264	-.244	.247	-.244	.244	-.258	.075	
.100	.242	-.246	.247	-.247	.250	-.242	.204	-.247	.204	-.247	.247	-.244	.244	-.258	.100	
.150	.160	-.042	.174	-.246	.161	-.247	.080	-.267	.139	-.247	.176	-.247	.163	-.257	.150	
.200	.112	-.028	.107	-.204	.125	-.255	.121	-.261	.121	-.244	.176	-.247	.163	-.257	.200	
.250	.092	-.046	.062	-.171	.094	-.234	.121	-.261	.121	-.244	.176	-.247	.163	-.257	.250	
.300	.073	-.034	.142	-.142	.046	-.213	.063	-.230	.063	-.238	.078	-.255	.088	-.265	.300	
.350	.040	-.037	.026	-.007	-.094	-.011	-.168	-.003	-.195	-.005	-.226	-.0238	-.047	-.254	.350	
.400	.050	-.005	-.038	-.047	-.051	-.059	-.136	-.057	-.173	-.040	-.211	-.018	-.222	-.012	-.239	.400
.500	.060	-.010	-.034	-.066	-.034	-.082	-.116	-.079	-.155	-.056	-.194	-.038	-.212	-.009	-.228	.500
.600	.065	-.050	-.058	-.077	-.045	-.094	-.101	-.124	-.065	-.120	-.050	-.198	-.044	-.208	.600	
.700	.050	-.066	-.058	-.077	-.045	-.094	-.101	-.124	-.065	-.120	-.050	-.198	-.044	-.208	.700	
.750	.050	-.066	-.058	-.077	-.045	-.094	-.101	-.124	-.065	-.120	-.050	-.198	-.044	-.208	.750	
.800	.050	-.052	-.058	-.093	-.057	-.092	-.089	-.095	-.122	-.068	-.178	-.052	-.181	-.045	-.186	.800
.850	.050	-.052	-.058	-.093	-.057	-.092	-.089	-.095	-.122	-.068	-.178	-.052	-.181	-.045	-.186	.850
.900	.050	-.052	-.058	-.093	-.057	-.092	-.089	-.095	-.122	-.068	-.178	-.052	-.181	-.045	-.186	.900
.950	.050	-.052	-.058	-.093	-.057	-.092	-.089	-.095	-.122	-.068	-.178	-.052	-.181	-.045	-.186	.950
$\alpha = -.02$																
$\alpha = .00$																
.0125	.447	-.246	.461	-.244	.459	-.243	.458	-.240	.491	-.233	.418	-.253	.428	-.277	.0125	
.025	.349	-.243	.386	-.244	.390	-.243	.366	-.239	.298	-.242	.303	-.256	.309	-.276	.025	
.050	.241	-.252	.280	-.241	.293	-.249	.316	-.242	.298	-.227	.166	-.251	.207	-.276	.050	
.075	.190	-.188	.210	-.248	.230	-.242	.192	-.231	.170	-.227	.166	-.251	.207	-.276	.075	
.100	.153	-.105	.156	-.234	.174	-.250	.192	-.231	.170	-.227	.166	-.251	.207	-.276	.100	
.150	.088	-.002	.088	-.154	.108	-.223	.119	-.246	.170	-.189	.088	-.218	.093	-.276	.150	
.200	.043	.020	.024	-.096	.043	-.159	.055	-.208	.040	-.170	.089	-.189	.088	-.218	.200	
.250	.033	.002	-.013	-.062	.018	-.116	.037	-.170	.040	-.120	.095	-.120	.105	-.129	.250	
.300	.011	.023	-.034	-.034	-.031	-.085	-.016	-.122	-.001	-.147	.004	-.188	.016	-.226	.300	
.400	-.032	.026	-.076	-.002	-.089	-.041	-.076	-.079	-.065	-.124	-.144	-.025	-.182	-.400		
.450	-.057	.015	-.109	-.027	-.123	-.022	-.124	-.051	-.104	-.108	-.083	-.136	-.065	-.164	.450	
.500	-.077	.018	-.124	-.021	-.147	-.011	-.143	-.040	-.120	-.095	-.128	-.105	-.129	-.155	.500	
.600	-.100	-.001	-.129	-.006	-.162	-.013	-.159	-.040	-.128	-.113	-.122	-.111	-.133	-.750	.700	
.700	-.100	-.017	-.129	-.005	-.018	-.165	-.033	-.133	-.080	-.119	-.115	-.116	-.126	-.800	.800	
.750	-.106	-.017	-.129	-.005	-.018	-.165	-.033	-.133	-.080	-.119	-.115	-.116	-.126	-.850	.850	
.800	-.098	-.016	-.139	-.017	-.151	-.022	-.160	-.034	-.148	-.137	-.043	-.137	-.086	-.900	.900	
.850	-.098	-.016	-.139	-.017	-.151	-.022	-.160	-.034	-.148	-.137	-.043	-.137	-.086	-.950	.950	
$\alpha = .02$																
.0125	.413	-.222	.427	-.232	.426	-.268	.433	-.258	.409	-.248	.390	-.266	.405	-.269	.0125	
.025	.310	-.217	.254	-.233	.355	-.258	.281	-.256	.281	-.261	.271	-.272	.271	-.268	.025	
.050	.200	-.230	.235	-.245	.232	-.256	.281	-.263	.281	-.261	.271	-.272	.271	-.268	.050	
.075	.150	-.143	.171	-.232	.186	-.262	.157	-.247	.157	-.231	.135	-.230	.129	-.257	.100	
.100	.117	-.051	.115	-.202	.125	-.242	.125	-.242	.125	-.231	.135	-.230	.172	-.267	.150	
.150	.049	.019	.053	-.091	.063	-.151	.086	-.231	.086	-.146	.054	-.186	.056	-.244	.200	
.200	.011	.046	-.017	-.004	-.000	-.042	.016	-.160	.016	-.146	.054	-.186	.056	-.244	.250	
.250	.005	.029	-.049	.015	-.024	.008	-.005	-.118	.005	-.146	.054	-.186	.056	-.244	.300	
.300	-.015	.053	-.022	-.070	.028	-.046	-.042	-.042	-.033	-.093	-.024	-.153	-.010	-.173	.350	
.400	-.052	.059	-.106	.043	-.122	.042	-.106	.041	-.084	-.077	-.103	-.052	-.133	-.400	.400	
.450	-.080	.045	-.135	.052	-.164	.034	-.150	.050	-.128	-.026	-.103	-.097	-.089	-.122	.450	.500
.500	-.098	.055	-.148	.039	-.179	.027	-.167	.044	-.142	-.003	-.124	-.088	-.110	-.114	.600	.600
.600	-.120	.027	-.152	.030	-.190	.019	-.181	.037	-.148	-.037	-.134	-.074	-.130	-.087	.700	.700
.700	-.118	.008	-.151	.019	-.174	.003	-.184	.010	-.153	-.003	-.137	-.043	-.137	-.086	.800	.800
.800	-.118	.008	-.151	.019	-.174	.003	-.184	.010	-.153	-.003	-.137	-.043	-.137	-.086	.850	.850
.900	-.110	.011	-.158	.004	-.174	.003	-.184	.010	-.153	-.003	-.137	-.043	-.137	-.086	.900	.900
.950	-.110	.011	-.158	-.004	-.174	-.003	-.184	-.010	-.153	-.003	-.137	-.043	-.137	-.086	.950	.950

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING
(b) M = 2.01 - Continued

x/c , nominal	C_p at $y/b/2$ of:														x/c , nominal	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = .04$																
.0125	.370	-.194	.390	-.217	.401	-.255	.407	-.244	.376	-.248	.365	-.254	.386	-.249	.0125	
.025	.258	-.187	.319	-.217	.329	-.250	.306	-.244	.376	-.248	.365	-.254	.386	-.249	.025	
.050	.157	-.183	.190	-.217	.217	-.249	.250	-.244	.226	-.234	.239	-.249	.239	-.249	.050	
.075	.107	-.103	.129	-.206	.148	-.244	.115	-.236	.179	-.179	.092	-.202	.143	-.246	.075	
.100	.074	-.006	.071	-.141	.096	-.215	.039	-.163	.094	-.179	.092	-.202	.143	-.246	.100	
.150	.014	.038	.011	-.013	.033	-.126	.019	-.015	.100	-.100	.015	-.137	.023	-.190	.150	
.200	-.022	.068	-.038	.065	-.029	.019	-.015	.042	-.028	.009	-.096	.015	-.137	.023	.200	
.250	-.023	.063	-.080	.082	-.053	.087	-.042	.074	-.017	.065	-.156	.029	-.133	.061	.250	
.300	-.045	.087	-.100	.086	-.099	.079	.072	-.068	-.013	-.063	-.084	-.037	-.111	.300	.350	
.350	-.081	.096	-.137	.091	-.145	.104	-.133	.108	-.119	.054	-.030	-.082	-.079	.400	.450	
.400	-.137	.079	-.160	.095	-.183	.086	-.176	.103	-.162	.058	-.138	.005	-.116	-.070	.500	
.450	-.106	.079	-.160	.095	-.183	.086	-.176	.103	-.162	.058	-.138	.005	-.116	-.070	.550	
.500	-.132	.091	-.175	.077	-.204	.074	-.193	.090	-.174	.065	-.156	.029	-.133	.061	.600	
.600	-.132	.091	-.175	.077	-.204	.074	-.193	.090	-.174	.065	-.156	.029	-.133	.061	.650	
.700	-.145	.061	-.175	.064	-.214	.061	-.205	.079	-.184	.050	-.170	.044	-.157	-.032	.750	
.750	-.137	.034	-.173	.053	-.049	-.209	.061	-.184	.050	-.170	.044	-.163	-.010	.800	.850	
.800	-.133	.040	-.176	.036	-.202	.036	-.209	.039	-.195	.050	-.170	.044	-.163	-.010	.900	
.900	-.133	.040	-.176	.036	-.202	.036	-.209	.039	-.195	.050	-.170	.044	-.163	-.010	.950	
$\alpha = .06$																
.0125	.333	-.163	.359	-.184	.366	-.228	.372	-.232	.351	-.223	.339	-.241	.354	-.240	.0125	
.025	.219	-.156	.279	-.184	.283	-.222	.259	-.230	.209	-.245	.200	-.235	.200	-.235	.025	
.050	.123	-.141	.159	-.183	.188	-.207	.212	-.235	.196	-.209	.209	-.245	.200	-.235	.050	
.075	.071	-.036	.040	-.179	.100	-.215	.170	-.210	.072	-.210	.065	-.136	.072	-.153	.075	
.100	.035	.033	.035	-.092	.052	-.215	.072	-.210	.120	-.210	.111	-.230	.111	-.230	.100	
.150	.014	.068	-.026	.057	-.008	.003	.004	-.113	.065	-.136	.072	-.153	.111	-.230	.150	
.200	-.040	.098	-.087	.115	-.064	.109	-.056	.013	.011	.054	-.005	-.090	-.003	-.133	.200	
.250	-.042	.087	-.108	.118	-.089	.135	-.078	.121	-.011	.054	-.005	-.090	-.003	-.133	.250	
.300	-.065	.121	-.135	.136	-.141	.114	-.162	-.088	.144	-.080	.036	-.077	-.046	.300	.350	
.350	-.099	.135	-.165	.135	-.174	.144	-.169	.164	-.136	.139	-.124	-.109	.011	.400	.450	
.400	-.123	.114	-.180	.127	-.209	.124	-.206	.149	-.173	.128	-.149	.110	-.138	.054	.500	
.450	-.123	.114	-.180	.127	-.209	.124	-.206	.149	-.173	.128	-.149	.110	-.138	.054	.550	
.500	-.142	.128	-.197	.108	-.223	.109	-.222	.130	-.185	.120	-.168	.102	-.164	.064	.600	
.600	-.160	.091	-.194	.097	-.233	.090	-.231	.114	-.194	.177	-.093	-.183	.087	.700	.750	
.700	-.150	.064	-.190	.085	-.232	.073	-.232	.096	-.195	.092	-.181	.087	-.189	.079	.800	
.800	-.147	.070	-.190	.064	-.217	.064	-.236	.079	-.195	.092	-.181	.087	-.189	.079	.900	
.900	-.147	.070	-.190	.064	-.217	.064	-.236	.079	-.195	.092	-.181	.087	-.189	.079	.950	
$\alpha = .08$																
.0125	.293	-.123	.326	-.136	.341	-.189	.345	-.180	.315	-.151	.301	-.054	.331	-.003	.0125	
.025	.181	-.117	.238	-.132	.251	-.188	.228	-.181	.187	-.162	.167	-.058	.158	-.004	.025	
.050	.082	-.111	.120	-.128	.150	-.213	.183	-.187	.120	-.162	.167	-.058	.158	-.004	.050	
.075	.036	-.027	.037	-.122	.061	-.210	.072	-.210	.023	.028	.028	-.036	.077	-.005	.075	
.100	.009	.077	-.003	.003	.014	-.060	.023	-.128	.075	-.009	.022	.028	-.036	.077	.100	
.150	-.042	.103	-.063	.136	-.043	.121	-.016	.075	-.009	.022	.028	-.036	.077	.150	.200	
.200	-.068	.127	-.098	.158	-.097	.175	-.085	.158	-.010	.075	-.009	.020	-.031	.026	.250	
.250	-.061	.123	-.128	.168	-.121	.186	-.109	.187	-.047	.137	-.051	.020	-.031	.026	.300	
.300	-.089	.172	-.184	.184	-.163	.191	-.139	.216	-.119	.222	-.119	.086	-.107	.057	.350	
.350	-.128	.173	-.189	.177	-.200	.191	-.186	.211	-.168	.195	-.136	.140	.082	.400	.450	
.400	-.143	.154	-.203	.167	-.229	.159	-.223	.188	-.202	.172	-.183	.154	-.172	.088	.500	
.450	-.143	.154	-.218	.149	-.246	.143	-.238	.171	-.216	.166	-.203	.161	-.190	.088	.550	
.500	-.168	.159	-.218	.149	-.246	.143	-.238	.171	-.216	.166	-.203	.161	-.190	.088	.600	
.600	-.181	.121	-.214	.135	-.257	.127	-.248	.157	-.221	.121	-.210	.156	-.205	.087	.700	
.700	-.181	.094	-.211	.120	-.250	.105	-.248	.128	-.226	.128	-.213	.151	-.205	.081	.800	
.800	-.170	.105	-.201	.108	-.260	.105	-.248	.111	-.235	.121	-.221	.197	-.218	.164	.850	
.900	-.162	.105	-.201	.108	-.260	.105	-.248	.111	-.235	.121	-.221	.189	-.211	.141	.900	
.950	-.173	.147	-.196	.142	-.264	.143	-.264	.151	-.235	.178	-.225	.189	-.211	.141	.950	
$\alpha = .10$																
.0125	.262	-.074	.293	-.083	.307	-.116	.313	-.046	.288	-.093	.264	-.102	.295	-.106	.0125	
.025	.146	-.065	.204	-.077	.224	-.109	.193	-.039	.288	-.093	.264	-.102	.295	-.106	.025	
.050	.057	-.039	.099	-.073	.125	-.160	.156	-.049	.130	-.127	.103	-.123	.098	-.075	.050	
.075	.010	.084	.013	-.021	.037	-.073	.027	-.021	.021	-.014	.114	-.020	.130	-.033	.100	
.100	-.014	.122	-.039	.119	-.023	.085	-.007	.021	.021	-.014	.114	-.020	.130	-.033	.110	
.150	-.053	.143	-.085	.196	-.011	.202	-.056	.117	-.014	.114	-.020	.130	-.033	.110	.150	
.200	-.080	.175	-.125	.209	-.123	.234	-.107	.179	-.074	.178	-.072	.156	-.048	.155	.200	
.250	-.079	.165	-.142	.224	-.148	.246	-.128	.234	-.074	.178	-.072	.156	-.048	.155	.300	
.300	-.098	.226	-.232	.182	-.246	-.164	.282	-.141	.229	-.137	.182	-.123	.193	-.350	.400	
.350	-.139	.224	-.206	.222	-.219	.242	-.207	.272	-.186	.228	-.221	-.165	.208	-.450	.500	
.400	-.152	.200	-.217	.209	-.245	.212	-.238	.246	-.214	.215	-.200	.211	-.190	.193	.550	
.450	-.152	.200	-.227	.191	-.258	.193	-.253	.219	-.225	.217	-.216	.205	-.207	.178	.600	
.500	-.174	.209	-.227	.191	-.269	.175	-.262	.207	-.235	.212	-.221	.197	-.218	.164	.650	
.600	-.174	.209	-.219	.159	-.264	.143	-.264	.151	-.235	.178	-.225	.189	-.211	.141	.700	
.700	-.185	.175	-.229	.174	-.269	.175	-.262	.207	-.235	.178	-.225	.189	-.211	.141	.750	
.800	-.174	.137	-.219	.159	-.264	.143	-.264	.151	-.235	.178	-.225	.189	-.211	.141	.800	
.900	-.173	.147	-.196	.142	-.264	.143	-.264	.151	-.235	.178						

TABLE II.- Continued

(b) M = 2.01 - Continued

TABLE II.- Concluded
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED WING
(b) $M = 2.01$ - Concluded

x/c , nominal	C_p at $y/b = \frac{1}{2}$ of :														x/c , nominal	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 20^\circ$																
.0125	.081	.177	.095	.347	.093	.457	.027	.575							.0125	
.0250	-.040	.209	.011	.385	.011	.488	-.079	.552	-.068	.595	-.113	.596	-.173		.0250	
.0500	-.114	.279	-.080	.414	-.080	.472	-.100	.541							.0500	
.0750	-.141	.327	-.124	.456	-.128	.515									.0750	
.1000	-.156	.347	-.172	.468	-.161	.526	-.159	.536							.1000	
.1500	-.172	.380	-.219	.468	-.200	.522	-.198	.552	-.194	.567	-.198	.541	-.221		.1500	
.2000	-.182	.420	-.248	.491	-.236	.530	-.238	.530							.2000	
.2500	-.199	.420	-.261	.499	-.254	.522	-.250	.529	-.232	.563	-.236	.493	-.241		.2500	
.3000	-.185	.489		.498	-.272	.509	-.273	.529							.3000	
.3500															.3500	
.4000	-.221	.495	-.298	.477	-.296	.491	-.293	.511							.4000	
.4500															.4500	
.5000	-.231	.470	-.311	.465	-.312	.446	-.305	.471							.5000	
.5500															.5500	
.6000	-.233	.421	-.298	.429	-.317	.417	-.315	.447							.6000	
.6500															.6500	
.7000	-.250	.372	-.296	.404	-.317	.395	-.319	.421							.7000	
.7500															.7500	
.8000	-.234	.359	-.298	.385		.368	-.285	.390							.8000	
.8500															.8500	
.9000	-.229	.380	-.295	.364	-.271	.358	-.266	.366							.9000	
.9500															.9500	

TABLE III
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING

TABLE III.—Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING

(a) M = 1.61 - Continued

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING
(a) $M = 1.61$ - Continued

x/c	Cp at $y/\frac{c}{2}$ of :														x/c	
	.05		.20		.35		.50		.70		.825		.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -04$																
.0125	.395	-.219	.413	-.395	.418	-.433	.470	-.436	.352	-.425	.335	-.434	-.074	-.386	.0125	
.025	.358	-.182	.336	-.357	.362	-.419	.343	-.425	.352	-.379	.244	-.382	.235	-.392	.025	
.050	.252	-.174	.279	-.357	.293	-.384	.280	-.401	.380	-.355	.115	-.339	.094	-.361	.050	
.075	.239	-.107	.233	-.334	.228	-.364	.106	-.332	.115	-.339	.094	-.361	.109	-.354	.075	
.100	.209	-.080	.136	-.098	.118	-.359	.192	-.299	.115	-.279	.081	-.206	.052	-.230	.100	
.125	.126	-.065	.045	-.016	.049	-.176	.029	-.125	.057	-.293	.027	-.289	.070	-.287	.125	
.200	.094	-.020	.002	-.000	.034	-.016	-.006	-.012	.057	-.293	.027	-.289	.070	-.287	.200	
.250	.039	-.057	.002	-.000	.034	-.016	-.006	-.012	.057	-.293	.027	-.289	.070	-.287	.250	
.300	.026	-.033	.043	-.083	.073	-.088	.056	-.147	.005	-.139	.0207	.143	-.210	.300	.300	
.350	-.037	.033	-.130	.069	-.162	.072	-.170	.081	-.206	.045	-.204	.007	-.211	.167	.400	
.400	-.056	.022	-.123	.042	-.204	.055	-.214	.052	-.220	.005	-.222	.022	-.230	.167	.500	
.500	-.056	-.019	-.077	-.040	-.143	-.034	-.185	-.033	-.222	-.022	-.230	.167	-.230	.600	.600	
.600	-.056	-.019	-.077	-.040	-.143	-.034	-.185	-.033	-.222	-.022	-.230	.167	-.230	.700	.700	
.650	-.028	-.101	-.142	.004	-.128	.024	-.142	.013	-.192	-.069	-.210	.136	-.228	.750	.750	
.700	.001	-.150	.089	-.227	.116	-.229	.114	-.222	.107	-.268	.086	-.285	.034	-.290	.800	.800
.800	.097	-.192	.168	-.272	.194	-.283	.186	-.276	-.000	-.000	-.000	-.000	-.000	-.000	.900	.900
.900	.097	-.192	.168	-.272	.194	-.283	.186	-.276	-.000	-.000	-.000	-.000	-.000	-.000	.950	.950
$\alpha = -02$																
.0125	.329	-.154	.351	-.352	.360	-.332	.303	-.339	.309	-.310	.282	-.313	-.120	.0125	.0125	
.025	.301	-.131	.286	-.309	.300	-.297	.246	-.255	.235	-.252	.192	-.255	-.120	.025	.025	
.050	.202	-.131	.231	-.294	.235	-.252	.192	-.255	.240	-.276	.184	-.247	.192	-.214	.050	
.075	.160	-.043	.188	-.274	.174	-.223	.136	-.240	.203	-.276	.184	-.247	.192	-.214	.075	
.100	.166	-.043	.188	-.274	.174	-.223	.136	-.240	.203	-.276	.184	-.247	.192	-.214	.100	
.125	.097	-.031	.079	-.230	.136	-.218	.094	-.175	.027	-.004	.066	-.234	.046	-.142	.150	
.200	.052	.015	.001	-.012	.012	-.027	.059	-.027	.027	-.004	.066	-.234	.046	-.142	.200	
.250	.008	.083	-.042	.064	-.078	.042	-.087	.087	-.000	-.008	.070	-.115	.102	-.103	.250	
.300	-.012	-.083	.098	-.125	.179	-.168	.145	-.179	.056	-.170	.057	-.192	.071	-.171	.300	
.350	-.065	.066	-.176	.101	-.187	.174	-.244	.120	-.242	.134	-.238	.022	-.243	.053	.400	
.400	-.083	.057	-.161	.072	-.243	.128	-.285	.120	-.257	.056	-.253	.024	-.259	.071	.500	
.500	-.090	.016	-.117	-.009	-.198	.029	-.254	.015	-.257	.056	-.253	.024	-.259	.071	.600	
.600	-.069	-.071	-.116	-.053	-.073	-.184	-.112	-.214	-.039	-.223	-.073	-.244	-.113	-.244	.650	
.700	-.069	-.071	-.116	-.053	-.073	-.184	-.112	-.214	-.039	-.223	-.073	-.244	-.113	-.244	.700	
.750	-.030	-.129	.047	-.204	.061	-.188	-.000	-.198	-.128	-.155	-.145	-.146	-.174	-.192	.750	
.800	-.030	-.129	.047	-.204	.061	-.188	-.000	-.198	-.128	-.155	-.145	-.146	-.174	-.192	.800	
.850	-.059	-.162	.120	-.250	.143	-.241	.057	-.257	-.030	-.239	-.031	-.222	-.065	-.242	.850	
.900	-.059	-.162	.120	-.250	.143	-.241	.057	-.257	-.030	-.239	-.031	-.222	-.065	-.242	.900	
.950	-.059	-.162	.120	-.250	.143	-.241	.057	-.257	-.030	-.239	-.031	-.222	-.065	-.242	.950	
$\alpha = 00$																
.0125	.264	-.099	.279	-.266	.312	-.294	.296	-.330	.261	-.261	.236	-.092	-.023	.0125	.0125	
.025	.241	-.083	.228	-.215	.248	-.292	.240	-.296	.261	-.261	.282	-.153	-.023	.025	.025	
.050	.159	-.068	.177	-.220	.191	-.250	.189	-.253	.240	-.276	.184	-.247	.192	-.214	.050	
.075	.149	-.013	.137	-.157	.119	-.220	.136	-.243	.156	-.074	.146	-.074	.146	-.024	.075	
.100	.116	-.003	.003	-.039	.038	-.190	.089	-.176	.018	-.035	.007	.159	.032	.048	.100	
.125	.053	-.003	.034	-.023	.060	-.022	.018	-.019	.019	-.035	.007	.159	.032	.048	.150	
.200	.013	.043	-.040	.083	-.040	.133	-.059	.060	-.048	-.004	.119	.140	-.136	.129	.200	
.250	-.026	.109	-.080	.117	.120	-.163	-.093	.170	-.048	-.004	.119	.140	-.136	.129	.250	
.300	-.048	-.123	.149	-.166	.182	-.172	.248	-.220	.244	-.215	.125	-.235	.152	.300	.300	
.350	-.121	.110	-.205	.143	-.237	.177	-.250	.195	-.277	.202	-.277	.102	-.283	.153	.400	
.400	-.119	.099	-.205	.112	-.267	.128	-.291	.139	-.293	.093	-.292	.055	-.291	.074	.450	
.500	-.142	.056	-.150	.024	-.254	.030	-.260	.032	-.293	.093	-.292	.055	-.291	.074	.500	
.600	-.098	-.030	-.083	-.107	-.074	-.192	-.089	-.260	-.018	-.268	-.030	-.282	-.039	-.282	.600	
.700	-.098	-.030	-.083	-.107	-.074	-.192	-.089	-.260	-.018	-.268	-.030	-.282	-.039	-.282	.700	
.750	-.034	-.117	.011	-.172	.006	-.190	-.010	-.182	-.077	-.224	-.104	-.197	-.060	-.247	.750	
.800	-.034	-.117	.011	-.172	.006	-.190	-.010	-.182	-.077	-.224	-.104	-.197	-.060	-.247	.800	
.850	-.035	-.146	.088	-.227	.088	-.242	.052	-.243	-.145	-.203	-.158	-.214	-.109	-.237	.850	
.900	-.035	-.146	.088	-.227	.088	-.242	.052	-.243	-.145	-.203	-.158	-.214	-.109	-.237	.900	
.950	-.035	-.146	.088	-.227	.088	-.242	.052	-.243	-.145	-.203	-.158	-.214	-.109	-.237	.950	
$\alpha = .02$																
.0125	.211	-.042	.213	-.161	.255	-.236	.229	-.183	.203	-.203	.162	.107	-.176	.025	.0125	
.025	.189	-.036	.159	-.110	.196	-.175	.179	-.165	.165	-.202	.096	.200	-.056	.219	.025	
.050	.124	.001	.126	-.097	.146	-.144	.140	-.157	.111	-.202	.096	.200	-.056	.219	.050	
.075	.109	.016	.086	-.008	.075	-.091	.040	-.067	.111	-.202	.096	.200	-.056	.219	.075	
.100	.077	.031	.045	-.015	.045	-.005	.040	-.013	.111	-.202	.096	.200	-.056	.219	.100	
.125	.015	.031	-.009	.063	-.019	.127	-.027	.188	-.036	.252	-.018	.227	-.034	.236	.150	
.200	-.013	.078	-.082	.138	-.088	.191	-.100	.225	-.084	.252	-.165	.224	-.162	.255	.200	
.250	-.046	.137	-.115	.164	-.149	.214	-.137	.235	-.084	.252	-.165	.224	-.162	.255	.250	
.300	-.081	-.157	.192	-.197	.225	-.205	.267	-.267	.250	-.222	-.243	.218	-.267	.234	.300	
.350	-.145	.145	-.230	.183	-.265	.208	-.278	.233	-.300	-.202	-.302	.188	-.324	.197	.400	
.400	-.145	.145	-.230	.183	-.265	.208	-.278	.233	-.300	-.202	-.302	.188	-.324	.197	.450	
.500	-.143	.134	-.241	.148	-.285	.160	-.322	.180	-.315	.115	.312	.113	-.334	.105	.500	
.600	-.157	.084	-.180	.059	-.276	.059	-.300	.072	-.290	.010	.300	-.016	-.309	-.032	.600	
.700	-.126	-.004	-.050	-.149	-.051	-.237	-.068	-.224	-.112	-.237	-.129	-.225	-.170	-.225	.700	
.750	-.061															

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING

(a) $M = 1.61$ - Continued

x/c	C_p at $y/\frac{c}{2}$ of:														x/c	
	.05		.20		.35		.50		.70		.825		.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = .04$																
.0125	.147	.020	.139	-.057	.178	-.047	.144	.052	.090		.031	.288		.306	.0125	
.025	.036	.020	.112	-.004	.129	-.003	.107	.066							.025	
.050	.052	.080	.034	.088	.042	.087	.097	.094							.050	
.075	.049	.039	.039	.021	.053	.197									.075	
.100	.041	.058	.061	-.008	.113	-.011	.213								.100	
.125	-.014	.066	-.049	.116	-.065	.201									.125	
.200	-.039	.113	-.120	.179	-.126	.249									.200	
.250	-.053	.154	-.149	.215	-.192	.266									.250	
.300	-.133		-.182	.238	-.231	.272									.300	
.350															.350	
.400	-.173	.192	-.257	.225	-.294	.251									.400	
.450															.450	
.500	-.172	.179	-.281	.187	-.315	.197									.500	
.550															.550	
.600	-.175	.122	-.213	.092	-.304	.088									.600	
.650															.650	
.700	-.145	.032		-.023	-.201	-.021									.700	
.750															.750	
.800	-.084	-.077	-.053	-.120	-.084	-.140									.800	
.850															.850	
.900	-.027	-.098	.022	-.182	-.015	-.188									.900	
.950															.950	
$\alpha = .06$																
.0125	.078	.080	.037	.075	.072	.132	.000	.244							.0125	
.025	.078	.075	.049	.100	.042	.133									.025	
.050	.056	.101	.037	.110	.027	.160									.050	
.075	.032	.092	-.014	.104	-.020	.181									.075	
.100	.012	.101		.120	-.074	.203									.100	
.125	-.044	.096	-.084	.162	-.107	.269									.125	
.200	-.068	.157	-.151	.242	-.167	.303									.200	
.250	-.071	.209	-.192	.266	-.226	.320									.250	
.300	-.150		-.220	.284	-.265	.317									.300	
.350															.350	
.400	-.199	.243	-.274	.270	-.321	.288									.400	
.450															.450	
.500	-.198	.225	-.321	.228	-.353	.234									.500	
.550															.550	
.600	-.203	.163	-.242	.126	-.319	.123									.600	
.650															.650	
.700	-.169	.065		.013	-.252	.007									.700	
.750															.750	
.800	-.099	-.046	-.042	-.096	-.119	-.110									.800	
.850															.850	
.900	-.042	-.072		-.007	-.154	-.061									.900	
.950															.950	
$\alpha = .08$																
.0125	.004	.149	-.099	.183	-.073	.281	-.186	.378							.0125	
.025	.031	.130	-.048	.186	-.066	.258	-.172	.344							.025	
.050	.022	.145	-.016	.182	-.050	.275	-.123	.359							.050	
.075	.005	.137		.060	.168	-.063	.273								.075	
.100	-.017				.185	-.115	.285	-.095	.358						.100	
.125	-.068	.135	-.124	.229	-.154	.329	-.117	.363							.125	
.200	-.094	.199	-.183	.298	-.200	.352	-.239	.374							.200	
.250	-.094	.261	-.219	.319	-.257	.363	-.242	.375							.250	
.300	-.164		-.253	.333	-.292	.358	-.316	.395							.300	
.350															.350	
.400	-.215	.291	-.286	.316	-.345	.330	-.361	.338							.400	
.450															.450	
.500	-.216	.269	-.341	.269	-.380	.273	-.396	.268							.500	
.550															.550	
.600	-.221	.204	-.272	.166	-.355	.157	-.387	.157							.600	
.650															.650	
.700	-.183	.104		.053	-.281	.039	-.355	.014							.700	
.750															.750	
.800	-.117	-.016	-.112	-.066	-.155	-.078	-.235	-.077							.800	
.850															.850	
.900	-.064	-.044	-.034	-.116	-.100	-.134	-.142	-.139							.900	
.950															.950	
$\alpha = .10$																
.0125	-.083	.219	-.205	.274	-.215	.399	-.304	.474							.0125	
.025	-.031	.192	-.184	.265	-.192	.365	-.270	.421							.025	
.050	-.043	.166	-.140	.245	-.190	.363	-.230	.423							.050	
.075	-.031	.175	-.126	.235	-.188	.345	-.177	.417							.075	
.100	-.101	.187		.254	-.195	.351	-.254	.419							.100	
.125	-.101	.173	-.167	.307	-.206	.384	-.262	.413							.125	
.200	-.125	.244	-.222	.349	-.241	.401	-.292	.422							.200	
.250	-.125	.314	-.253	.369	-.294	.409	-.315	.417							.250	
.300	-.192		-.293	.379	-.321	.401	-.355	.432							.300	
.350															.350	
.400	-.241	.332	-.305	.361	-.374	.374	-.393	.375							.400	
.450															.450	
.500	-.247	.312	-.358	.315	-.400	.307	-.421	.301							.500	
.550															.550	
.600	-.247	.246	-.301	.203	-.394	.187	-.412	.185							.600	
.650															.650	
.700	-.207	.139		.090	-.306	.078	-.382	.051							.700	
.750															.750	
.800	-.140	.015	-.150	-.034	-.207	-.051	-.299	-.058							.800	
.850															.850	
.900	-.087	-.014	-.069	-.084	-.153	-.102	-.176	-.117							.900	
.950															.950	

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING

X/C	Cp at $\frac{V}{2}$ of:														X/C	
	.05		.20		.35		.50		.70		.825		.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 12$																
.0125	-+.148	+.284	-.259	-.351	-.307	+.487	-.377	+.538	-.429		-.450	+.539			+.493	.0125
.025	-.088	+.250	-.260	-.335	-.288	+.453	-.378	+.476	-.429		-.418	+.476			+.050	.025
.050	-.087	+.238	-.234	-.304	-.272	+.431	-.345	+.482							+.075	.050
.075	-.061	+.223	-.224	-.310	-.266	+.414	-.345	+.471	-.388	+.499	-.418	+.476			+.100	.075
.100	-.082	+.234	-.220	-.318	-.270	+.414	-.325	+.476							+.125	.100
.150	-.130	+.221	-.210	-.373	-.291	+.443	-.331	+.462	-.385	+.477	-.411	+.453	-.440	+.444	+.150	.125
.200	-.149	+.295	-.246	+.400	-.320	+.449	-.355	+.470							+.200	.150
.250	-.158	+.369	-.279	+.420	-.355	+.453	-.371	+.459	-.393	+.484	-.435	+.402	-.447	+.425	+.250	.200
.300	-.209	+.315	-.428	-.381	-.440	-.398	-.463								+.300	.250
.350															+.400	.300
.400															+.400	.350
.450															+.500	.400
.500															+.500	.450
.550															+.550	.500
.600															+.600	.550
.650															+.650	.600
.700															+.700	.650
.750															+.750	.700
.800															+.800	.750
.850															+.850	.800
.900															+.900	.850
.950															+.950	.900
$a = 14$																
.0125	-.203	+.358	-.316	+.416	-.377	+.566	-.441	+.612	-.471		-.481	+.569			+.553	.0125
.025	-.151	+.321	-.318	+.413	-.359	+.533	-.430	+.554	-.471		-.468	+.518			+.050	.025
.050	-.151	+.290	-.300	+.387	-.344	+.508	-.412	+.573	-.441	+.570	-.468	+.518			+.075	.050
.075	-.102	+.280	-.293	+.385	-.334	+.496	-.354	+.554	-.441	+.570	-.468	+.518			+.100	.075
.100	-.114	+.288	-.291	+.398	-.334	+.496	-.391	+.560	-.436	+.542	-.447	+.495	-.448	+.518	+.125	.100
.150	-.166	+.280	-.291	+.435	-.345	+.530	-.390	+.544	-.441	+.516	-.464	+.466	-.443	+.507	+.150	.125
.200	-.177	+.353	-.298	+.467	-.367	+.526	-.405	+.541	-.471	+.522	-.484	+.474	-.431	+.475	+.200	.150
.250	-.189	+.431	-.307	+.480	-.395	+.531	-.415	+.522	-.439	+.530	-.472	+.476	-.424	+.427	+.250	.200
.300	-.218	+.336	-.494	-.417	-.522										+.300	.250
.350															+.400	.350
.400															+.400	.350
.450															+.500	.400
.500															+.500	.400
.550															+.550	.400
.600															+.600	.400
.650															+.650	.400
.700															+.700	.400
.750															+.750	.400
.800															+.800	.400
.850															+.850	.400
.900															+.900	.400
.950															+.950	.400
$a = 16$																
.0125	-.252	+.436	-.360	+.502	-.434	+.679	-.467	+.657	-.479		-.488	+.624			+.588	.0125
.025	-.205	+.396	-.360	+.506	-.424	+.657	-.461	+.698	-.479		-.488	+.608			+.593	.025
.050	-.224	+.351	-.344	+.472	-.412	+.637	-.453	+.637	-.472	+.616	-.484	+.593			+.675	.050
.075	-.139	+.342	-.338	+.480	-.400	+.632	-.429	+.622	-.471	+.591	-.496	+.564			+.700	.075
.100	-.143	+.349	-.346	+.524	-.397	+.694	-.447	+.626	-.482	+.591	-.516	+.544			+.750	.100
.150	-.183	+.364	-.352	+.568	-.439	+.620	-.446	+.609	-.466	+.595	-.526	+.511			+.800	.150
.200	-.197	+.514	-.366	+.598	-.411	+.610	-.445	+.596	-.471	+.514	-.542	+.466			+.850	.200
.250	-.209	+.510	-.370	+.605	-.429	+.605	-.457	+.574	-.471	+.591	-.529	+.464			+.900	.250
.300	-.230	+.510	-.369	+.610	-.445	+.684	-.468	+.580	-.488	+.545	-.513	+.411			+.950	.300
.350															+.350	.300
.400															+.400	.350
.450															+.450	.400
.500															+.500	.400
.550															+.550	.400
.600															+.600	.400
.650															+.650	.400
.700															+.700	.400
.750															+.750	.400
.800															+.800	.400
.850															+.850	.400
.900															+.900	.400
.950															+.950	.400
$a = 18$																
.0125	-.282	+.498	-.406	+.635	-.468	+.728	-.473	+.681	-.479		-.437	+.631			+.592	.0125
.025	-.242	+.454	-.403	+.642	-.459	+.718	-.469	+.639	-.479		-.437	+.634			+.650	.025
.050	-.264	+.396	-.385	+.629	-.454	+.705	-.464	+.693	-.479		-.437	+.634			+.700	.050
.075	-.174	+.392	-.377	+.625	-.443	+.692	-.468	+.688	-.480		-.437	+.634			+.750	.075
.100	-.169	+.397	-.361	+.631	-.439	+.661	-.458	+.669	-.482		-.437	+.634			+.800	.100
.150	-.196	+.450	-.387	+.645	-.478	+.680	-.488	+.665	-.493		-.437	+.634			+.850	.150
.200	-.209	+.538	-.405	+.663	-.488	+.663	-.482	+.649	-.505		-.437	+.634			+.900	.200
.250	-.224	+.638	-.409	+.669	-.461	+.653	-.464	+.622	-.505		-.437	+.634			+.950	.250
.300	-.242	+.415	-.461	+.661	-.473	+.629	-.467	+.637	-.505		-.437	+.634			+.950	.300
.350															+.350	.300
.400															+.400	.350
.450															+.450	.400
.500															+.500	.400
.550															+.550	.400
.600															+.600	.400
.650															+.650	.400
.700															+.700	.400
.750															+.750	.400
.800															+.800	.400
.850															+.850	.400
.900															+.900	.400
.950															+.950	.400

TABLE III.—Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING

(a) M = 1.61 - Concluded

x/c	Cp at $\frac{\gamma}{2}$ of :												x/c	
	+.05		+.20		+.35		+.50		+.70		+.85			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower
a = 20														
.0125	-.312	.582	-.457	.722	-.478	.771	-.455	.714	-.401		-.360	.636		.0125
.025	-.275	.555	-.444	.738	-.468	.776	-.445	.679	-.401		-.360	.636		.025
.050	-.285	.506	-.438	.771	-.460	.770	-.455	.747	-.396		-.355	.650		.050
.075	-.213	.514	-.428	.716	-.469	.756	-.451	.751	-.396		-.355	.650		.075
.100	-.192	.528	-.420	.711	-.467	.731	-.452	.727	-.392		-.343	.642		.100
.150	-.209	.596	-.430	.721	-.470	.737	-.448	.726	-.392		-.343	.642		.150
.200	-.220	.665	-.440	.739	-.465	.719	-.445	.709	-.390		-.343	.610		.200
.250	-.241	.711	-.447	.731	-.459	.707	-.443	.687	-.390		-.343	.589		.250
.300	-.248		-.454	.729	-.454	.688	-.440	.697	-.385		-.339	.565		.300
.350									-.378		-.331	.526		.350
.400	-.312	.724	-.444	.685	-.461	.681	-.434	.651	-.378		-.331	.526		.400
.450									-.371		-.327	.459		.450
.500	-.321	.661	-.428	.605	-.452	.584	-.430	.579	-.371		-.327	.459		.500
.550									-.365		-.323	.515		.550
.600	-.308	.562	-.414	.580	-.449	.471	-.424	.465	-.365		-.323	.515		.600
.650									-.360		-.320	.496		.650
.700	-.282	.418		.350	-.443	.350	-.416	.320	-.360		-.320	.496		.700
.750									-.355		-.316	.462		.750
.800	-.201	.303	-.356	.243	-.436	.226	-.412	.232	-.355		-.316	.462		.800
.850									-.350		-.316	.430		.850
.900	-.125	.263	-.256	.223	-.415	.215	-.407	.188	-.350		-.316	.430		.900
.950									-.345		-.316	.400		.950

TABLE III.- Continued

PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING

(b) $M = 2.01$

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING
(b) M = 2.01 - Continued

x/c	Cp at $y/\frac{c}{2}$ of:														x/c	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -12^\circ$																
.0125	.560	-.259	.586	-.299	.585	-.312	.557	-.312	.509	-.290	.491	-.289	.488	-.288	.0125	
.025	.528	-.236	.525	-.287	.531	-.312	.520	-.306	.506	-.296	.491	-.289	.488	-.288	.025	
.050	.439	-.246	.464	-.286	.474	-.307	.474	-.306	.466	-.296	.456	-.289	.450	-.289	.050	
.075	.404	-.233	.427	-.283	.424	-.305	.404	-.303	.429	-.290	.429	-.287	.423	-.299	.075	
.100	.363	-.218	.382	-.289	.382	-.295	.396	-.274	.302	-.288	.308	-.283	.292	-.292	.100	
.150	.292	-.212	.320	-.280	.306	-.300	.307	-.293	.243	-.308	.298	-.288	.288	-.288	.150	
.200	.251	-.101	.243	-.236	.253	-.293	.193	-.298	.266	-.288	.193	-.283	.184	-.288	.200	
.250	.199	-.087	.183	-.204	.165	-.247	.125	-.232	.119	-.270	.079	-.278	.090	-.283	.195	-.258
.300	.195	-.052	.137	-.187	.125	-.232	.125	-.232	.119	-.270	.079	-.278	.090	-.283	.095	-.350
.350	.099	-.072	.042	-.165	.045	-.194	.047	-.221	.021	-.230	.023	-.256	.035	-.227	.400	-.450
.400	.080	-.098	.033	-.151	-.004	-.154	-.005	-.210	.001	-.223	.006	-.228	.022	-.227	.500	-.500
.550	.095	-.113	.063	-.152	.044	-.171	.014	-.226	.048	-.245	.061	-.251	.067	-.246	.650	-.650
.600	.114	-.153	-.181	-.130	-.211	-.096	-.259	.170	-.270	.173	-.275	.197	-.272	.750	-.750	
.700	.164	-.181	.220	-.221	.235	-.251	.245	-.286	.303	-.271	.291	-.273	.286	-.269	.800	-.850
.800	.227	-.212	.284	-.246	.311	-.270	.325	-.259	-.271	-.271	-.271	-.271	-.271	-.271	.900	-.900
.900	.227	-.212	.284	-.246	.311	-.270	.325	-.259	-.271	-.271	-.271	-.271	-.271	-.271	.950	-.950
$\alpha = -10^\circ$																
.0125	.514	-.224	.555	-.289	.558	-.316	.526	-.316	.481	-.288	.463	-.285	.462	-.280	.0125	
.025	.479	-.208	.486	-.270	.487	-.308	.487	-.307	.481	-.288	.463	-.285	.462	-.280	.050	
.050	.380	-.212	.427	-.264	.432	-.298	.431	-.297	.393	-.280	.394	-.278	.394	-.289	.075	
.075	.354	-.199	.386	-.266	.386	-.298	.355	-.266	.280	-.275	.275	-.275	.285	-.285	.100	
.100	.319	-.180	.347	-.247	.354	-.291	.355	-.280	.290	-.275	.275	-.275	.200	-.200	.150	
.150	.250	-.146	.276	-.256	.267	-.280	.271	-.280	.266	-.280	.259	-.277	.255	-.274	.250	
.200	.208	-.084	.202	-.227	.209	-.249	.208	-.266	.194	-.287	.136	-.259	.215	-.264	.300	
.250	.161	-.060	.147	-.152	.126	-.245	.136	-.259	.228	-.277	.155	-.274	.142	-.264	.350	
.300	.159	-.039	.097	-.096	.084	-.216	.076	-.231	.041	-.224	.052	-.250	.062	-.213	.400	
.400	.066	-.062	.012	-.066	.017	-.162	.001	-.180	-.017	-.183	-.015	-.200	-.002	-.192	.450	
.500	.050	-.076	-.004	-.087	-.039	-.115	-.036	-.171	-.037	-.187	-.031	-.190	-.020	-.197	.550	
.600	.058	-.098	.027	-.106	-.005	-.126	-.011	-.192	-.008	-.215	-.017	-.218	.018	-.223	.650	
.700	.081	-.130	-.155	-.079	-.174	-.044	-.234	-.099	-.253	-.101	-.258	.114	-.257	.750	-.800	
.750	.123	-.167	.176	-.206	.188	-.225	.177	-.268	.220	-.274	.221	-.278	.199	-.274	.850	
.800	.193	-.193	.246	-.233	.263	-.254	.263	-.271	-.271	-.271	-.271	-.271	-.271	-.271	.900	
.900	.193	-.193	.246	-.233	.263	-.254	.263	-.271	-.271	-.271	-.271	-.271	-.271	-.271	.950	
$\alpha = -08^\circ$																
.0125	.483	-.203	.498	-.292	.511	-.311	.492	-.315	.446	-.308	.429	-.291	.429	-.263	.0125	
.025	.409	-.190	.433	-.266	.456	-.281	.449	-.308	.481	-.278	.454	-.271	.454	-.250	.050	
.050	.322	-.196	.369	-.253	.384	-.291	.392	-.296	.356	-.276	.394	-.273	.354	-.271	.100	
.075	.303	-.186	.332	-.254	.330	-.281	.307	-.250	.356	-.276	.394	-.273	.354	-.265	.150	
.100	.268	-.158	-.259	-.259	.293	-.278	.207	-.256	.222	-.267	.240	-.276	.235	-.265	.200	
.150	.201	-.126	.217	-.246	.207	-.266	.152	-.238	.153	-.258	.184	-.255	.112	-.242	.094	
.200	.152	-.076	.142	-.211	.157	-.238	.074	-.219	.108	-.231	.184	-.255	.124	-.225	.250	
.250	.109	-.051	.092	-.121	.074	-.219	.037	-.219	.188	-.218	.184	-.255	.112	-.242	.300	
.300	.111	-.037	.044	-.062	.029	-.185	.034	-.195	-.004	-.172	.011	-.184	.019	-.180	.350	
.400	.021	-.039	-.027	-.035	-.046	-.113	-.039	-.147	-.057	-.142	-.057	-.146	-.036	-.159	.400	
.500	.003	-.055	-.056	-.047	-.075	-.058	-.075	-.136	-.075	-.153	-.073	-.147	-.058	-.165	.550	
.600	.018	-.081	-.018	-.086	-.071	-.091	-.062	-.163	-.039	-.186	-.031	-.187	-.028	-.198	.650	
.700	.037	-.127	-.144	-.118	-.213	-.095	-.227	-.133	-.269	-.152	-.269	-.124	-.262	-.080	.700	
.800	.076	-.162	.117	-.206	.118	-.213	.095	-.227	-.133	-.269	-.152	-.269	-.124	-.262	.850	
.900	.146	-.190	.188	-.231	.198	-.242	.200	-.252	-.133	-.207	-.080	-.231	.116	-.235	.071	-.230
.950	.112	-.156	.152	-.184	.150	-.217	.133	-.207	-.080	-.231	.116	-.235	.071	-.230	.950	
$\alpha = -06^\circ$																
.0125	.509	-.151	.453	-.258	.474	-.274	.455	-.292	.424	-.282	.410	-.275	.426	-.226	.0125	
.025	.363	-.153	.391	-.251	.414	-.274	.340	-.252	.349	-.248	.327	-.238	.324	-.234	.050	
.050	.282	-.154	.334	-.225	.287	-.212	.287	-.239	.322	-.233	.327	-.238	.324	-.234	.075	
.075	.264	-.148	.292	-.212	.287	-.212	.253	-.212	.264	-.219	.212	-.216	.204	-.223	.100	
.100	.220	-.110	.211	-.202	.162	-.202	.162	-.213	.188	-.222	.204	-.216	.204	-.223	.150	
.150	.164	-.085	.182	-.202	.110	-.189	.114	-.208	.150	-.190	.086	-.185	.061	-.164	.250	
.200	.115	-.048	.107	-.175	.057	-.163	.076	-.180	.150	-.190	.086	-.185	.061	-.164	.300	
.250	.078	-.017	.055	-.115	.037	-.163	.076	-.180	.150	-.190	.086	-.185	.061	-.164	.350	
.300	.078	-.017	.012	-.010	-.003	-.122	-.004	-.135	-.022	-.109	-.007	-.124	-.004	-.130	.350	
.350	-.002	-.004	-.050	-.013	-.076	-.022	-.071	-.078	-.077	-.077	-.071	-.084	-.062	-.107	.400	
.400	-.023	-.026	-.086	-.009	-.114	-.001	-.108	-.066	-.087	-.069	-.062	-.089	-.090	-.113	.500	
.500	-.006	-.048	-.052	-.052	-.098	-.049	-.087	-.069	-.062	-.131	-.060	-.139	-.055	-.153	.600	
.600	-.010	-.096	-.123	-.184	.150	-.217	.133	-.207	.080	-.231	.116	-.235	.071	-.230	.650	
.700	.010	-.139	.084	-.185	.072	-.183	.037	-.171	.080	-.231	.116	-.235	.071	-.230	.700	
.800	.049	-.139	.152	-.184	.150	-.217	.133	-.207	.080	-.231	.116	-.235	.071	-.230	.850	
.900	.112	-.156	.152	-.184	.150	-.217	.133	-.207	.080	-.231	.116	-.235	.071	-.230	.900	
.950	.112	-.156	.152	-.184	.150	-.217	.133	-.207	.080	-.231	.116	-.235	.071	-.230	.950	

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING
(b) $M = 2.01$ - Continued

x/c	C_p at $y/L = 0$														x/c	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -.04$																
.0125	.353	-.116	.407	-.227	.438	-.249	.416	-.270	.386	-.212	.372	-.259	.288	-.214	.0125	
.025	.316	-.132	.341	-.221	.376	-.201	.304	-.218	.310	-.242	.291	-.212	.288	-.214	.025	
.050	.243	-.122	.287	-.201	.304	-.191	.274	-.210	.220	-.201	.220	-.212	.288	-.214	.050	
.075	.225	-.115	.247	-.186	.274	-.166	.206	-.210	.214	-.186	.206	-.212	.288	-.214	.075	
.100	.188	-.078	.140	-.159	.194	-.134	.174	-.215	.220	-.191	.156	-.185	.158	-.194	.100	
.150	.129	-.052	.140	-.159	.159	-.134	.176	-.152	.152	-.195	.156	-.185	.158	-.194	.150	
.200	.083	-.021	.073	-.141	.080	-.066	.156	-.080	.178	-.042	.147	-.104	.158	-.151	.200	
.250	.053	-.014	.015	-.018	.006	-.012	.125	-.042	.147	-.027	.104	-.049	.158	-.151	.250	
.300	.033	-.001	.018	-.041	-.036	-.080	-.027	-.105	-.056	-.027	.056	-.040	.086	-.041	.300	
.350	-.016	-.018	-.018	-.041	-.036	-.080	-.027	-.105	-.056	-.027	.056	-.040	.086	-.041	.350	
.400	-.028	.016	-.075	.049	-.078	.053	-.098	-.027	-.112	-.028	-.104	-.031	-.095	-.062	.400	
.450	-.043	.001	-.106	.030	-.134	.033	-.137	.024	-.125	-.049	-.119	-.047	-.116	-.068	.450	
.500	-.033	-.025	-.075	-.026	-.110	-.025	-.126	-.019	-.101	-.010	-.093	-.107	-.093	-.117	.500	
.550	-.016	-.082	-.106	-.060	-.105	-.084	-.093	-.042	-.163	-.037	-.172	-.039	-.177	-.050	.550	
.600	-.016	-.082	-.106	-.060	-.105	-.084	-.093	-.042	-.163	-.037	-.172	-.039	-.177	-.050	.600	
.650	-.016	-.082	-.106	-.060	-.105	-.084	-.093	-.042	-.163	-.037	-.172	-.039	-.177	-.050	.650	
.700	-.016	-.082	-.106	-.060	-.105	-.084	-.093	-.042	-.163	-.037	-.172	-.039	-.177	-.050	.700	
.750	-.016	-.082	-.106	-.060	-.105	-.084	-.093	-.042	-.163	-.037	-.172	-.039	-.177	-.050	.750	
.800	.021	-.129	.053	-.170	.034	-.172	-.005	-.159	-.042	-.163	-.037	-.172	-.039	-.177	.800	
.850	-.009	-.148	.109	-.205	.108	-.210	.077	-.196	.026	-.213	.075	-.218	.021	-.212	.850	
.900	-.077	-.148	.109	-.205	.108	-.210	.077	-.196	-.027	-.213	-.075	-.218	.021	-.212	.900	
.950	-.050	-.148	.109	-.205	.108	-.210	.077	-.196	-.027	-.213	-.075	-.218	.021	-.212	.950	
$\alpha = -.02$																
.0125	.296	-.071	.350	-.178	.388	-.205	.315	-.224	.367	-.225	.345	-.223	.329	-.223	.0125	
.025	.228	-.108	.295	-.174	.324	-.177	.267	-.190	.295	-.178	.255	-.174	.250	-.174	.025	
.050	.190	-.089	.244	-.145	.261	-.177	.206	-.187	.274	-.178	.250	-.175	.250	-.174	.050	
.075	.172	-.085	.198	-.143	.206	-.177	.206	-.187	.274	-.178	.250	-.175	.250	-.174	.075	
.100	.143	-.047	.132	-.132	.194	-.139	.179	-.154	.205	-.154	.177	-.138	.122	-.144	.100	
.150	.089	-.031	.097	-.108	.104	-.070	.047	-.111	.041	-.120	.021	-.099	.001	-.097	.150	
.200	.046	-.003	.035	-.070	.031	-.050	.031	-.073	.008	-.072	.071	-.108	.017	-.099	.200	
.250	.014	-.044	.015	-.049	.015	-.050	.031	-.073	.008	-.072	.071	-.108	.017	-.099	.250	
.300	.015	-.025	.057	-.083	.068	-.067	.058	-.082	-.086	-.015	-.072	-.021	-.085	-.037	.300	
.350	-.009	-.063	.051	-.110	.081	-.132	.097	-.126	.085	-.133	.044	-.130	.041	-.120	.400	
.400	-.063	.051	-.110	.081	-.132	.097	-.126	.085	-.133	.044	-.130	.041	-.120	.400	.400	
.450	-.079	.028	-.115	.062	-.160	.067	-.164	.090	-.149	.029	.146	.016	.141	.003	.500	
.500	-.064	.005	-.107	.003	-.151	.010	-.150	.021	-.130	-.021	-.128	-.053	-.124	-.057	.600	
.550	-.008	-.061	-.086	-.098	-.075	-.117	-.061	-.149	-.101	-.075	-.124	-.075	-.132	-.132	.700	
.600	-.009	-.112	.020	-.152	-.006	-.154	-.050	-.133	-.017	-.164	.033	-.180	-.027	-.177	.800	
.650	-.046	-.128	.079	-.190	.072	-.195	.028	-.179	-.017	-.164	.033	-.180	-.027	-.177	.850	
.700	-.046	-.128	.079	-.190	.072	-.195	.028	-.179	-.017	-.164	.033	-.180	-.027	-.177	.900	
.750	-.050	-.128	.079	-.190	.072	-.195	.028	-.179	-.017	-.164	.033	-.180	-.027	-.177	.950	
$\alpha = .00$																
.0125	.251	-.019	.296	-.119	.343	-.142	.239	-.170	.307	-.157	.294	-.164	.215	-.126	.0125	
.025	.213	-.074	.232	-.115	.286	-.142	.255	-.138	.272	-.115	.219	-.123	.215	-.123	.025	
.050	.167	-.050	.183	-.100	.224	-.111	.225	-.118	.229	-.115	.220	-.115	.215	-.123	.050	
.075	.144	-.032	.162	-.089	.166	-.111	.166	-.123	.220	-.115	.215	-.115	.215	-.123	.075	
.100	.114	-.031	.074	-.131	.131	-.123	.136	-.154	.205	-.154	.177	-.155	.177	-.155	.100	
.150	.063	-.003	.061	-.050	.060	-.079	.080	-.090	.079	-.079	.105	-.084	.086	-.086	.150	
.200	.023	.033	.006	-.007	.010	-.046	.015	-.067	.026	-.025	.044	-.043	.005	-.039	.200	
.250	-.004	.072	-.043	-.085	-.050	-.036	-.025	-.016	.044	-.043	-.005	-.030	-.020	-.039	.250	
.300	-.006	.063	-.089	-.117	-.089	-.133	-.084	-.141	-.104	-.142	-.090	-.084	-.104	-.058	.300	
.350	-.080	.086	-.136	-.110	-.146	-.142	-.142	-.161	-.150	-.155	-.141	-.162	-.143	-.123	.400	
.400	-.091	.060	-.153	-.090	-.184	-.101	-.182	-.135	-.166	-.107	-.161	-.116	-.154	-.108	.450	
.450	-.083	.027	-.134	-.026	-.171	-.042	-.173	-.064	-.148	-.031	-.141	-.027	-.140	-.034	.500	
.500	-.055	-.034	.042	-.015	.007	.021	-.009	.037	-.025	.042	-.008	.080	-.014	.067	.600	
.550	-.060	-.021	.016	-.020	.014	-.054	.089	-.046	.027	-.028	.003	-.028	-.026	-.020	.650	
.600	-.083	-.027	.052	-.031	.075	-.029	.038	-.028	.028	-.028	.003	-.028	-.026	-.020	.700	
.650	-.078	-.003	.062	-.031	.062	-.023	.017	-.017	.006	-.017	.065	-.170	-.069	-.171	.800	
.700	-.016	-.034	.103	-.088	.118	-.087	.149	-.068	.161	-.008	.060	-.053	.158	-.043	.700	
.750	-.016	-.034	.103	-.088	.118	-.087	.149	-.068	.161	-.008	.060	-.053	.158	-.043	.750	
.800	-.016	-.096	-.009	-.127	-.046	-.130	-.085	-.108	-.046	-.129	-.009	.131	-.053	-.123	.800	
.850	-.006	-.026	-.109	-.049	-.169	-.033	-.174	-.020	-.157	-.017	-.083	-.107	-.025	-.099	.850	
.900	-.006	-.083	-.039	-.109	-.089	-.112	-.120	-.088	-.132	-.034	-.128	-.026	-.129	-.029	.900	
.950	-.006	-.083	-.039	-.109	-.089	-.112	-.120	-.088	-.132	-.034	-.128	-.026	-.129	-.029	.950	
$\alpha = .02$																
.0125	.195	-.019	.229	-.050	.273	-.075	.259	-.104	.217	-.095	.260	-.103	.247	-.103	.004	.0125
.025	.166	-.039	.195	-.065	.220	-.075	.204	-.125	.247	-.125	.204	-.124	.247	-.124	.025	
.050	.140	-.005	.168	-.045	.175	-.055	.192	-.079	.227	-.125	.227	-.124	.227	-.124	.050	
.075	.106	-.006	.106	-.035	.119	-.050	.192	-.079	.227	-.125	.227	-.124	.227	-.124	.075	
.100	.080	-.016	.080	-.024	.082	-.054	.089	-.046	.227	-.125	.227	-.124	.227	-.124	.100	
.150	.034	.042	-.015	-.007	.021	-.009	.037	-.025	.042	-.008	.080	-.014	.075	-.014	.150	
.200	-.003	.062	-.031	-.075	-.029	-.038	-.028	-.028	.037	-.028						

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING

(b) M 2.01 - Continued

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING
(b) $M = 2.01$ - Continued

TABLE III.- Concluded
PRESSURE COEFFICIENTS FOR REFLEX CAMBERED WING
(b) M = 2.01 - Concluded

x/c	C _p at $y/\frac{c}{2}$ of :														x/c	
	+.05		+.20		+.35		+.50		+.70		+.825		+.95			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 20$																
.0125	-.167	.517	-.220	.619	-.249	.649	-.294	.702	-.299		-.288	.645		.602	.0125	
.025	-.186	.474	-.228	.577	-.238	.626	-.289	.652	-.279	.662	-.285	.635	-.291	.617	.025	
.050	-.169	.421	-.223	.521	-.245	.626	-.279	.652	-.279	.650	-.285	.663	-.285	.635	.050	
.075	-.176	.413	-.224	.511	-.245	.608	-.272	.658	-.272	.639	-.287	.651	-.267	.608	.075	
.100	-.184	.414	-.248	.509	-.246	.605	-.272	.658	-.278	.639	-.287	.651	-.267	.608	.100	
.150	-.164	.412	-.263	.528	-.256	.621	-.278	.659	-.284	.645	-.288	.637	-.281	.572	.150	
.200	-.172	.470	-.263	.589	-.269	.631	-.284	.645	-.288	.630	-.288	.637	-.281	.572	.200	
.250	-.188	.511	-.274	.608	-.282	.622	-.298	.630	-.288	.630	-.288	.637	-.281	.577	.250	
.300	-.175	.583	-.283	.614	-.291	.616	-.302	.638	-.298	.658	-.280	.588	-.280	.541	.300	
.350															.350	
.400															.400	
.450															.450	
.500	-.238	.540	-.297	.596	-.309	.513	-.309	.516	-.299	.515	-.284	.503	-.290	.507	.500	
.550															.550	
.600	-.238	.459	-.275	.418	-.309	.402	-.312	.400	-.297	.432	-.284	.435	-.296	.428	.600	
.650															.650	
.700	-.219	.349		.307	-.301	.283	-.305	.275	-.294	.216	-.282	.213	-.289	.189	.700	
.750															.750	
.800	-.179	.200	-.268	.173	-.290	.149	-.300	.163	-.292	.128	-.151	.125	-.287	.121	.800	
.850															.850	
.900	-.142	.167	-.235	.103	-.260	.083	-.294	.102							.900	
.950															.950	

TABLE IV
PRESSURE COEFFICIENTS FOR FLAT WING

[From reference 1]

(a) M = 1.61

TABLE IV.- Continued

PRESSURE COEFFICIENTS FOR FLAT WING

[From reference 1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = 0.8$														
.0125	-+124	.386	-+225	+429	-+268	+460	-+237	+469	-+364	+398	-+378	+348	.0125	
.0250	-+162	.325	-+272	+368	-+318	+394	-+353	+395	-+364	+398	-+378	+348	.0250	
.0500	-+128	.293	-+281	+289	-+313	+323	-+355	+346	-+368	+349	-+378	+348	.0500	
.1000	-+072	.234	-+262	+246	-+301	+274	-+332	+279	-+345	+293	-+366	+284	.1000	
.1500	-+075	.224	-+225	+228	-+228	+255	-+251	+321	-+334	+244	-+363	+246	.1500	
.2000	-+087	.198	-+159	+221	-+278	+240	-+313	+234	-+330	+232	-+358	+214	.2000	
.2500	-+083	.202	-+142	+182	-+272	+201	-+309	+201	-+328	+206	-+355	+181	.2500	
.3000	-+090	.170	-+133	+178	-+264	+172	-+311	+178	-+332	+182	-+356	+169	.3000	
.3500	-+107	.165	-+132	+157	-+214	+154	-+313	+163	-+337	+148	-+358	+152	.3500	
.4000	-+111	.153	-+155	+141	-+200	+137	-+311	+144	-+337	+148	-+358	+141	.4000	
.4500	-+096	.173	-+165	+130	-+196	+124	-+315	+125	-+346	+108	-+358	+4500		
.5000	-+119	.128	-+165	+115	-+200	+104	-+321	+103	-+350	+099	-+368	+095	.5000	
.5500	-+145	.105	-+168	+086	-+212	+076	-+321	+083	-+352	+078	-+370	+095	.5500	
.6000	-+140	.104	-+182	+067	-+221	+058	-+301	+062	-+355	+059	-+369	+077	.6000	
.6500	-+151	.092	-+184	+069	-+223	+065	-+270	+058	-+357	+048	-+371	+058	.6500	
.7000	-+151	.076	-+176	+072	-+225	+047	-+255	+059	-+354	+060	-+374	+054	.7000	
.7500	-+141	.054	-+182	+062	-+210	+073	-+255	+054	-+358	+045	-+372	+044	.7500	
.8000	-+139	.066	-+181	+059	-+207	+065	-+243	+055	-+358	+048	-+375	+040	.8000	
.8500	-+143	.070	-+174	+067	-+204	+064	-+239	+062	-+358	+046	-+376	+033	.8500	
.9000	-+147	.071	-+168	+070	-+205	+064	-+237	+055	-+357	+044	-+376	+000	.9000	
.9500	-+155	.067											.9500	
$\alpha = 10$														
.0125	-+168	.450	-+283	+491	-+324	+515	-+307	+505	-+408	+453	-+416	+392	.0125	
.0250	-+207	.392	-+321	+437	-+374	+455	-+392	+444	-+408	+453	-+416	+392	.0250	
.0500	-+181	.318	-+325	+346	-+368	+390	-+396	+404	-+414	+402	-+416	+392	.0500	
.1000	-+092	.301	-+310	+313	-+350	+334	-+378	+333	-+387	+348	-+410	+342	.1000	
.1500	-+101	.275	-+284	+296	-+340	+314	-+359	+316	-+378	+301	-+404	+293	.1500	
.2000	-+111	.290	-+264	+276	-+333	+302	-+359	+292	-+374	+281	-+399	+261	.2000	
.2500	-+111	.254	-+213	+278	-+331	+256	-+357	+261	-+374	+264	-+396	+234	.2500	
.3000	-+107	.218	-+173	+232	-+333	+222	-+359	+239	-+378	+235	-+394	+220	.3000	
.3500	-+128	.214	-+161	+210	-+329	+211	-+357	+220	-+380	+199	-+395	+210	.3500	
.4000	-+132	.256	-+175	+196	-+292	+192	-+357	+195	-+378	+189	-+394	+199	.4000	
.4500	-+120	.223	-+183	+180	-+259	+170	-+361	+170	-+380	+150	-+380	+150	.4500	
.5000	-+139	.167	-+185	+160	-+247	+149	-+365	+145	-+384	+141	-+403	+145	.5000	
.5500	-+161	.154	-+186	+135	-+247	+119	-+368	+128	-+387	+115	-+404	+141	.5500	
.6000	-+159	.148	-+221	+115	-+253	+109	-+368	+119	-+390	+105	-+404	+123	.6000	
.6500	-+168	.135	-+199	+114	-+251	+111	-+365	+110	-+391	+107	-+406	+109	.6500	
.7000	-+168	.113	-+193	+115	-+252	+99	-+359	+109	-+388	+119	-+406	+113	.7000	
.7500	-+157	.093	-+199	+105	-+240	+113	-+330	+108	-+391	+095	-+407	+106	.7500	
.8000	-+156	.109	-+195	+101	-+233	+109	-+303	+103	-+392	+099	-+408	+101	.8000	
.8500	-+161	.110	-+192	+113	-+228	+107	-+290	+103	-+391	+103	-+407	+090	.8500	
.9000	-+168	.115	-+186	+112	-+224	+108	-+281	+097	-+388	+092	-+402	+090	.9000	
.9500	-+172	.110											.9500	
$\alpha = 12$														
.0125	-+215	.508	-+333	+538	-+375	+545	-+367	+526	-+448	+488	-+457	+443	.0125	
.0250	-+249	.442	-+371	+482	-+406	+498	-+433	+493	-+448	+442	-+457	+420	.0250	
.0500	-+240	.368	-+400	+404	-+445	+431	-+413	+384	-+430	+393	-+450	+404	.0500	
.1000	-+116	.351	-+357	+359	-+390	+384	-+375	+400	-+364	+419	-+445	+349	.1000	
.1500	-+175	.317	-+331	+343	-+378	+375	-+390	+375	-+387	+346	-+445	+320	.1500	
.2000	-+134	.296	-+308	+316	-+371	+352	-+378	+347	-+416	+332	-+440	+295	.2000	
.2500	-+135	.301	-+269	+275	-+367	+306	-+396	+311	-+414	+311	-+416	+300	.2500	
.3000	-+126	.267	-+220	+278	-+370	+273	-+395	+286	-+414	+278	-+434	+284	.3000	
.3500	-+151	.251	-+224	+257	-+371	+259	-+394	+264	-+416	+265	-+434	+261	.3500	
.4000	-+155	.244	-+219	+239	-+372	+239	-+394	+241	-+416	+233	-+434	+248	.4000	
.4500	-+145	.264	-+217	+219	-+341	+213	-+397	+210	-+417	+197	-+436	+160	.4500	
.5000	-+155	.254	-+208	+192	-+307	+188	-+400	+186	-+420	+193	-+436	+202	.5000	
.5500	-+185	.197	-+211	+180	-+300	+168	-+402	+174	-+420	+172	-+437	+194	.5500	
.6000	-+180	.185	-+222	+155	-+303	+154	-+405	+159	-+421	+158	-+437	+174	.6000	
.6500	-+170	.172	-+219	+151	-+292	+160	-+399	+153	-+424	+157	-+430	+164	.6500	
.7000	-+188	.148	-+211	+156	-+290	+147	-+402	+156	-+422	+165	-+428	+169	.7000	
.7500	-+175	.127	-+220	+142	-+275	+156	-+398	+152	-+424	+147	-+425	+160	.7500	
.8000	-+178	.146	-+215	+141	-+266	+154	-+361	+148	-+423	+143	-+425	+162	.8000	
.8500	-+181	.146	-+207	+158	-+259	+154	-+423	+144	-+420	+142	-+422	+156	.8500	
.9000	-+185	.152	-+201	+156	-+253	+152	-+426	+139	-+329	+140	-+400	+9000	.9000	
.9500	-+191	.146											.9500	
$\alpha = 14$														
.0125	-+269	.565	-+375	+575	-+420	+564	-+424	+533	-+474	+515	-+481	+497	.0125	
.0250	-+285	.499	-+404	+535	-+444	+539	-+465	+532	-+474	+480	-+481	+500	.0250	
.0500	-+292	.477	-+402	+444	-+447	+417	-+429	+428	-+469	+462	-+474	+446	.0500	
.1000	-+173	.401	-+389	+417	-+429	+428	-+449	+428	-+466	+426	-+466	+397	.1000	
.1500	-+140	.369	-+361	+408	-+416	+407	-+436	+429	-+467	+417	-+467	+350	.1500	
.2000	-+153	.348	-+361	+329	-+416	+410	-+431	+400	-+446	+387	-+463	+378	.2000	
.2500	-+156	.356	-+330	+329	-+407	+353	-+429	+358	-+443	+354	-+458	+366	.2500	
.3000	-+146	.314	-+268	+330	-+408	+326	-+426	+332	-+440	+325	-+459	+358	.3000	
.3500	-+146	.301	-+271	+308	-+410	+308	-+424	+304	-+443	+292	-+456	+339	.3500	
.4000	-+177	.294	-+286	+411	-+421	+281	-+424	+282	-+441	+285	-+456	+331	.4000	
.4500	-+167	.316	-+265	+237	-+423	+237	-+447	+235	-+467	+253	-+458	+283	.4500	
.5000	-+177	.246	-+251	+237	-+427	+237	-+447	+235	-+467	+239	-+458	+277	.5000	
.5500	-+200	.245	-+241	+225	-+416	+211	-+429	+220	-+446	+239	-+458	+277	.5500	
.6000	-+198	.234	-+244	+201	-+431	+206	-+443	+205	-+466	+239	-+454	+262	.6000	
.6500	-+208	.219	-+240	+197	-+410	+205	-+424	+203	-+446	+233	-+454	+261	.6500	
.7000	-+204	.194	-+227	+199	-+334	+182	-+432	+197	-+445	+244	-+453	+274	.7000	
.7500	-+188	.169	-+226	+181	-+313	+201	-+434	+206	-+449	+229	-+451	+284</		

TABLE IV.- Continued
PRESSURE COEFFICIENTS FOR FLAT WING
[From reference 1]

(a) $M = 1.61$ - Concluded

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = 16^\circ$														
.0125	-0.316	.624	-0.414	.616	-0.455	.588	-0.457	.542	-0.486	.549	-0.491	.579	.0125	
.0250	-0.325	.557	-0.441	.588	-0.471	.575	-0.486	.569	-0.490	.534	-0.491	.579	.0250	
.0500	-0.327	.491	-0.434	.527	-0.467	.549	-0.483	.551	-0.491	.534	-0.491	.579	.0500	
.1000	-0.246	.459	-0.422	.481	-0.455	.506	-0.472	.487	-0.481	.491	-0.487	.494	.1000	
.1500	-0.199	.422	-0.410	.466	-0.446	.488	-0.463	.467	-0.471	.467	-0.481	.514	.1500	
.2000	-0.181	.401	-0.400	.422	-0.439	.456	-0.459	.450	-0.469	.462	-0.470	.516	.2000	
.2500	-0.181	.418	-0.352	.388	-0.436	.413	-0.445	.406	-0.466	.440	-0.478	.529	.2500	
.3000	-0.163	.375	-0.307	.396	-0.435	.379	-0.451	.384	-0.463	.408	-0.479	.529	.3000	
.3500	-0.191	.358	-0.300	.366	-0.435	.358	-0.452	.360	-0.467	.390	-0.478	.513	.3500	
.4000	-0.195	.350	-0.304	.338	-0.435	.326	-0.447	.342	-0.467	.378	-0.477	.515	.4000	
.4500	-0.188	.372	-0.300	.312	-0.436	.309	-0.450	.326	-0.463	.354	-0.477	.515	.4500	
.5000	-0.198	.299	-0.293	.299	-0.400	.291	-0.451	.311	-0.467	.351	-0.475	.474	.5000	
.5500	-0.219	.297	-0.292	.280	-0.375	.259	-0.452	.313	-0.466	.338	-0.469	.460	.5500	
.6000	-0.218	.285	-0.292	.255	-0.368	.262	-0.455	.303	-0.467	.333	-0.461	.448	.6000	
.6500	-0.228	.268	-0.284	.252	-0.368	.279	-0.445	.302	-0.466	.335	-0.443	.439	.6500	
.7000	-0.222	.263	-0.271	.252	-0.367	.269	-0.453	.308	-0.390	.357	-0.430	.442	.7000	
.7500	-0.203	.217	-0.271	.244	-0.366	.292	-0.443	.313	-0.352	.366	-0.409	.424	.7500	
.8000	-0.218	.266	-0.263	.273	-0.368	.297	-0.348	.313	-0.350	.391	-0.404	.406	.8000	
.8500	-0.216	.251	-0.254	.290	-0.366	.308	-0.317	.309	-0.342	.402	-0.398	.368	.8500	
.9000	-0.224	.262	-0.247	.289	-0.362	.304	-0.303	.303	-0.347	.383	-0.383	.9000	.9000	
.9500	-0.230	.268	-0.247	.289	-0.362	.304	-0.303	.303	-0.347	.383	-0.383	.9500	.9500	
$\alpha = 18^\circ$														
.0125	-0.350	.677	-0.443	.649	-0.480	.600	-0.489	.534	-0.509	.598	-0.463	.648	.0125	
.0250	-0.359	.611	-0.443	.626	-0.490	.615	-0.503	.600	-0.509	.601	-0.463	.648	.0250	
.0500	-0.361	.552	-0.439	.584	-0.486	.595	-0.500	.593	-0.506	.601	-0.458	.641	.1000	
.1000	-0.332	.513	-0.447	.549	-0.476	.562	-0.494	.534	-0.500	.588	-0.453	.624	.1500	
.1500	-0.236	.480	-0.425	.504	-0.466	.540	-0.488	.531	-0.492	.558	-0.453	.624	.2000	
.2000	-0.179	.462	-0.408	.473	-0.464	.506	-0.482	.534	-0.496	.542	-0.449	.615	.2500	
.2500	-0.187	.478	-0.377	.440	-0.459	.464	-0.477	.500	-0.494	.533	-0.454	.610	.3000	
.3000	-0.173	.422	-0.361	.448	-0.456	.426	-0.476	.475	-0.485	.500	-0.450	.599	.3500	
.3500	-0.217	.415	-0.350	.417	-0.455	.407	-0.473	.452	-0.485	.495	-0.448	.582	.4000	
.4000	-0.211	.411	-0.358	.385	-0.453	.397	-0.470	.440	-0.485	.499	-0.451	.577	.4500	
.4500	-0.216	.426	-0.365	.350	-0.446	.393	-0.471	.428	-0.484	.495	-0.451	.565	.5000	
.5000	-0.217	.346	-0.371	.349	-0.437	.384	-0.473	.409	-0.487	.513	-0.444	.513	.5000	
.5500	-0.219	.354	-0.377	.328	-0.428	.373	-0.474	.401	-0.482	.513	-0.443	.499	.5500	
.6000	-0.230	.338	-0.374	.321	-0.419	.373	-0.474	.387	-0.474	.514	-0.436	.475	.6000	
.6500	-0.245	.322	-0.369	.336	-0.418	.375	-0.463	.400	-0.455	.515	-0.429	.462	.6500	
.7000	-0.234	.318	-0.352	.346	-0.416	.359	-0.473	.404	-0.421	.515	-0.427	.460	.7000	
.7500	-0.214	.316	-0.334	.354	-0.410	.382	-0.459	.412	-0.401	.484	-0.421	.442	.7500	
.8000	-0.220	.314	-0.348	.365	-0.410	.387	-0.391	.432	-0.385	.468	-0.420	.409	.8000	
.8500	-0.230	.338	-0.403	.385	-0.351	.454	-0.375	.446	-0.415	.371	-0.385	.9000	.9000	
.9000	-0.236	.364	-0.254	.372	-0.384	.385	-0.336	.454	-0.374	.401	-0.384	.9500	.9500	
$\alpha = 20^\circ$														
.0125	-0.389	.729	-0.471	.684	-0.498	.614	-0.502	.531	-0.478	.659	-0.256	.657	.0125	
.0250	-0.391	.670	-0.459	.674	-0.496	.660	-0.507	.660	-0.478	.688	-0.256	.657	.0250	
.0500	-0.392	.623	-0.456	.633	-0.491	.634	-0.501	.663	-0.479	.692	-0.273	.655	.1000	
.1000	-0.384	.671	-0.459	.601	-0.488	.624	-0.499	.652	-0.480	.630	-0.479	.698	.1500	
.1500	-0.338	.548	-0.450	.650	-0.494	.613	-0.498	.630	-0.479	.686	-0.283	.650	.2000	
.2000	-0.186	.576	-0.423	.542	-0.486	.586	-0.496	.614	-0.473	.707	-0.288	.650	.2500	
.2500	-0.212	.539	-0.413	.505	-0.482	.554	-0.496	.577	-0.467	.686	-0.296	.642	.3000	
.3000	-0.245	.482	-0.404	.523	-0.486	.533	-0.496	.557	-0.467	.662	-0.296	.635	.3500	
.3500	-0.226	.476	-0.405	.493	-0.486	.520	-0.471	.538	-0.458	.650	-0.297	.620	.4000	
.4000	-0.227	.474	-0.39	.468	-0.440	.513	-0.401	.527	-0.458	.630	-0.303	.610	.4500	
.4500	-0.227	.494	-0.415	.469	-0.436	.497	-0.487	.510	-0.455	.606	-0.313	.542	.5000	
.5000	-0.249	.434	-0.424	.662	-0.426	.640	-0.426	.525	-0.446	.570	-0.321	.529	.5500	
.5500	-0.255	.438	-0.431	.644	-0.426	.646	-0.425	.553	-0.440	.554	-0.329	.506	.6000	
.6000	-0.260	.442	-0.425	.634	-0.426	.643	-0.425	.579	-0.425	.544	-0.333	.491	.6500	
.6500	-0.245	.445	-0.418	.645	-0.436	.645	-0.449	.591	-0.409	.533	-0.340	.491	.7000	
.7000	-0.216	.424	-0.383	.646	-0.433	.599	-0.442	.573	-0.403	.500	-0.339	.461	.7500	
.7500	-0.239	.448	-0.342	.660	-0.435	.527	-0.415	.553	-0.402	.486	-0.341	.427	.8000	
.8000	-0.249	.456	-0.34	.648	-0.435	.547	-0.396	.524	-0.400	.462	-0.342	.389	.8500	
.8500	-0.257	.458	-0.275	.669	-0.396	.523	-0.370	.484	-0.400	.411	-0.341	.9000	.9000	
.9000	-0.261	.443	-0.306	.662	-0.396	.523	-0.370	.484	-0.400	.411	-0.341	.9500	.9500	

TABLE IV.- Continued

PRESSURE COEFFICIENTS FOR FLAT WING

[From reference 1]

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
a = 00														
.0125	+.174		+.168		+.184		+.227		+.076		+.089		.0125	
.0250	+.095		+.080		+.107		+.096		+.084		+.077		.0250	
.0375	+.056		+.068		+.084		+.085		+.084		+.077		.0375	
.0500	+.047		+.035		+.062		+.088		+.103		+.062		.0500	
.0625	+.035		+.020		+.035		+.045		+.076		+.062		.0625	
.0750	+.021		+.014		+.020		+.028		+.022		+.055		.0750	
.0875	+.011		+.003		+.010		+.013		+.007		+.039		.0875	
.1000	+.006		+.004		+.009		+.003		+.010		+.008		.1000	
.1125	+.006		+.008		+.018		+.010		+.025		+.024		.1125	
.1250	+.011		+.021		+.029		+.022		+.037		+.032		.1250	
.1375	+.007		+.013		+.018		+.017		+.047		+.047		.1375	
.1500	+.007		+.013		+.018		+.017		+.051		+.058		.1500	
.1625	+.029		+.038		+.048		+.051		+.058		+.060		.1625	
.1750	+.030		+.051		+.057		+.061		+.065		+.067		.1750	
.1875	+.035		+.064		+.071		+.072		+.074		+.073		.1875	
.2000	+.046		+.079		+.087		+.093		+.091		+.094		.2000	
.2125	+.046		+.084		+.096		+.097		+.092		+.092		.2125	
.2250	+.046		+.084		+.096		+.097		+.098		+.098		.2250	
.2375	+.046		+.084		+.096		+.097		+.098		+.098		.2375	
.2500	+.046		+.084		+.096		+.097		+.098		+.098		.2500	
.2625	+.046		+.084		+.096		+.097		+.098		+.098		.2625	
.2750	+.046		+.084		+.096		+.097		+.098		+.098		.2750	
.2875	+.046		+.084		+.096		+.097		+.098		+.098		.2875	
.3000	+.046		+.084		+.096		+.097		+.098		+.098		.3000	
.3125	+.046		+.084		+.096		+.097		+.098		+.098		.3125	
.3250	+.046		+.084		+.096		+.097		+.098		+.098		.3250	
.3375	+.046		+.084		+.096		+.097		+.098		+.098		.3375	
.3500	+.046		+.084		+.096		+.097		+.098		+.098		.3500	
.3625	+.046		+.084		+.096		+.097		+.098		+.098		.3625	
.3750	+.046		+.084		+.096		+.097		+.098		+.098		.3750	
.3875	+.046		+.084		+.096		+.097		+.098		+.098		.3875	
.4000	+.046		+.084		+.096		+.097		+.098		+.098		.4000	
.4125	+.046		+.084		+.096		+.097		+.098		+.098		.4125	
.4250	+.046		+.084		+.096		+.097		+.098		+.098		.4250	
.4375	+.046		+.084		+.096		+.097		+.098		+.098		.4375	
.4500	+.046		+.084		+.096		+.097		+.098		+.098		.4500	
.4625	+.046		+.084		+.096		+.097		+.098		+.098		.4625	
.4750	+.046		+.084		+.096		+.097		+.098		+.098		.4750	
.4875	+.046		+.084		+.096		+.097		+.098		+.098		.4875	
.5000	+.046		+.084		+.096		+.097		+.098		+.098		.5000	
.5125	+.046		+.084		+.096		+.097		+.098		+.098		.5125	
.5250	+.046		+.084		+.096		+.097		+.098		+.098		.5250	
.5375	+.046		+.084		+.096		+.097		+.098		+.098		.5375	
.5500	+.046		+.084		+.096		+.097		+.098		+.098		.5500	
.5625	+.046		+.084		+.096		+.097		+.098		+.098		.5625	
.5750	+.046		+.084		+.096		+.097		+.098		+.098		.5750	
.5875	+.046		+.084		+.096		+.097		+.098		+.098		.5875	
.6000	+.046		+.084		+.096		+.097		+.098		+.098		.6000	
.6125	+.046		+.084		+.096		+.097		+.098		+.098		.6125	
.6250	+.046		+.084		+.096		+.097		+.098		+.098		.6250	
.6375	+.046		+.084		+.096		+.097		+.098		+.098		.6375	
.6500	+.046		+.084		+.096		+.097		+.098		+.098		.6500	
.6625	+.046		+.084		+.096		+.097		+.098		+.098		.6625	
.6750	+.046		+.084		+.096		+.097		+.098		+.098		.6750	
.6875	+.046		+.084		+.096		+.097		+.098		+.098		.6875	
.7000	+.046		+.084		+.096		+.097		+.098		+.098		.7000	
.7125	+.046		+.084		+.096		+.097		+.098		+.098		.7125	
.7250	+.046		+.084		+.096		+.097		+.098		+.098		.7250	
.7375	+.046		+.084		+.096		+.097		+.098		+.098		.7375	
.7500	+.046		+.084		+.096		+.097		+.098		+.098		.7500	
.7625	+.046		+.084		+.096		+.097		+.098		+.098		.7625	
.7750	+.046		+.084		+.096		+.097		+.098		+.098		.7750	
.7875	+.046		+.084		+.096		+.097		+.098		+.098		.7875	
.8000	+.046		+.084		+.096		+.097		+.098		+.098		.8000	
.8125	+.046		+.084		+.096		+.097		+.098		+.098		.8125	
.8250	+.046		+.084		+.096		+.097		+.098		+.098		.8250	
.8375	+.046		+.084		+.096		+.097		+.098		+.098		.8375	
.8500	+.046		+.084		+.096		+.097		+.098		+.098		.8500	
.8625	+.046		+.084		+.096		+.097		+.098		+.098		.8625	
.8750	+.046		+.084		+.096		+.097		+.098		+.098		.8750	
.8875	+.046		+.084		+.096		+.097		+.098		+.098		.8875	
.9000	+.046		+.084		+.096		+.097		+.098		+.098		.9000	
.9125	+.046		+.084		+.096		+.097		+.098		+.098		.9125	
.9250	+.046		+.084		+.096		+.097		+.098		+.098		.9250	
.9375	+.046		+.084		+.096		+.097		+.098		+.098		.9375	
.9500	+.046		+.084		+.096		+.097		+.098		+.098		.9500	

TABLE IV.- Continued
PRESSURE COEFFICIENTS FOR FLAT WING
[From reference 1]
(b) M = 2.01 - Continued

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
a=0														
.0125	-.018	.384	-.059	.422	-.080	.453	-.045	.477	-.174	.392	-.185	.254	.0125	
.0250	-.028	.297	-.119	.346	-.135	.374	-.159	.384	-.174	.392	-.185	.254	.0250	
.0500	-.101	.264	-.140	.298	-.144	.322	-.165	.332	-.184	.350	-.185	.354	.0500	
.1000	-.077	.196	-.145	.227	-.152	.248	-.159	.279	-.170	.301	-.182	.300	.1000	
.1500	-.063	.180	-.154	.200	-.149	.237	-.154	.249	-.166	.242	-.185	.259	.1500	
.2000	-.076	.161	-.155	.185	-.153	.213	-.159	.233	-.170	.230	-.183	.233	.2000	
.2500	-.082	.151	-.146	.161	-.161	.181	-.164	.202	-.174	.204	-.184	.201	.2500	
.3000	-.173	.150	-.124	.162	-.169	.163	-.169	.173	-.181	.180	-.187	.177	.3000	
.3500	-.106	.144	-.124	.141	-.178	.142	-.175	.159	-.189	.158	-.193	.169	.3500	
.4000	-.132	.131	-.129	.121	-.184	.128	-.179	.142	-.193	.146	-.193	.158	.4000	
.4500	-.100	.122	-.133	.106	-.187	.111	-.185	.118	-.200	.119	-.208	.119	.4500	
.5000	-.116	.094	-.137	.095	-.193	.096	-.194	.099	-.205	.105	-.208	.119	.5000	
.5500	-.119	.094	-.148	.079	-.192	.077	-.199	.085	-.208	.091	-.212	.107	.5500	
.6000	-.122	.085	-.155	.059	-.181	.059	-.204	.069	-.213	.076	-.216	.095	.6000	
.6500	-.124	.078	-.155	.061	-.178	.053	-.209	.058	-.216	.066	-.217	.086	.6500	
.7000	-.123	.073	-.153	.066	-.182	.039	-.209	.058	-.215	.073	-.217	.079	.7000	
.7500	-.127	.067	-.152	.054	-.174	.054	-.214	.047	-.218	.054	-.220	.066	.7500	
.8000	-.112	.042	-.149	.055	-.173	.053	-.220	.042	-.219	.049	-.223	.058	.8000	
.8500	-.116	.053	-.144	.052	-.171	.056	-.219	.039	-.229	.046	-.225	.052	.8500	
.9000	-.120	.056	-.142	.046	-.177	.050	-.221	.038	-.222	.039	-.225	.040	.9000	
.9500	-.124	.054	-.142	.046	-.177	.050	-.221	.038	-.222	.039	-.225	.040	.9500	
a=10														
.0125	-.742	.438	-.115	.475	-.131	.502	-.105	.514	-.218	.446	-.227	.407	.0125	
.0250	-.125	.354	-.162	.442	-.182	.433	-.201	.440	-.218	.446	-.227	.309	.0250	
.0500	-.143	.337	-.179	.344	-.189	.378	-.211	.396	-.225	.405	-.227	.200	.0500	
.1000	-.115	.242	-.182	.291	-.192	.308	-.201	.334	-.212	.346	-.226	.350	.1000	
.1500	-.085	.218	-.186	.250	-.187	.296	-.195	.301	-.207	.311	-.227	.309	.1500	
.2000	-.098	.207	-.193	.233	-.191	.268	-.196	.282	-.208	.284	-.221	.278	.2000	
.2500	-.142	.195	-.191	.207	-.195	.241	-.201	.255	-.212	.262	-.222	.247	.2500	
.3000	-.149	.194	-.174	.207	-.201	.211	-.208	.226	-.217	.232	-.224	.231	.3000	
.3500	-.127	.184	-.158	.186	-.177	.193	-.211	.208	-.222	.206	-.226	.218	.3500	
.4000	-.119	.182	-.156	.183	-.212	.172	-.212	.189	-.226	.193	-.227	.207	.4000	
.4500	-.122	.163	-.155	.145	-.219	.156	-.219	.165	-.231	.161	-.237	.162	.4500	
.5000	-.135	.135	-.154	.136	-.227	.138	-.223	.144	-.235	.150	-.237	.162	.5000	
.5500	-.136	.135	-.167	.138	-.224	.137	-.228	.127	-.237	.134	-.241	.152	.5500	
.6000	-.139	.122	-.178	.106	-.231	.111	-.235	.112	-.240	.120	-.242	.141	.6000	
.6500	-.142	.114	-.175	.100	-.227	.105	-.236	.102	-.244	.109	-.245	.127	.6500	
.7000	-.143	.111	-.168	.102	-.219	.076	-.238	.099	-.242	.115	-.246	.123	.7000	
.7500	-.143	.078	-.171	.092	-.204	.084	-.241	.089	-.248	.095	-.248	.112	.7500	
.8000	-.127	.078	-.169	.089	-.195	.094	-.243	.083	-.244	.090	-.246	.104	.8000	
.8500	-.132	.087	-.163	.099	-.195	.083	-.244	.082	-.245	.088	-.236	.097	.8500	
.9000	-.138	.096	-.159	.084	-.193	.089	-.246	.082	-.246	.082	-.246	.090	.9000	
.9500	-.141	.086	-.159	.084	-.193	.089	-.246	.082	-.246	.082	-.246	.090	.9500	
a=12														
.0125	-.083	.492	-.156	.520	-.169	.541	-.156	.541	-.248	.492	-.254	.553	.0125	
.0250	-.153	.417	-.196	.450	-.193	.480	-.229	.480	-.253	.485	-.256	.576	.0250	
.0500	-.175	.351	-.207	.388	-.216	.433	-.234	.447	-.253	.456	-.261	.576	.0500	
.1000	-.142	.294	-.209	.332	-.216	.358	-.227	.381	-.241	.396	-.251	.576	.1000	
.1500	-.114	.264	-.214	.305	-.213	.343	-.242	.321	-.234	.345	-.249	.576	.1500	
.2000	-.113	.252	-.216	.294	-.215	.313	-.228	.328	-.237	.335	-.245	.576	.2000	
.2500	-.120	.236	-.216	.254	-.219	.287	-.226	.296	-.237	.313	-.245	.576	.2500	
.3000	-.111	.239	-.217	.256	-.224	.257	-.230	.272	-.249	.283	-.246	.576	.3000	
.3500	-.142	.229	-.196	.231	-.229	.237	-.230	.255	-.244	.254	-.249	.576	.3500	
.4000	-.135	.215	-.186	.227	-.227	.216	-.237	.216	-.244	.232	-.249	.576	.4000	
.4500	-.135	.198	-.181	.191	-.237	.196	-.237	.193	-.249	.197	-.250	.576	.4500	
.5000	-.150	.176	-.175	.178	-.241	.179	-.241	.182	-.254	.199	-.256	.576	.5000	
.5500	-.154	.177	-.181	.186	-.244	.185	-.245	.186	-.255	.186	-.257	.576	.5500	
.6000	-.154	.16	-.197	.159	-.245	.149	-.248	.152	-.258	.154	-.260	.576	.6000	
.6500	-.157	.158	-.189	.138	-.245	.137	-.251	.140	-.260	.156	-.261	.576	.6500	
.7000	-.155	.150	-.184	.144	-.246	.146	-.252	.152	-.261	.158	-.268	.576	.7000	
.7500	-.154	.142	-.187	.125	-.244	.144	-.254	.154	-.263	.156	-.268	.576	.7500	
.8000	-.141	.116	-.163	.128	-.239	.111	-.265	.126	-.268	.136	-.275	.576	.8000	
.8500	-.146	.107	-.177	.124	-.246	.111	-.266	.126	-.268	.134	-.276	.576	.8500	
.9000	-.150	.129	-.171	.117	-.244	.115	-.261	.125	-.267	.126	-.276	.576	.9000	
.9500	-.153	.122	-.171	.117	-.244	.115	-.261	-.267	-.276	-.276	-.276	-.276	.9500	
a=14														
.0125	-.122	.546	-.193	.562	-.213	.576	-.176	.567	-.276	.535	-.282	.489	.0125	
.0250	-.193	.478	-.224	.511	-.246	.535	-.262	.539	-.281	.542	-.282	.506	.0250	
.0500	-.204	.394	-.239	.430	-.249	.475	-.263	.494	-.281	.492	-.282	.489	.0500	
.1000	-.184	.346	-.235	.384	-.245	.410	-.255	.428	-.269	.443	-.278	.441	.1000	
.1500	-.171	.310	-.239	.356	-.241	.394	-.249	.404	-.260	.394	-.276	.421	.1500	
.2000	-.131	.296	-.242	.332	-.241	.368	-.248	.378	-.262	.381	-.272	.364	.2000	
.2500	-.134	.278	-.243	.304	-.244	.331	-.252	.349	-.263	.354	-.271	.341	.2500	
.3000	-.135	.283	-.243	.302	-.248	.352	-.255	.354	-.263	.352	-.271	.340	.3000	
.3500	-.151	.272	-.225	.275	-.253	.283	-.257	.297	-.267	.297	-.271	.345	.3500	
.4000	-.154	.256	-.222	.246	-.256	.260	-.258	.277	-.271	.284	-.271	.340	.4000	
.4500	-.149	.244	-.213	.230	-.261	.270	-.263	.284	-.275	.275	-.271	.340	.4500	
.5000	-.164	.217	-.216	.219	-.268	.271	-.264	.286	-.277	.284	-.271	.340	.5000	
.5500	-.164	.217	-.216	.217	-.268	.271	-.264	.286	-.277	.284	-.271	.340	.5500	
.6000	-.168	.200	-.213	.176	-.271	.150	-.271	.192	-.277	.166	-.271	.249	.6000	
.6500	-.170	.194	-.211	.177	-.267	.157	-.272	.181	-.279	.164	-.271	.245	.6500	
.7000	-.169	.190	-.216	.178	-.267	.156	-.272	.181	-.278	.162	-.271	.245	.7000	
.7500	-.165	.147	-.207	.167	-.263	.176	-.270							

TABLE IV.- Concluded
PRESSURE COEFFICIENTS FOR FLAT WING
[From reference 1]

TABLE V
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(a) M = 1.61

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = -20$														
.0125	.895	-+465			-+457	.646	-+474	.532	-+475	.599	-+464		.0125	
.0250	.838	-+481			-+458	.651	-+479	.636	-+473	.622	-+457	.639	-+313	
.0500	.834	-+498	.702	-+458			-+479		-+473		-+457		.0500	
.1000	.722	-+469	.647	-+475	.620	-+481	.609	-+472	.631	-+464	.639	-+309	.1000	
.1500	.623	-+423	.580	-+472	.572	-+482	.591	-+472	.610	-+461	.661	-+301	.1500	
.2000	.535	-+317	.529	-+484	.539	-+482	.541	-+470	.553	-+462	.590	-+297	.2000	
.2500	.518		.452	-+492	.466	-+483	.472	-+474	.491	-+461	.563	-+294	.2500	
.3000	.509	-+310	.385	-+497	.432	-+483	.427	-+475	.470	-+461	.551	-+290	.3000	
.3500	.454		.382	-+501	.420	-+487	.412	-+473	.441	-+456	.535	-+288	.3500	
.4000	.387		.337	-+503	.407	-+491	.390	-+478	.438	-+438	.491	-+283	.4000	
.4500	.369		.344	-+509	.353	-+497	.363			-+433	.419	-+289	.4500	
.5000	.323	-+213	.348	-+508	.345	-+493	.331	-+485	.424	-+410	.445	-+318	.5000	
.6000	.284	-+210	.320	-+492	.313	-+490	.303	-+487	.422	-+422	.437	-+355	.6000	
.7000	.303	-+214	.294	-+434	.294	-+481	.360	-+483	.451	-+425	.403	-+322	.7000	
.8000	.333	-+202	.329	-+362		-+456	.438	-+445	.433	-+402	.398	-+293	.8000	
.9000	.358	-+220	.367	-+275	.405	-+425	.450	-+387	.410	-+379			.9000	
$\alpha = -18$														
.0125	.879	-+485			.584	-+471	.503	-+461					.0125	
.0250	.794	-+493			-+447	.633	-+472	.597	-+459	.585	-+441		.0250	
.0500	.779	-+506	.664	-+448	.624	-+473	.582	-+458	.575	-+437	.599	-+287	.0500	
.1000	.663	-+432	.592	-+465	.574	-+472	.527	-+457	.566	-+443	.588	-+283	.1000	
.1500	.557	-+393	.525	-+470	.515	-+471	.495	-+457	.525	-+440	.578	-+276	.1500	
.2000	.474	-+281	.478	-+481	.456	-+472	.442	-+454	.457	-+441	.501	-+273	.2000	
.2500	.451		.403	-+480	.384	-+473	.388	-+456	.379	-+441	.459	-+270	.2500	
.3000	.457	-+299	.328	-+479	.341	-+474	.349	-+456	.352	-+439	.440	-+267	.3000	
.3500	.397		.317	-+487	.310	-+479	.340	-+456	.311	-+438	.433	-+264	.3500	
.4000	.332		.284	-+499	.295	-+481	.306	-+461	.305	-+429	.397	-+260	.4000	
.4500	.318		.259	-+507	.251	-+489	.274			-+411	-+407	-+377	-+259	.4500
.5000	.272	-+196	.245	-+500	.254	-+487	.242	-+468	.300	-+395	.371	-+262	.5000	
.6000	.206	-+196	.197	-+466	.223	-+489	.218	-+473	.256	-+403	.393	-+243	.6000	
.7000	.198	-+204	.214	-+366	.224	-+478	.214	-+475	.330	-+415	.378	-+227	.7000	
.8000	.226	-+179	.244	-+259		-+451	.254	-+441	.372	-+387	.382	-+264	.8000	
.9000	.278	-+203	.283	-+194	.276	-+431	.325	-+383	.386	-+357			.9000	
$\alpha = -16$														
.0125	.835	-+489			.577	-+460	.494	-+444					.0125	
.0250	.754	-+494			-+443	.613	-+461	.577	-+443	.560	-+425		.0250	
.0500	.723	-+509	.624	-+441	.590	-+461	.549	-+443	.522	-+422	.518	-+283	.0500	
.1000	.609	-+421	.541	-+445	.529	-+461	.484	-+443	.501	-+425	.495	-+278	.1000	
.1500	.484	-+335	.467	-+463	.466	-+460	.448	-+443	.452	-+424	.471	-+272	.1500	
.2000	.413	-+246	.427	-+474	.402	-+459	.379	-+440	.377	-+425	.384	-+267	.2000	
.2500	.393		.357	-+463	.327	-+466	.309	-+441	.299	-+425	.346	-+263	.2500	
.3000	.401	-+254	.247	-+461	.289	-+470	.260	-+446	.276	-+423	.327	-+258	.3000	
.3500	.353		.258	-+470	.262	-+469	.241	-+448	.236	-+422	.314	-+256	.3500	
.4000	.283		.247	-+487	.236	-+466	.210	-+456	.223	-+420	.266	-+253	.4000	
.4500	.268		.210	-+491	.178	-+464	.189			-+407	.231	-+252	.4500	
.5000	.222	-+179	.198	-+476	.173	-+468	.163	-+463	.172	-+391	.223	-+250	.5000	
.6000	.141	-+170	.134	-+394	.107	-+477	.113	-+464	.113	-+390	.244	-+240	.6000	
.7000	.145	-+188	.124	-+224	.127	-+474	.129	-+463	.156	-+383	.259	-+237	.7000	
.8000	.143	-+183	.123	-+163		-+453	.153	-+442	.179	-+377	.282	-+244	.8000	
.9000	.174	-+190	.201	-+159	.187	-+433	.188	-+390	.240	-+345			.9000	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = -12$														
.0125	.721	-.476			.551	-.402	.485	-.394	.551	-.382			.0125	
.0250	.658	-.481			.567	-.404	.538	-.392	.473	-.378	.452	-.304	.0250	
.0500	.603	-.498	.545	-.428	.527	-.407	.497	-.392	.426	-.381	.412	-.296	.0500	
.1000	.491	-.436	.439	-.432	.448	-.417	.397	-.393	.354	-.378	.366	-.286	.1000	
.1500	.370	-.204	.344	-.464	.369	-.416	.342	-.396	.354	-.378	.380	-.281	.1500	
.2000	.303	-.145	.302	-.461	.297	-.419	.273	-.394	.274	-.380	.244	-.281	.2000	
.2500	.287		.268	-.440	.228	-.448	.214	-.398	.187	-.380	.191	-.275	.2500	
.3000	.293	-.168	.186	-.418	.176	-.452		-.402	.145	-.381	.158	-.269	.3000	
.3500	.246		.157	-.417	.162	-.452	.153	-.407	.115	-.377	.150	-.266	.3500	
.4000	.192		.148	-.436	.151	-.452	.114	-.417	.103	-.377	.107	-.259	.4000	
.4500	.173		.113	-.394	.063	-.445	.079		.069	-.379	.043	-.258	.4500	
.5000	.130	-.129	.107	-.254	.086	-.433	.067	-.421	.061	-.374	.030	-.253	.5000	
.6000	.080	-.136	.047	-.089	.020	-.408	.003	-.415	-.016	-.364	-.008	-.244	.6000	
.7000	.061	-.143	.040	-.099	.012	-.391	-.015	-.412	-.020	-.349	-.028	-.239	.7000	
.8000	.057	-.127	.039	-.118		-.371	.003	-.409	-.017	-.343	-.026	-.235	.8000	
.9000	.074	-.149	.071	-.122	.035	-.356	.024	-.306	.003	-.334			.9000	
$\alpha = -10$														
.0125		-.465			.406	.536	-.374	.474	-.369	.506	-.350		.0125	
.0250		-.466			.412	.538	-.377	.509	-.368	.440	-.347	.422	.0250	
.0500		-.478	.507	-.412	.480	.379	.462	-.367	.382	.351	.385	-.293	.0500	
.1000		-.400	.385	-.419	.395	-.393	.350	-.368	.382	.331	.385	-.283	.1000	
.1500		-.120	.282	-.439	.314	-.389	.288	-.367	.304	.348	.337	-.283	.1500	
.2000	.252	-.113	.239	-.447	.244	-.400	.215	-.366	.230	.350	.199	-.277	.2000	
.2500	.235		.218	-.427	.182	-.432	.154	-.370	.141	.347	.143	-.270	.2500	
.3000	.238	-.122	.142	-.401	.123	-.430	.102	-.383	.099	.345	.108	-.264	.3000	
.3500	.197		.106	-.396	.098	-.425	.102	-.389	.068	.344	.096	-.260	.3500	
.4000	.147		.100	-.386	.110	-.414	.081	-.390	.066	.347	.061	-.255	.4000	
.4500	.129		.058	-.192	.005	-.408	.006		-.008	.349	-.033	-.254	.4500	
.5000	.087	-.099	.063	-.045	.040	-.390	.000	-.388	-.009	.344	-.038	-.252	.5000	
.6000	.038	-.107	.000	-.063	-.026	-.360	-.043	-.372	-.068	.338	-.068	-.242	.6000	
.7000	.024	-.113	-.004	-.080	-.036	-.342	-.060	-.365	-.068	.323	-.076	-.237	.7000	
.8000	.019	-.099	-.004	-.096		-.313	-.054	-.357	-.062	.303	-.066	-.236	.8000	
.9000	.035	-.121	.024	-.098	-.011	-.224	-.025	-.270	-.057	-.287			.9000	
$\alpha = -08$														
.0125	.657	-.449			.518	-.346	.460	-.339					.0125	
.0250	.580	-.447			.384	.509	-.348	.482	-.336	.474	-.315	.381	.0250	
.0500	.498	-.459	.476	-.387	.445	-.350	.430	-.337	.407	-.315	.314	.344	.0500	
.1000	.393	-.330	.336	-.405	.339	-.361	.308	-.336	.326	-.313	.297	.287	.1000	
.1500	.267	-.081	.217	-.422	.260	-.359	.242	-.333	.252	-.313	.314	.145	.1500	
.2000	.201	-.098	.184	-.413	.190	-.379	.168	-.335	.173	-.310	.050	-.271	.2000	
.2500	.187		.166	-.406	.139	-.403	.105	-.344	.085	-.313	.084	-.276	.2500	
.3000	.187	-.097	.098	-.371	.077	-.394	.043	-.358	.040	-.310	.010	-.266	.3000	
.3500	.152		.060	-.355	.043	-.382	.032	-.360	.028	-.313	.029	-.262	.4000	
.4000	.103		.056	-.211	.063	-.364	.043	-.358	.028	-.313	.064	-.263	.4500	
.4500	.085		.010	-.010	-.037	-.350	-.028		-.037	-.310	-.077	-.264	.5000	
.5000	.045	-.072	.022	-.000	-.016	-.336	-.050	-.344	-.045	-.309	-.077	-.264	.6000	
.6000	-.000	-.079	-.038	-.049	-.064	-.311	-.084	-.325	-.114	-.310	-.116	-.263	.7000	
.7000	-.015	-.088	-.043	-.056	-.078	-.284	-.099	-.318	-.125	-.298	-.128	-.259	.7000	
.8000	-.019	-.076	-.039	-.074		-.169	-.096	-.308	-.117	-.274	-.126	-.261	.8000	
.9000	-.004	-.093	-.016	-.079	-.052	-.057	-.074	-.282	-.094	-.256			.9000	
$\alpha = -06$														
.0125	.611	-.430			.495	-.316	.441	-.302	.438	-.273	.331	-.266	.0125	
.0250	.544	-.428			.356	.469	-.315	.445	-.300	.369	-.272	.293	.0250	
.0500	.447	-.443	.427	-.359	.403	-.318	.380	-.300	.369	-.24	.270	.259	.0500	
.1000	.347	-.211	.285	-.384	.285	-.322	.251	-.298	.292	-.272	.293	-.265	.1000	
.1500	.218	-.053	.165	-.403	.200	-.326	.194	-.294	.204	-.270	.259	-.265	.1500	
.2000	.153	-.071	.134	-.389	.135	-.344	.127	-.296	.131	-.269	.101	-.264	.2000	
.2500	.142		.113	-.366	.097	-.353	.064	-.307	.042	-.267	.023	-.260	.2500	
.3000	.143	-.080	.060	-.331	.040	-.347	-.002	-.319	-.008	-.265	-.013	-.256	.3000	
.3500	.108		.018	-.219	-.002	-.334	-.025	-.313	-.047	-.265	-.030	-.253	.3500	
.4000	.063		.019	-.032	.014	-.316	-.017	-.302	-.047	-.265	-.040	-.248	.4000	
.4500	.044		.035	-.046	-.071	-.298	-.067		-.093	-.265	-.114	-.247	.4500	
.5000	.008	-.045	.016	-.017	-.063	-.286	-.089	-.283	-.092	-.266	-.127	-.248	.5000	
.6000	-.038	-.049	-.076	-.026	-.099	-.252	-.142	-.269	-.141	-.269	-.162	-.248	.6000	
.7000	-.052	-.061	-.077	-.027	-.115	-.164	-.136	-.258	-.154	-.258	-.170	-.241	.7000	
.8000	-.052	-.059	-.071	-.048		-.001	-.135	-.239	-.162	-.246	-.165	-.231	.8000	
.9000	-.038	-.067	-.051	-.054	-.091	-.012	-.127	-.226	-.157	-.215			.9000	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
	$\alpha = -04$													
.0125	.507	-.409			.330	.474	-.279	.424	-.260	.404	-.228		.0125	
.0250	.399	-.426	.389	-.333		.357	-.283	.334	-.260	.321	-.228	.291	.0250	
.0500	.299	-.075	.236	-.351		.232	-.283	.199	-.256	.244	-.227	.253	.0500	
.1000	.173	-.022	.112	-.361		.145	-.293	.128	-.250	.150	-.226	.215	.1000	
.2000	.111	-.039	.084	-.330		.079	-.304	.074	-.256	.086	-.225	.067	.2000	
.2500	.100		.065	-.300		.047	-.309	.022	-.266	.005	-.221	.023	.2500	
.3000	.101	-.056	.019	-.250		.001	-.304	.039	-.272	.046	-.220	.059	.3000	
.3500	.068		.022	.029		.042	-.284	.066	-.259	.086	-.220	.081	.3500	
.4000	.023		.021	.094		.038	-.260	.064	-.241	.092	-.220	.090	.4000	
.4500	.008		.075	.067		.111	-.242	.116		.138	-.220	.154	.4500	
.5000	-.027	-.018	.055	.031		.103	-.235	.134	-.230	.143	-.218	.170	.5000	
.6000	-.070	-.020	.109	-.002		.136	-.154	.176	-.205	.191	-.220	.204	.6000	
.7000	-.083	-.035	.111	-.004		.180	-.003	.188	-.193	.199	-.208	.213	.7000	
.8000	-.073	-.038	.105	-.023		.034	-.170	.171	-.199	.205	-.215	.200	.8000	
.9000	-.063	-.045	.081	-.034		.124	-.002	.162	-.154	.193	-.175		.9000	
$\alpha = -02$														
.0125	.537	-.392			.300	.452	-.238	.400	-.212	.365	-.179		.0125	
.0250	.459	-.387			.353	.405	-.240	.380	-.210				.0250	
.0500	.347	-.390	.353	-.303		.321	-.241	.295	-.211	.283	-.176	.252	.0500	
.1000	.255	-.006	.199	-.322		.184	-.240	.152	-.208	.200	-.176	.206	.1000	
.1500	.132	.001	.065	-.301		.087	-.251	.075	-.200	.094	-.176	.170	.1500	
.2000	.074	-.010	.030	-.254		.028	-.258	.020	-.205	.026	-.174	.028	.2000	
.2500	.060		.015	-.238		.015	-.264	.042	-.219	.043	-.172	.057	.2500	
.3000	.065	-.034	.024	-.096		.044	-.243	.073	-.218	.089	-.170	.099	.3000	
.3500	.029		.059	.128		.075	-.216	.101	-.201				.3500	
.4000	-.014		.062	.107		.077	-.196	.105	-.184				.4000	
.4500	-.029		.111	.083		.149	-.177	.148					.4500	
.5000	-.060	-.010	.094	.053		.143	-.147	.169	-.160	.184	-.164	.206	.5000	
.6000	-.102	-.012	.110	.026		.181	-.013	.216	-.136	.232	-.163	.241	.6000	
.7000	-.114	-.006	.143	.023		.181	-.085	.220	-.113	.239	-.149	.247	.7000	
.8000	-.099	-.017	.136	.004		.049	-.218	.092	-.238	.151	-.248	.161	.8000	
.9000	-.094	-.016	.113	-.011		.157	-.016	.190	-.075	.234	-.136		.9000	
$\alpha = 00$														
.0125	.501	-.366			.262	.426	-.186	.369	-.152	.332	-.111	.208	.0125	
.0250	.414	-.359				.368	-.189	.343	-.152				.0250	
.0500	.297	-.316	.309	-.264		.282	-.188	.256	-.152				.0500	
.1000	.212	-.033	.154	-.278		.142	-.188	.101	-.147	.163	-.110	.157	.1000	
.1500	.095	-.035	.013	-.230		.033	-.197	.014	-.138	.048	-.108	.113	.1500	
.2000	.033	-.017	.024	-.199		.032	-.194	.037	-.147	.034	-.108	.023	.2000	
.2500	.025		.037	-.130		.071	-.194	.089	-.157	.087	-.105	.100	.2500	
.3000	.031	-.009	.067	.138		.100	-.163	.135	-.147	.130	-.101	.149	.3000	
.3500	-.009		.095	.138		.124	-.140	.156	-.130	.172	-.099	.167	.3500	
.4000	-.048		.097	.134		.118	-.115	.153	-.104	.179	-.094	.181	.4000	
.4500	-.063		.146	.113		.183	-.073	.188					.4500	
.5000	-.089	-.046	.132	.087		.177	-.016	.203	-.073	.232	-.087	.248	.5000	
.6000	-.129	-.051	.170	.058		.218	-.107	.251	-.037	.273	-.083	.285	.6000	
.7000	-.140	-.031	.174	.056		.216	-.104	.258	-.019	.277	-.066	.291	.7000	
.8000	-.125	-.015	.164	.038		.061	-.255	.006	-.272	.063	-.293	.094	.8000	
.9000	-.122	-.022	.140	.023		.194	-.039	.240	-.005	.268	-.060		.9000	
$\alpha = 02$														
.0125	.464	-.343			.402	.512	-.342	.342	-.091	.292	-.043		.0125	
.0250	.377	-.334			.222	.391	-.133	.306	-.091				.0250	
.0500	.262	-.241	.271	-.222		.249	-.133	.223	-.089	.202	-.042	.169	.0500	
.1000	.167	-.055	.120	-.224		.102	-.132	.065	-.084	.129	-.042	.120	.1000	
.1500	.065	-.062	.029	-.173		.011	-.140	.029	-.076	.003	-.041	.077	.1500	
.2000	.001	-.048	.066	-.127		.080	-.132	.082	-.084	.080	-.039	.068	.2000	
.2500	-.005		.083	.093		.122	-.124	.132	-.091	.130	-.034	.120	.2500	
.3000	.003	-.023	.108	.176		.146	-.087	.184	-.076	.174	-.025	.186	.3000	
.3500	-.047		.126	.148		.169	-.061	.204	-.048	.209	-.022	.210	.3500	
.4000	-.081		.132	.155		.161	-.015	.203	-.018	.218	-.010	.236	.4000	
.4500	-.090		.184	.142		.208	-.050	.235		.258	-.006	.272	.4500	
.5000	-.117	-.075	.170	.114		.215	-.118	.247	-.019	.267	-.001	.280	.5000	
.6000	-.156	-.083	.200	.092		.251	-.147	.276	-.062	.306	-.007	.313	.6000	
.7000	-.167	-.062	.204	.084		.255	-.118	.286	-.067	.316	-.026	.317	.7000	
.8000	-.148	-.046	.193	.065		.082	-.286	.076	-.314	.027	-.322	.019	.8000	
.9000	-.148	-.053	.167	.055		.224	-.062	.277	-.060	.306	-.028		.9000	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
	$\alpha = 04$													
.0125	.424	-.311			.372	-.062	.306	-.016	.248	.043			.0125	
.0250	.341	-.295	-.166		.295	-.064	.266	-.018					.0250	
.0500	.226	-.124	.230	-.171	.213	-.062	.179	-.015	.161	.043	.117	.017	.0500	
.1000	.126	.081	.074	-.153	.065	-.060	.027	-.008	.082	.045	.066	.027	.1000	
.1500	.031	.092	-.065	-.086	.052	-.060	-.072	.001	-.032	.048	.024	.028	.1500	
.2000	-.034	.084	-.118	.046	.125	-.049	-.127	.001	-.131	.053	-.108	.030	.2000	
.2500	-.030		-.135	.232	.171	-.028	-.175	.009	-.175	.066	-.175	.050	.2500	
.3000	-.024	.057	-.146	.193	.199	.016	-.223	.032	-.208	.079	-.218	.074	.3000	
.3500	-.089		-.153	.177	.222	.060	-.248	.064	-.244	.090	-.243	.098	.3500	
.4000	-.110		-.167	.186	.210	.117	-.253	.105	-.251	.101	-.269	.139	.4000	
.4500	-.120		-.214	.176	.245	.188	-.282		-.288	.108	-.310	.145	.4500	
.5000	-.144	.115	-.210	.149	.234	.205	-.293	.128	-.297	.117	-.325	.160	.5000	
.6000	-.180	.123	-.225	.133	.281	.168	-.328	.149	-.336	.111	-.351	.144	.6000	
.7000	-.191	.101	-.229	.123	.298	.140	-.318	.135	-.346	.118	-.356	.143	.7000	
.8000	-.171	.083	-.217	.102	.117	.309	.124	-.344	.104	-.358	.128	.8000		
.9000	-.173	.091	-.194	.093	-.255	.091	-.310	.097	-.344	.100			.9000	
$\alpha = 06$														
.0125	.385	-.279			.336	.002	.269	.062	.197	.118	.067	.175	.0125	
.0250	.308	-.292			.111	.260	.000	.226	.061	.114	.118	.067	.0250	
.0500	.192	.000	.194	-.114	.170	.004	.141	.064	.114	.118	.067	.175	.0500	
.1000	.092	.099	.039	-.070	.032	.013	-.009	.069	.043	.127	.019	.192	.1000	
.1500	-.008	.113	-.099	.043	-.088	.025	-.103	.085	-.064	.145	-.024	.198	.1500	
.2000	-.064	.121	-.153	.210	.157	.047	-.165	.105	-.166	.169	-.142	.201	.2000	
.2500	-.038		-.173	.243	.208	.081	-.208	.119	-.220	.187	-.203	.225	.2500	
.3000	-.062	.095	-.179	.212	.236	.135	-.254	.145	-.247	.203	-.246	.238	.3000	
.3500	-.125		-.177	.208	.262	.182	-.278	.175	-.277	.209	-.266	.253	.3500	
.4000	-.138		-.193	.217	.262	.216	-.285	.204	-.285	.216	-.291	.259	.4000	
.4500	-.147		-.243	.210	.295	.248	-.315		.318	.210	-.333	.243	.4500	
.5000	-.169	.156	-.237	.182	.254	.232	-.321	.201	-.326	.206	-.343	.235	.5000	
.6000	-.202	.163	-.245	.172	.301	.192	-.357	.201	-.363	.178	-.369	.190	.6000	
.7000	-.213	.138	-.249	.160	.322	.168	-.364	.170	-.373	.173	-.375	.187	.7000	
.8000	-.191	.119	-.236	.140		.149	-.342	.153	-.372	.145	-.380	.164	.8000	
.9000	-.193	.130	-.213	.131	-.280	.127	-.326	.122	-.372	.135			.9000	
$\alpha = 08$														
.0125	.341	-.242			.290	.084	.214	.146	.128	.224			.0125	
.0250	.258	-.161			.050	.212	.081	.173	.140	.280	-.063	.385	.0250	
.0500	.154	.090	.149	-.038	.130	.090	.093	.153	.058	.249	-.027	.324	.0500	
.1000	.055	.136	.000	.047	-.013	.105	-.048	.175	-.004	.280			.1000	
.1500	-.043	.154	-.138	.181	.127	.134	-.147	.221	-.113	.309	-.097	.352	.1500	
.2000	-.094	.160	-.191	.280	.191	.181	-.201	.254	-.193	.307	-.209	.309	.2000	
.2500	-.064		-.215	.269	.242	.226	-.244	.258	-.253	.315	-.262	.318	.2500	
.3000	-.102	.167	-.224	.250	.275	.268	-.289	.267	-.289	.311	-.295	.317	.3000	
.3500	-.157		-.210	.250	.301	.273	-.311	.273	-.322	.304	-.312	.325	.3500	
.4000	-.169		-.222	.258	.307	.267	-.321	.283	-.324	.299	-.328	.313	.4000	
.4500	-.176		-.270	.252	.340	.273	-.347		.351	.282	-.368	.297	.4500	
.5000	-.195	.207	-.273	.225	.333	.256	-.357	.255	-.358	.277	-.384	.277	.5000	
.6000	-.224	.210	-.277	.215	.319	.228	-.390	.241	-.389	.226	-.406	.239	.6000	
.7000	-.235	.181	-.276	.199	.340	.203	-.398	.214	-.399	.220	-.410	.223	.7000	
.8000	-.214	.162	-.262	.179		.189	-.397	.194	-.401	.182	-.412	.204	.8000	
.9000	-.216	.171	-.240	.175	-.311	.162	-.368	.157	-.400	.177			.9000	
$\alpha = 10$														
.0125	.299	-.189			.037	.157	.168	.149	.210	.012	.379		.0125	
.0250	.216	-.087				.070	.087	.194	.040	.297	-.028	.381	-.179	.0408
.0500	.111	.153	.114			.060	.226	-.092	.335	-.069	.378	-.175	.427	.1000
.1000	.018	.171	-.041	.175		.060	.226	-.092	.335	-.069	.378	-.175		.1000
.1500	-.070	.191	-.166	.287		.159	.274	-.188	.358	-.148	.387	-.184	.396	.1500
.2000	-.116	.198	-.224	.317		.225	.310	-.242	.355	-.234	.364	-.257	.364	.2000
.2500	-.092		-.253	.304	.270	.311	-.275	.330	-.289	.372	-.303	.361	-.2500	
.3000	-.134	.210	-.271	.291	.302	.327	-.318	.330	-.320	.361	-.330	.363	-.3000	
.3500	-.174		-.253	.294	.331	.307	-.340	.325	-.350	.352	-.345	.375	-.3500	
.4000	-.191		-.245	.299	.335	.297	-.348	.325	-.354	.349	-.366	.358	-.4000	
.4500	-.201		-.285	.293	.369	.305	-.372	.335	-.384	.322	-.393	.346	-.4500	
.5000	-.217	.251	-.292	.263	.368	.296	-.380	.295	-.388	.319	-.404	.329	-.5000	
.6000	-.245	.251	-.302	.259	.373	.267	-.410	.278	-.416	.268	-.421	.293	-.6000	
.7000	-.253	.222	-.295	.242	.356	.238	-.417	.248	-.421	.255	-.426	.277	-.7000	
.8000	-.235	.203	-.282	.224		.230	-.417	.228	-.421	.222	-.426	.261	-.8000	
.9000	-.235	.215	-.259	.216	-.333	.201	-.293	.197	-.418	.219			.9000	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN												X/C		
	0.050		0.200		0.350		0.500		0.700		0.900				
	U	L	U	L	U	L	U	L	U	L	U	L			
	$\alpha = 12$														
.0125	.254	-.115			.176	.259	.045	.372					.0125		
.0250	.174	.015			.152	.097	.295	.005	.384	-.171	.467		.0250		
.0500	.077	.193	-.077		.201	.036	.356	-.033	.409	-.160	.445	-.321	.458	.0500	
.1000	-.019	.213	-.080		.297	-.096	.375	-.127	.409	-.167	.428	-.320	.467	.1000	
.1500	-.103	.233	-.203		.362	-.201	.372	-.220	.418	-.216	.435	-.326	.431	.1500	
.2000	-.145	.251	-.256		.356	-.255	.377	-.278	.415	-.275	.418	-.361	.406	.2000	
.2500	-.135		-.286		.350	-.297	.365	-.312	.384	-.318	.423	-.388	.402	.2500	
.3000	-.149	.255	-.310		.342	-.330	.372	-.350	.386	-.350	.419	-.404	.413	.3000	
.3500	-.207		-.303		.342	-.356	.345	-.368	.376	-.379	.405	-.417	.422	.3500	
.4000	-.219		-.270		.346	-.367	.340	-.375	.373	-.384	.400	-.424	.413	.4000	
.4500	-.223		-.305		.338	-.395	.348	-.398		-.411	.377	-.443	.393	.4500	
.5000	-.234	.297	-.312		.311	-.395	.342	-.408	.341	-.416	.366	-.447	.383	.5000	
.6000	-.262	.300	-.330		.300	-.415	.310	-.436	.326	-.443	.308	-.460	.342	.6000	
.7000	-.271	.271	-.317		.281	-.410	.283	-.441	.295	-.450	.305	-.464	.330	.7000	
.8000	-.250	.251	-.302		.267	-.441	.273	-.449	.268	-.449	.271	-.462	.339	.8000	
.9000	-.253	.257	-.280		.256	-.312	.241	-.298	.240	-.422	.271			.9000	
$\alpha = 14$															
.0125	.205	-.026			.300	.075	.376	-.076	.472					.0125	
.0250	.130	.096			.035	.339	-.026	.493	-.129	.478	-.292	.497	-.393	.527	.0250
.0500	.043	.251			.130	.375	-.135	.441	-.182	.455	-.290	.471	-.392	.531	.0500
.1000	-.048	.264			.232	.408	-.236	.430	-.257	.468	-.313	.488	-.393	.495	.1000
.1500	-.132	.288			.284	.402	-.293	.435	-.307	.466	-.344	.468	-.413	.473	.2000
.2000	-.175	.304			.313	.395	-.331	.416	-.340	.439	-.374	.467	-.432	.482	.2500
.2500	-.157		.304		.340	.389	-.355	.421	-.374	.431	-.394	.461	-.445	.486	.3000
.3000	-.241		.349		.395	.380	-.395	.394	-.426	.411	-.454	.452	-.452	.494	.3500
.4000	-.239		.307		.391	-.389	.389	-.406	.426	-.414	.442	-.428	.490	.4000	
.4500	-.240		.323		.377	-.416	.405	-.422		-.435	.426	-.471	.484	.4500	
.5000	-.253	.347			.329	.355	-.422	.391	-.427	.388	-.439	.426	-.477	.484	.5000
.6000	-.281	.351			.356	-.433	.359	-.452	.368	-.458	.379	-.467	.507	.6000	
.7000	-.285	.315	-.337		.327	-.442	.328	-.457	.346	-.466	.378	-.466	.538	.7000	
.8000	-.265	.296	-.322		.312	-.453	.320	-.453	.344	-.413	.350	-.462	.488	.8000	
.9000	-.269	.301	-.299		.300	-.326	.313	-.320	.309	-.358	.361			.9000	
$\alpha = 16$															
.0125		.073			.388	-.050	.493	-.200	.545					.0125	
.0250		.178			.402	-.110	.494	-.227	.543	-.389	.593			.0250	
.0500	.008	.300	-.018		.466	-.174	.482	-.268	.503	-.374	.548	-.463	.641	.0500	
.1000	-.077	.322	-.168		.462	-.325	.495	-.320	.522	-.387	.567	-.465	.632	.1000	
.1500	-.159	.345	-.272		.453	-.268	.495	-.320	.527	-.408	.541	-.464	.636	.2000	
.2000	-.196	.359	-.313		.442	-.325	.499	-.351	.527	-.425	.550	-.472	.637	.3000	
.2500	-.190		.341		.454	-.357	.470	-.374	.488	-.425	.539	-.472	.640	.3500	
.3000	-.166	.364	-.366		.446	-.381	.472	-.399	.479	-.439	.537	-.474	.642	.4000	
.4000	-.266		.380		.455	-.407	.444	-.416	.479	-.452	.537	-.474	.640	.4500	
.4500	-.253		.353		.444	-.414	.444	-.424	.497	-.455	.524	-.474	.625	.5000	
.5000	-.268	.398	-.348		.412	-.436	.438	-.447	.479	-.468	.510	-.475	.583	.5000	
.6000	-.295	.404	-.372		.409	-.450	.424	-.468	.460	-.473	.490	-.468	.540	.6000	
.7000	-.300	.366	-.355		.381	-.462	.420	-.474	.441	-.419	.546	-.466	.521	.7000	
.8000	-.278	.353	-.338		.407	-.449	.418	-.393	.415	-.394	.546	-.461	.475	.8000	
.9000	-.283	.384	-.318		.394	-.385	.389	-.346	.401	-.385	.495			.9000	
$\alpha = 18$															
.0125	.088	.147			.466	.193	.562	-.320	.601	-.448	.657			.0125	
.0250	.058	.293			.472	-.203	.553	-.317	.579	-.438	.650	-.466	.682	.0500	
.0500	-.025	.334	-.077		.498	-.401	.521	-.430	.582	-.467	.686	-.455	.647	.3000	
.1000	-.106	.380	-.193		.533	-.235	.541	-.336	.567	-.433	.638	-.462	.678	.1000	
.1500	-.183	.400	-.304		.497	-.306	.553	-.368	.628	-.436	.644	-.461	.692	.1500	
.2000	-.220	.414	-.345		.511	-.351	.547	-.393	.624	-.447	.658	-.459	.642	.2000	
.2500	-.216		.367		.516	-.377	.517	-.412	.587	-.460	.679	-.457	.645	.2500	
.3000	-.178	.432	-.389		.498	-.401	.521	-.430	.582	-.467	.686	-.455	.647	.3000	
.3500	-.274		.405		.510	-.422	.499	-.441	.587	-.465	.692	-.451	.648	.3500	
.4000	-.270		.395		.491	-.434	.527	-.448	.593	-.465	.689	-.451	.633	.4000	
.4500	-.270		.389		.477	-.449	.557	-.459	.557	-.453	.660	-.448	.592	.3000	
.5000	-.280	.446	-.376		.470	-.454	.545	-.465	.557	-.453	.660	-.448	.553	.6000	
.6000	-.305	.461	-.393		.492	-.464	.531	-.480	.548	-.427	.599	-.444	.553	.6000	
.7000	-.312	.428	-.378		.493	-.472	.503	-.483	.565	-.432	.584	-.440	.538	.7000	
.8000	-.286	.460	-.360		.504	-.507	.507	-.393	.590	-.434	.534	-.441	.480	.8000	
.9000	-.292	.485	-.341		.479	-.375	.492	-.363	.530	-.433	.486			.9000	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING
[From reference 1]

(a) $M = 1.61$ - Concluded

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
	$a = 20$													
.0125	.028	.231	.403	.548	.263	.651	.417	.723	.454	.759	.398	.683	.0125	
.0250	-.015	.331	.400	-.134	.561	.275	.628	.418	.715	.453	.760	.400	.685	.0250
.0500	-.056	.400	-.227	.592	.284	.618	-.403	.707	-.451	.763	-.400	.685	.1000	.0500
.1000	-.138	.455	-.227	.592	.312	.613	-.404	.717	-.451	.763	-.401	.683	.1500	.1000
.1500	-.205	.474	-.320	.554	.362	.634	-.421	.733	-.453	.758	-.401	.683	.2000	.1500
.2000	-.241	.480	-.367	.582	.391	.653	-.435	.702	-.449	.752	-.405	.653	.2000	.2000
.2500	-.231	-.395	-.412	.580	.408	.657	-.447	.680	-.445	.746	-.405	.655	.2500	.2500
.3000	-.206	.505	-.412	.560	.423	.668	-.459	.703	-.436	.736	-.401	.654	.3000	.3000
.3500	-.274	-.424	-.421	.575	.438	.646	-.463	.719	-.432	.727	-.401	.660	.3500	.3500
.4000	-.294	-.421	-.421	.574	-.443	.657	-.466	.731	-.430	.714	-.400	.649	.4000	.4000
.4500	-.284	-.421	-.421	.612	-.460	.661	-.476	.718	-.424	.681	-.400	.609	.5000	.5000
.5000	-.290	.516	-.403	.629	-.464	.667	-.477	.718	-.424	.681	-.400	.609	.5000	.5000
.6000	-.314	.577	-.410	.621	-.469	.656	-.477	.683	-.414	.624	-.399	.578	.6000	.6000
.7000	-.320	.573	-.401	.612	-.477	.669	-.459	.654	-.410	.614	-.398	.548	.7000	.7000
.8000	-.291	.591	-.387	.624	-.433	.651	-.403	.604	-.409	.552	-.398	.487	.8000	.8000
.9000	-.301	.592	-.362	.628	-.410	.575	-.403	.536	-.408	.500				.9000

L-3000

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(b) M = 2.01

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
	$\alpha = -20$													
.0125	.853	-.325	.683	-.305	.630	-.323	.574	-.324					.0125	
.0250	.812	-.327	.713	-.303	.676	-.323	.630	-.323	.604	-.313			.0250	
.0500	.782	-.331	.689	-.303	.662	-.324	.628	-.323	.605	-.298	.583	-.267	.0500	
.1000	.700	-.287	.617	-.306	.626	-.324	.580	-.322	.576	-.313	.564	-.266	1000	
.1500	.564	-.185	.538	-.314	.576	-.324	.538	-.320	.544	-.313	.535	-.266	1500	
.2000	.462	-.083	.487	-.320	.499	-.323	.483	-.320	.480	-.310	.451	-.267	2000	
.2500	.446		.450	-.319	.425	-.322	.417	-.317	.391	-.313	.413	-.264	2500	
.3000	.437	-.248	.379	-.314	.377	-.324	.366	-.316	.357	-.309	.361	-.263	3000	
.3500	.416		.341	-.309	.356	-.324	.347	-.316	.334	-.303	.364	-.263	3500	
.4000	.359		.320	-.306	.335	-.325	.302	-.316	.320	-.295	.319	-.257	4000	
.4500	.342		.278	-.300	.256	-.324	.285		.288	-.288	.269	-.257	4500	
.5000	.288	-.185	.275	-.297	.269	-.324	.255	-.316	.263	-.282	.246	-.254	5000	
.6000	.239	-.174	.207	-.296	.197	-.323	.186	-.314	.193	-.278	.219	-.253	6000	
.7000	.221	-.175	.195	-.304	.179	-.320	.167	-.311	.183	-.292	.198	-.258	7000	
.8000	.203	-.165	.188	-.305	.155	-.315	.172	-.303	.186	-.295	.198	-.249	8000	
.9000	.221	-.174	.202	-.284	.180	-.308	.180	-.252	.189	-.292			.9000	
$\alpha = -18$														
.0125	.818	-.326	.666	-.303	.621	-.321	.564	-.321	.594	-.310			.0125	
.0250	.781	-.328	.690	-.303	.660	-.321	.614	-.321	.583	-.297	.554	-.265	.0250	
.0500	.726	-.335	.658	-.308	.630	-.322	.604	-.319	.544	-.311	.532	-.265	1000	
.1000	.643	-.300	.569	-.307	.578	-.321	.536	-.318					1500	
.1500	.504	-.164	.481	-.313	.523	-.322	.493	-.318	.502	-.310	.496	-.265	2000	
.2000	.419	-.026	.435	-.318	.447	-.321	.439	-.317	.487	-.308	.498	-.266	2500	
.2500	.394		.394	-.318	.380	-.323	.365	-.312	.348	-.310	.385	-.266	3000	
.3000	.381	-.247	.333	-.314	.325	-.323	.319	-.312	.313	-.308	.321	-.265	3500	
.3500	.363		.290	-.310	.302	-.325	.307	-.312	.282	-.305	.313	-.262	4000	
.4000	.310		.278	-.303	.291	-.323	.268	-.314	.266	-.300	.278	-.259	4500	
.4500	.287		.223	-.300	.194	-.323	.211		.222	-.298	.225	-.257	5000	
.5000	.244	-.188	.231	-.296	.221	-.320	.207	-.312	.223	-.294	.204	-.257	6000	
.6000	.193	-.169	.164	-.298	.154	-.313	.146	-.310	.152	-.285	.171	-.250	7000	
.7000	.177	-.167	.156	-.299	.184	-.306	.124	-.305	.138	-.283	.150	-.247	8000	
.8000	.155	-.154	.150	-.279	.129	-.308	.127	-.300	.138	-.279	.148	-.248	9000	
.9000	.177	-.165	.160	-.218	.133	-.294	.136	-.242	.142	-.275			.9000	
$\alpha = -16$														
.0125	.775	-.328	.661	-.304	.610	-.314	.553	-.312					.0125	
.0250	.741	-.329	.669	-.302	.642	-.314	.605	-.313	.584	-.300			.0250	
.0500	.670	-.336	.628	-.306	.604	-.315	.586	-.312	.562	-.290	.538		.0500	
.1000	.587	-.309	.525	-.307	.532	-.317	.501	-.310	.517	-.300	.512		1000	
.1500	.452	-.160	.427	-.314	.475	-.318	.453	-.310	.467	-.299	.477		1500	
.2000	.371	-.024	.382	-.318	.400	-.318	.387	-.310	.398	-.298	.366		2000	
.2500	.347		.343	-.315	.339	-.321	.321	-.304	.314	-.298	.322		2500	
.3000	.333	-.202	.287	-.310	.282	-.322	.267	-.304	.278	-.298	.286		3000	
.3500	.319		.243	-.304	.252	-.319	.264	-.305	.250	-.296	.274		3500	
.4000	.268		.235	-.299	.258	-.316	.236	-.307	.239	-.295	.231		4000	
.4500	.246		.180	-.292	.157	-.313	.169		.173	-.293			4500	
.5000	.207	-.163	.188	-.287	.169	-.308	.151	-.304	.172	-.292			5000	
.6000	.158	-.151	.124	-.280	.119	-.300	.109	-.300	.110	-.282			6000	
.7000	.139	-.155	.117	-.266	.093	-.290	.087	-.295	.103	-.269			7000	
.8000	.114	-.142	.113	-.211	.110	-.280	.085	-.291	.105	-.258			8000	
.9000	.139	-.153	.119	-.144	.100	-.276	.096	-.228	.105	-.255			.9000	
$\alpha = -14$														
.0125	.734	-.326	.641	-.302	.592	-.304	.537	-.303	.564	-.289			.0125	
.0250	.701	-.326	.642	-.301	.611	-.305	.577	-.303	.531	-.280	.507	-.255	.0250	
.0500	.609	-.333	.591	-.305	.568	-.307	.549	-.301	.571	-.289	.479	-.256	1000	
.1000	.525	-.317	.475	-.309	.479	-.309	.450	-.299	.477	-.289	.445	-.259	1500	
.1500	.393	-.178	.370	-.315	.417	-.308	.398	-.298	.417	-.288	.325	-.261	2000	
.2000	.320	-.029	.325	-.309	.340	-.311	.336	-.298	.344	-.288	.325	-.258	2500	
.2500	.293		.288	-.305	.286	-.318	.271	-.292	.260	-.288	.279	-.258	3000	
.3000	.279	-.153	.233	-.299	.235	-.317	.210	-.297	.223	-.288	.242	-.258	3500	
.3500	.267		.196	-.289	.194	-.309	.196	-.295	.200	-.287	.233	-.256	4000	
.4000	.217		.187	-.283	.204	-.302	.190	-.293	.200	-.287	.200	-.252	4500	
.4500	.195		.131	-.273	.114	-.297	.128		.134	-.286	.121	-.250	5000	
.5000	.162	-.131	.140	-.266	.117	-.288	.108	-.286	.127	-.284	.113	-.247	5000	
.6000	.115	-.133	.082	-.255	.073	-.279	.054	-.281	.066	-.279	.083	-.243	6000	
.7000	.095	-.138	.079	-.184	.049	-.268	.045	-.273	.050	-.261	.066	-.235	7000	
.8000	.069	-.125	.073	-.115	.056	-.259	.041	-.268	.046	-.247	.060	-.227	8000	
.9000	.095	-.141	.077	-.119	.056	-.258	.046	-.220	.055	-.237			.9000	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(b) M = 2.01 - Continued

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
	<i>a</i> = -12													
.0125	.695	-.319	.616	-.298	.576	-.294	.522	-.290	.545	-.276	.582	-.291	.0125	
.0250	.651	-.321	.612	-.295	.582	-.295	.555	-.291	.498	-.271	.475	-.272	.0250	
.0500	.549	-.321	.551	-.300	.529	-.295	.513	-.289	.441	-.276	.440	-.272	.0500	
.1000	.467	-.318	.426	-.304	.430	-.298	.401	-.288	.441	-.276	.409	-.272	.1000	
.1500	.344	-.177	.313	-.305	.362	-.298	.349	-.285	.371	-.276	.409	-.272	.1500	
.2000	.271	-.115	.271	-.295	.286	-.303	.290	-.285	.299	-.275	.274	-.269	.2000	
.2500	.245		.238	-.284	.237	-.307	.226	-.281	.214	-.275	.222	-.269	.2500	
.3000	.230	-.112	.184	-.274	.188	-.297	.163	-.286	.173	-.276	.186	-.268	.3000	
.3500	.222		.154	-.261	.151	-.289	.144	-.283	.142	-.275	.179	-.264	.3500	
.4000	.173		.144	-.252	.155	-.277	.146	-.276	.145	-.273	.162	-.265	.4000	
.4500	.151		.088	-.242	.077	-.270	.088		.092	-.272	.088	-.265	.4500	
.5000	.121	-.121	.098	-.231	.075	-.264	.069	-.265	.085	-.270	.074	-.264	.5000	
.6000	.079	-.110	.043	-.201	.031	-.250	.012	-.256	.028	-.265	.041	-.258	.6000	
.7000	.060	-.118	.041	-.095	.014	-.235	-.004	-.248	.011	-.251	.025	-.250	.7000	
.8000	.036	-.109	.037	-.090		-.227	.003	-.243	.006	-.228	.020	-.236	.8000	
.9000	.058	-.123	.041	-.103	.018	-.231	.008	-.220	.008	-.211			.9000	
<i>a</i> = -10														
.0125	.661	-.308	.585	-.288	.559	-.281	.510	-.275	.513	-.264	.452	-.279	.0125	
.0250	.611	-.312	.582	-.284	.547	-.282	.516	-.276					.0250	
.0500	.502	-.304	.516	-.288	.489	-.282	.464	-.274	.458	-.258			.0500	
.1000	.417	-.301	.389	-.293	.381	-.284	.348	-.274	.400	-.264	.405	-.280	.1000	
.1500	.295	-.134	.269	-.294	.309	-.287	.295	-.272	.325	-.264	.376	-.279	.1500	
.2000	.217	-.099	.225	-.267	.239	-.293	.246	-.274	.263	-.263	.235	-.281	.2000	
.2500	.204		.192	-.248	.198	-.286	.185	-.273	.179	-.264	.178	-.281	.2500	
.3000	.187	-.044	.147	-.236	.148	-.275	.122	-.273	.132	-.264	.141	-.283	.3000	
.3500	.180		.112	-.226	.106	-.262	.099	-.263	.102	-.262	.129	-.281	.3500	
.4000	.133		.108	-.216	.112	-.250	.097	-.253	.098	-.256	.118	-.272	.4000	
.4500	.112		.056	-.201	.040	-.238	.050		.053	-.253	.048	-.267	.4500	
.5000	.087	-.101	.061	-.185	.044	-.229	.033	-.239	.046	-.248	.035	-.259	.5000	
.6000	.042	-.087	.013	-.136	-.006	-.213	-.021	-.224	-.006	-.239	.003	-.252	.6000	
.7000	.026	-.097	.006	-.056	-.017	-.200	-.036	-.215	.017	-.229	-.013	-.239	.7000	
.8000	.002	-.086	.008	-.074		-.194	-.037	-.208	-.024	-.215	-.016	-.235	.8000	
.9000	.025	-.100	.013	-.088	-.015	-.197	-.027	-.200	-.025	-.184			.9000	
<i>a</i> = -08														
.0125	.629	-.303	.564	-.280	.546	-.269	.496	-.259	.496	-.247	.412	-.276	.0125	
.0250	.570	-.301	.551	-.277	.526	-.270	.490	-.259					.0250	
.0500	.459	-.300	.479	-.282	.462	-.270	.431	-.259	.431	-.244			.0500	
.1000	.369	-.273	.343	-.286	.348	-.271	.309	-.258	.365	-.249	.374	-.277	.1000	
.1500	.244	-.085	.217	-.273	.270	-.276	.252	-.258	.283	-.248	.339	-.278	.1500	
.2000	.173	-.077	.178	-.237	.199	-.276	.199	-.262	.228	-.250	.204	-.280	.2000	
.2500	.164		.139	-.220	.150	-.266	.138	-.255	.145	-.250	.137	-.281	.2500	
.3000	.149	-.020	.105	-.201	.115	-.248	.089	-.250	.101	-.248	.103	-.281	.3000	
.3500	.142		.069	-.185	.075	-.230	.063	-.233	.067	-.240	.088	-.278	.3500	
.4000	.097		.073	-.169	.072	-.215	.055	-.219	.060	-.233	.078	-.259	.4000	
.4500	.077		.019	-.149	.012	-.203	.010		.015	-.225	.013	-.240	.4500	
.5000	.054	-.081	.024	-.130	.018	-.190	-.005	-.204	.010	-.217	.000	-.224	.5000	
.6000	.008	-.066	-.015	-.069	-.029	-.172	-.051	-.189	-.039	-.207	-.033	-.218	.6000	
.7000	-.005	-.079	-.024	-.033	-.044	-.161	-.064	-.178	-.050	-.197	-.047	-.208	.7000	
.8000	.024	-.071	-.024	-.052	-.156	-.070	-.052	-.172	-.055	-.193	-.052	-.207	.8000	
.9000	-.004	-.080	-.017	-.065	-.047	-.156	-.062	-.170	-.055	-.177			.9000	
<i>a</i> = -06														
.0125	.592	-.288	.543	-.266	.520	-.252	.475	-.243	.463	-.233			.0125	
.0250	.530	-.283	.519	-.264	.487	-.253	.464	-.242	.401	-.229	.367	-.272	.0250	
.0500	.415	-.285	.444	-.268	.420	-.253	.396	-.242					.0500	
.1000	.322	-.232	.297	-.272	.298	-.253	.267	-.241	.322	-.235	.335	-.271	.1000	
.1500	.202	-.042	.173	-.243	.211	-.263	.199	-.242	.232	-.236	.300	-.272	.1500	
.2000	.128	-.057	.124	-.208	.144	-.253	.154	-.246	.178	-.238	.170	-.272	.2000	
.2500	.124		.093	-.181	.097	-.233	.092	-.231	.104	-.238	.095	-.275	.2500	
.3000	.110	-.009	.064	-.163	.064	-.211	.043	-.218	.060	-.226	.062	-.275	.3000	
.3500	.106		.033	-.144	.030	-.192	.024	-.199	.029	-.210	.045	-.264	.3500	
.4000	.065		.037	-.130	.026	-.175	.018	-.182	.022	-.199	.023	-.215	.4000	
.4500	.037		.012	-.100	.030	-.158	-.023		.022	-.187	.024	-.188	.4500	
.5000	.022	-.059	-.011	-.078	-.027	-.147	-.038	-.161	-.027	-.177	-.037	-.181	.5000	
.6000	-.019	-.046	-.045	-.031	-.065	-.129	-.086	-.144	-.072	-.171	-.068	-.175	.6000	
.7000	-.034	-.055	-.057	-.025	-.084	-.121	-.096	-.137	-.082	-.160	-.080	-.172	.7000	
.8000	-.042	-.060	-.051	-.042	-.111	-.095	-.132	-.086	-.159	-.086	-.172	-.000	.8000	
.9000	-.031	-.057	-.048	-.054	-.080	-.112	-.095	-.130	-.088	-.155			.9000	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(b) M = 2.01 - Continued

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = -04$														
.0125	.554	-.277	.515	-.251	.499	-.237	.453	-.224	.429	-.216	.395	-.204	.0125	
.0250	.487	-.267	.488	-.249	.457	-.240	.432	-.226	.367	-.216	.335	-.204	.0250	
.0500	.370	-.264	.405	-.252	.384	-.237	.363	-.225	.367	-.216	.335	-.204	.0500	
.1000	.277	-.187	.258	-.250	.261	-.239	.226	-.226	.289	-.220	.296	-.261	.1000	
.1500	.158	-.017	.129	-.208	.169	-.246	.154	-.229	.189	-.222	.265	-.261	.1500	
.2000	.092	-.037	.077	-.159	.097	-.220	.107	-.219	.134	-.223	.134	-.261	.2000	
.2500	.084		.051	-.127	.048	-.193	.049	-.194	.069	-.204	.064	-.260	.2500	
.3000	.074	.011	.020	-.113	.020	-.165	-.002	-.173	.026	-.180	.030	-.243	.3000	
.3500	.069		-.003	-.083	-.010	-.143	-.018	-.151	-.007	-.159	.013	-.202	.3500	
.4000	.028		-.003	-.063	-.015	-.125	-.025	-.131	-.016	-.146	-.009	-.153	.4000	
.4500	.009		.038	-.037	-.067	-.106	-.063	-.079	-.111	-.063	-.127	-.069	.4500	
.5000	-.010	-.029	-.045	-.017	-.063	-.093	-.079	-.111	-.111	-.127	-.130	-.5000		
.6000	-.052	-.022	-.075	-.000	-.097	-.079	-.120	-.092	-.104	-.124	-.102	-.129	.6000	
.7000	-.060	-.029	-.085	-.007	-.115	-.067	-.130	-.088	-.114	-.116	-.111	-.127	.7000	
.8000	-.063	-.039	-.079	-.024	-.060	-.132	-.083	-.116	-.115	-.117	-.131	-.8000		
.9000	-.051	-.039	-.071	-.033	-.110	-.062	-.123	-.084	-.117	-.112	-.000	-.9000		
$\alpha = -02$														
.0125	.441	-.214	.434	-.206	.428	-.208	.433	-.206	.348	-.204	.239	-.196	.0125	
.0250	.357	-.195	.392	-.205	.365	-.207	.340	-.217	.270	-.199	.195	-.199	.0250	
.0500	.292	-.179	.294	-.209	.283	-.208	.271	-.216	.203	-.205	.195	-.199	.1000	
.1000	.151	.008	.149	-.098	.155	-.208	.123	-.216	.166	-.175	.159	-.199	.1500	
.1500	.052	.044	.026	-.018	.051	-.085	.042	-.106	.085	-.175	.038	-.182	.2000	
.2000	-.002	.045	-.030	.024	-.009	-.019	-.011	-.030	-.009	-.075	.028	-.094	.2500	
.2500	-.008		.058	.057	-.061	.018	-.055	.015	-.033	-.030	-.023	-.094	.3000	
.3000	-.006	.079	-.084	.070	-.089	.049	-.097	.046	-.067	.011	-.066	-.037	.3500	
.3500	-.019		-.103	.081	-.115	.066	-.117	.071	-.097	.050	-.090	-.012	.4000	
.4000	-.055		-.099	.094	-.122	.075	-.127	.081	-.106	.076	-.110	.011	.4500	
.4500	-.075		-.121	.093	-.159	.084	-.154	-.139	-.079	-.144	.029	-.182	.5000	
.5000	-.086	.059	-.120	.083	-.159	.091	-.164	.093	-.145	.091	-.156	.037	.6000	
.6000	-.116	.063	-.151	.077	-.186	.071	-.199	.085	-.179	.082	-.180	.042	.7000	
.7000	-.122	.056	-.151	.064	-.186	.059	-.208	.067	-.191	.074	-.185	.054	.8000	
.8000	-.116	.037	-.146	.050	-.049	-.210	.060	-.192	.059	-.189	.049	-.8000		
.9000	-.112	.039	-.135	.037	-.182	.033	-.205	.043	-.194	.052	-.000	-.9000		
$\alpha = 00$														
.0125	.479	-.237	.465	-.216	.458	-.204	.407	-.195	.380	-.184	.272	-.222	.0125	
.0250	.402	-.223	.426	-.215	.397	-.204	.378	-.195	.305	-.183	.228	-.222	.0250	
.0500	.291	-.214	.336	-.217	.323	-.203	.302	-.195	.234	-.188	.228	-.222	.1000	
.1000	.194	-.064	.187	-.177	.190	-.211	.161	-.203	.116	-.186	.194	-.223	.1500	
.1500	.084	.020	.060	-.097	.084	-.164	.076	-.176	.116	-.186	.159	-.219	.2000	
.2000	.030	.013	.007	-.046	.023	-.111	.028	-.125	.054	-.128	.068	-.160	.2500	
.2500	.020		-.022	-.014	-.025	-.075	-.024	-.083	-.001	-.090	.002	-.160	.3000	
.3000	.018	.049	-.054	-.007	-.058	-.042	-.069	-.058	-.037	-.066	-.048	-.095	.3500	
.3500	.007		-.072	.033	-.084	-.020	-.090	-.030	-.064	-.043	-.069	-.063	.4000	
.4000	-.030		-.069	.051	-.091	-.001	-.097	-.007	-.079	-.027	-.083	-.043	.4500	
.4500	-.049		-.093	.060	-.129	.019	-.128	-.112	-.021	-.117	.031	-.157	.5000	
.5000	-.062	.028	-.097	.054	-.133	.031	-.140	.011	-.121	-.010	-.130	-.027	.6000	
.6000	-.095	.034	-.127	.049	-.156	.033	-.178	.028	-.156	-.013	-.157	-.028	.7000	
.7000	-.104	.024	-.131	.038	-.181	.030	-.187	.018	-.169	-.002	-.164	-.027	.8000	
.8000	-.099	.009	-.125	.022	-.021	-.189	.018	-.171	-.005	-.170	.035	-.8000		
.9000	-.095	.010	-.115	.012	-.160	.006	-.183	.009	-.171	-.003	-.003	-.9000		
$\alpha = 02$														
.0500	.250	-.183	.293	-.189	.283	-.172	.271	-.165	.803	-.157	.191	-.162	.0500	
.1000	.149	.019	.146	-.164	.154	-.172	.129	-.165	.803	-.155	.155	-.159	.1000	
.1500	.054	.047	.016	-.051	.046	-.143	.048	-.161	.803	-.155	.031	-.159	.2000	
.2000	.006	.043	-.032	.014	-.023	-.043	-.014	-.058	.803	-.103	.005	-.068	.3000	
.2500	-.008		-.064	.060	-.064	-.007	-.054	.803	-.038	-.029	-.144	-.2500		
.3000	.001	.078	-.088	.072	-.098	-.038	-.095	.803	-.005	-.090	-.020	-.3500		
.3500	-.020		-.103	.087	-.120	.065	-.115	.803	-.034	-.093	-.020	-.4000		
.4000	-.054		-.095	.096	-.132	-.078	-.125	.803	-.056	-.111	.024	-.4500		
.4500	-.069		-.118	.090	-.157	.093	-.152	.803	-.063	-.147	.044	-.5000		
.5000	-.083	.058	-.122	.083	-.165	.100	-.162	.803	-.144	.072	-.156	-.048	.6000	
.6000	-.115	.060	-.153	.063	-.186	.071	-.195	.803	-.176	.070	-.182	.041	.7000	
.7000	-.123	.054	-.156	.062	-.190	.065	-.206	.803	-.190	.067	-.188	.047	.8000	
.8000	-.115	.031	-.149	.047	-.054	-.205	.803	-.190	.056	-.194	.041	-.8000		
.9000	-.110	.035	-.138	.034	-.180	-.037	-.201	.803	-.190	.056	-.000	-.9000		

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(b) M = 2.01 - Continued

X/C	FRACTION OF SEMISPAN												X/C					
	0.050		0.200		0.350		0.500		0.700		0.900							
	U	L	U	L	U	L	U	L	U	L	U	L						
	$\alpha = 04$																	
.0125	.403	-.184	.403	-.194	.398	-.210	.349	-.211	.312	-.181	.203	-.152	.0125					
.0250	.320	-.162	.362	-.192	.329	-.208	.312	-.211	.236	-.179	.203	-.152	.0250					
.0500	.214	-.124	.255	-.194	.253	-.212	.236	-.213	.236	-.179	.203	-.152	.0500					
.1000	.112	-.050	.110	-.010	.121	-.150	.093	-.185	.173	-.182	.157	-.155	.1000					
.1500	.022	-.072	-.007	.062	.018	-.015	.010	-.049	.065	-.084	.122	-.153	.1500					
.2000	-.030	.071	-.042	.099	-.042	.093	-.042	.087	-.022	.027	.008	-.086	.2000					
.2500	-.035	-.093	.128	-.089	.117	-.082	.132	-.068	.104	-.054	.034	-.2500	.2500					
.3000	-.027	.110	-.120	.121	-.120	.137	-.123	.150	-.094	.148	-.089	.087	-.3000	.3000				
.3500	-.051	-.137	.116	-.144	.137	-.144	.162	-.123	.174	-.115	.128	-.128	.3500	.3500				
.4000	-.087	-.131	.126	-.155	.138	-.155	.147	-.133	.175	-.138	.158	-.158	.4000	.4000				
.4500	-.098	-.145	.128	-.185	.134	-.179	.163	-.171	.172	-.172	.165	-.165	.4500	.4500				
.5000	-.109	.091	-.144	.118	-.186	.140	-.189	.159	-.170	.172	-.179	.161	-.5000	.5000				
.6000	-.137	.096	-.174	.109	-.211	.102	-.222	.133	-.201	.139	-.206	.143	-.6000	.6000				
.7000	-.146	.088	-.173	.095	-.218	.093	-.231	.108	-.212	.125	-.212	.133	-.7000	.7000				
.8000	-.135	.065	-.166	.083	-.234	.084	-.234	.099	-.214	.108	-.218	.113	-.8000	.8000				
.9000	-.135	.067	-.155	.068	-.204	.065	-.230	.079	-.213	.099				.9000				
$\alpha = 06$																		
.0125	.364	-.157	.375	-.164	.367	-.173	.315	-.166	.277	-.065	.168	-.051	.0125					
.0250	.279	-.132	.334	-.163	.294	-.172	.274	-.169	.277	-.065	.168	-.051	.0250					
.0500	.178	-.054	.225	-.160	.222	-.175	.200	-.171	.201	-.062	.144	-.055	.1000					
.1000	.077	.074	.044	.074	.079	-.034	.071	-.077	.144	-.051	.124	-.055	.1000					
.1500	-.004	.101	-.037	.122	-.008	.102	-.018	.107	.033	.039	.087	.069	.1500					
.2000	-.048	.096	-.087	.150	-.069	.159	-.070	.171	-.050	.098	-.016	.083	.2000					
.2500	-.053	-.116	.175	-.111	.173	-.109	.187	-.100	.149	-.076	.107	-.125	.3000					
.3000	-.043	.147	-.144	.165	-.140	.185	-.148	.201	-.131	.189	-.114	.123	.3000					
.3500	-.077	-.159	.158	-.164	.181	-.166	.211	-.151	.229	-.135	.141	-.135	.3500					
.4000	-.106	-.154	.169	-.174	.181	-.175	.200	-.157	.238	-.158	.175	-.158	.4000					
.4500	-.118	-.169	.167	-.201	.180	-.197	.184	-.208	.199	-.190	.228	-.187	.194					
.5000	-.129	.127	-.160	.154	-.204	.182	-.208	.199	-.190	.228	-.197	.200	.5000					
.6000	-.152	.130	-.189	.145	-.225	.140	-.238	.179	-.219	.189	-.223	.206	.6000					
.7000	-.157	.120	-.189	.131	-.235	.128	-.243	.148	-.229	.169	-.227	.197	.7000					
.8000	-.150	.097	-.180	.115	-.220	.098	-.245	.114	-.232	.148	-.232	.172	.8000					
.9000	-.151	.099	-.169	.094	-.220	.098	-.245	.114	-.232	.140				.9000				
$\alpha = 08$																		
.0125	.326	-.112	.339	-.115	.333	-.087	.281	-.003	.234	.114	.118	.128	.0125					
.0250	.240	-.089	.297	-.113	.258	-.086	.238	.005	.201	.168	.051	.0500	.0250					
.0500	.148	.018	.199	-.083	.186	-.088	.168	.001	.161	.118	.085	.223	.1000					
.1000	.049	.103	.032	.118	.046	.057	.037	.058	.107	.135	.085	.217	.1500					
.1500	-.030	.134	-.070	.188	-.040	.160	-.056	.125	.009	.173	.050	.216	.2000					
.2000	-.073	.129	-.116	.192	-.095	.216	-.098	.188	-.068	.199	-.054	.216	.2500					
.2500	-.067	-.145	.216	-.137	.230	-.135	.225	-.123	.231	-.101	.238	-.135	.3000					
.3000	-.069	.194	-.172	.204	-.165	.240	-.172	.251	-.156	.252	-.135	.280	.3500					
.3500	-.104	-.184	.200	-.191	.231	-.188	.271	-.176	.271	-.154	.280	-.154	.4000					
.4000	-.131	-.183	.210	-.199	.231	-.199	.260	-.187	.275	-.177	.300			.4000				
.4500	-.138	-.192	.206	-.222	.230	-.220	.206	-.269	.207	.304	.4500			.4500				
.5000	-.147	.167	-.181	.197	-.226	.229	-.226	.248	-.213	.267	-.216	.282	.5000			.5000		
.6000	-.170	.167	-.208	.186	-.245	.183	-.256	.227	-.237	.226	-.239	.245	.6000			.6000		
.7000	-.174	.156	-.211	.171	-.255	.170	-.261	.194	-.246	.216	-.243	.232	.7000			.7000		
.8000	-.165	.132	-.199	.155	-.207	.254	-.218	.274	-.218	.175	-.248	.191	-.248	.207	.8000			.8000
.9000	-.168	.134	-.186	.139	-.246	.136	-.262	.156	-.248	.182				.9000				
$\alpha = 10$																		
.0125	.286	-.085	.297	-.056	.292	.011	.238	.114	.180	.237	.063	.355	.0125					
.0250	.203	-.039	.258	-.049	.216	.011	.194	.126	.114	.249	.031	.405	.0250					
.0500	.117	.055	.166	-.018	.149	.020	.129	.126	.114	.278	.004	.385	.1000					
.1000	.021	.148	.001	.202	.020	.141	.012	.184	.067	.314	.044	.348	.1500					
.1500	-.052	.170	-.096	.237	-.071	.239	-.074	.242	-.016	.314	.025	.348	.2000					
.2000	-.087	.167	-.141	.239	-.124	.284	-.126	.286	-.092	.325	-.089	.348	.2500					
.2500	-.087	-.170	.267	-.160	.280	-.163	.298	-.142	.345	-.133	.350	-.133	.3000					
.3000	-.083	.239	-.192	.243	-.187	.288	-.191	.309	-.172	.342	-.161	.354	.3500					
.3500	-.127	-.205	.239	-.207	.281	-.208	.318	-.195	.344	-.180	.360	-.195	.3500					
.4000	-.147	-.207	.254	-.218	.274	-.218	.301	-.205	.333	-.195	.350	-.205	.4000					
.4500	-.155	-.216	.253	-.237	.280	-.236	-.223	.317	-.221	.341	-.221	-.221	.4500					
.5000	-.162	.209	-.203	.239	-.242	.273	-.242	.284	-.229	.309	-.229	.322	.5000					
.6000	-.181	.212	-.221	.225	-.260	.228	-.267	.267	-.251	.265	-.251	.285	.6000					
.7000	-.185	.203	-.226	.209	-.268	.213	-.272	.234	-.240	.257	-.255	.272	.7000					
.8000	-.177	.165	-.215	.192	-.266	.173	-.274	.216	-.262	.232	-.259	.251	.8000					
.9000	-.177	.172	-.200	.173	-.266	.176	-.272	.191	-.262	.219				.9000				

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED WING
[From reference 1]

(b) M = 2.01 - Continued

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = 12$														
.0125	.235	-.039	.251	.016	.246	.127	.185	.226	.099	.387			.0125	
.0250	.158	.023	.208	.025	.167	.129	.140	.231	.053	.436	-.060	.448	.0250	
.0500	.080	.110	.128	.059	.109	.158	.079	.283	.053				.0500	
.1000	-.015	.185	-.031	.278	-.007	.261	-.015	.327	.011	.429	-.067	.459	.1000	
.1500	-.079	.211	-.127	.290	-.097	.333	-.102	.366	-.049	.428	-.081	.442	.1500	
.2000	-.117	.213	-.167	.290	-.150	.347	-.152	.374	-.111	.401	-.142	.407	.2000	
.2500	-.120		-.196	.319	-.186	.326	-.184	.365	-.160	.405	-.175	.401	.2500	
.3000	-.104	.283	-.217	.286	-.209	.340	-.214	.372	-.189	.397	-.198	.404	.3000	
.3500	-.152		-.227	.291	-.233	.331	-.232	.370	-.210	.399	-.210	.415	.3500	
.4000	-.172		-.230	.304	-.240	.321	-.239	.351	-.218	.380	-.223	.400	.4000	
.4500	-.176		-.247	.302	-.258	.333	-.255		.239	.363	-.246	.393	.4500	
.5000	-.182	.258	-.235	.283	-.260	.319	-.263		.244	.352	-.253	.369	.5000	
.6000	-.202	.263	-.238	.269	-.276	.276	-.283		.215	.309	-.248	.334	.6000	
.7000	-.205	.246	-.244	.253	-.289	.260	-.289		.281	.271	-.271	.324	.7000	
.8000	-.194	.206	-.234	.234	-.241	.289	-.284		.264	-.272	.277	-.275	.299	.8000
.9000	-.198	.215	-.221	.213	-.258	.215	-.249		.238	-.272	.260		.9000	
$\alpha = 14$														
.0125	.197	.001	.213	.118	.200	.235	.127	.363					.0125	
.0250	.124	.091	.175	.134	.122	.242	.078	.374	-.018	.497			.0250	
.0500	.053	.161	.085	.224	.063	.315	.026	.437	-.093	.486	-.145	.502	.0500	
.1000	-.037	.231	-.050	.330	-.028	.375	-.048	.415	-.057	.472	-.151	.506	.1000	
.1500	-.098	.261	-.139	.341	-.119	.394	-.127	.454	-.103	.468	-.159	.483	.1500	
.2000	-.132	.263	-.185	.345	-.170	.394	-.164	.428	-.144	.451	-.194	.450	.2000	
.2500	-.135		-.207	.371	-.200	.374	-.199	.422	-.187	.458	-.217	.444	.2500	
.3000	-.117	.330	-.220	.334	-.222	.390	-.230	.423	-.210	.447	-.234	.454	.3000	
.3500	-.159		-.242	.342	-.242	.379	-.245	.412	-.229	.449	-.243	.459	.3500	
.4000	-.185		-.248	.355	-.251	.388	-.252	.396	-.235	.430	-.251	.450	.4000	
.4500	-.188		-.259	.355	-.266	.380	-.268		.251	.406	-.265	.437	.4500	
.5000	-.192	.310	-.252	.339	-.269	.362	-.273	.373	-.255	.405	-.272	.414	.5000	
.6000	-.210	.311	-.251	.318	-.284	.318	-.292	.358	-.273	.353	-.283	.383	.6000	
.7000	-.213	.294	-.254	.300	-.290	.303	-.298	.322	-.280	.347	-.280	.370	.7000	
.8000	-.203	.255	-.244	.278		.285	-.298	.304	-.280	.321	-.279	.347	.8000	
.9000	-.204	.261	-.233	.259	-.243	.258	-.229	.276	-.279	.306			.9000	
$\alpha = 16$														
.0125	.158	.059	.167	.225	.143	.354	.048	.463					.0125	
.0250	.088	.139	.130	.252	.069	.376	.002	.471	-.113	.562			.0250	
.0500	.024	.184	.044	.346	.017	.437	-.037	.481	-.119	.536	-.207	.560	.0500	
.1000	-.062	.281	-.075	.380	-.061	.450	-.099	.470	-.130	.522	-.210	.581	.1000	
.1500	-.114	.301	-.169	.391	-.145	.450	-.163	.494	-.161	.522	-.216	.520	.1500	
.2000	-.154	.303	-.207	.402	-.193	.444	-.198	.482	-.191	.504	-.239	.490	.2000	
.2500	-.155		-.233	.418	-.219	.422	-.220	.478	-.220	.505	-.255	.495	.2500	
.3000	-.140	.371	-.253	.384	-.243	.440	-.244	.472	-.237	.493	-.269	.496	.3000	
.3500	-.165		-.259	.394	-.261	.424	-.241	.457	-.252	.496	-.276	.503	.3500	
.4000	-.208		-.244	.403	-.266	.417	-.269	.447	-.262	.472	-.280	.492	.4000	
.4500	-.197		-.276	.404	-.283	.427	-.281		.269	.452	-.290	.482	.4500	
.5000	-.202	.386	-.274	.384	-.284	.406	-.288	.418	-.273	.452	-.293	.461	.5000	
.6000	-.220	.359	-.269	.364	-.295	.366	-.303	.407	-.286	.399	-.294	.428	.6000	
.7000	-.224	.341	-.267	.346	-.304	.349	-.307	.369	-.291	.393	-.291	.413	.7000	
.8000	-.211	.298	-.258	.324		.332	-.301	.349	-.290	.365	-.290	.391	.8000	
.9000	-.213	.304	-.249	.301	-.244	.304	-.232	.319	-.253	.346			.9000	
$\alpha = 18$														
.0125	.109	.129	.108	.339	.059	.472	-.038	.556					.0125	
.0250	.042	.189	.082	.382	-.002	.483	-.075	.555	-.186	.623			.0250	
.0500	-.015	.244	.002	.404	-.040	.500	-.106	.555	-.187	.590	-.250	.592	.0500	
.1000	-.090	.338	-.108	.440	-.104	.510	-.148	.539	-.192	.572	-.250	.607	.1000	
.1500	-.140	.355	-.191	.453	-.174	.511	-.194	.556	-.210	.576	-.254	.566	.1500	
.2000	-.173	.398	-.226	.472	-.211	.503	-.225	.551	-.229	.555	-.266	.538	.2000	
.2500	-.175		-.248	.468	-.236	.485	-.246	.534	-.250	.557	-.278	.545	.2500	
.3000	-.164	.420	-.264	.437	-.253	.502	-.264	.521	-.262	.548	-.286	.547	.3000	
.3500	-.177		-.273	.452	-.271	.482	-.273	.510	-.272	.544	-.286	.551	.3500	
.4000	-.221		-.278	.458	-.279	.481	-.281	.499	-.274	.526	-.287	.547	.4000	
.4500	-.212		-.290	.461	-.292	.480	-.290		.285	.511	-.293	.529	.4500	
.5000	-.217	.409	-.290	.436	-.293	.455	-.294	.473	-.287	.505	-.294	.512	.5000	
.6000	-.233	.416	-.284	.415	-.304	.426	-.309	.458	-.291	.452	-.293	.481	.6000	
.7000	-.234	.396	-.280	.397	-.311	.397	-.313	.421	-.289	.447	-.291	.463	.7000	
.8000	-.222	.351	-.274	.374		.382	-.286	.403	-.281	.413	-.291	.448	.8000	
.9000	-.225	.358	-.266	.354	-.286	.352	-.246	.368	-.259	.398			.9000	

TABLE V.- Concluded

PRESSURE COEFFICIENTS FOR CAMBERED WING

[From reference 1]

(b) M = 2.01 - Concluded

X/C	FRACTION OF SEMISPAN												X/C	
	0.050		0.200		0.350		0.500		0.700		0.900			
	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = 20$														
.0125	.077	.189	.054	.432	-.008	.562	-.116	.617	-.229	.661	-.305	.605	.0125	
.0250	.007	.239	.034	.456	-.057	.557	-.135	.613	-.229	.627	-.303	.618	.0250	
.0500	-.045	.303	-.031	.474	-.090	.562	-.156	.608	-.229	.606	-.303	.618	.0500	
.1000	-.107	.377	-.129	.496	-.143	.559	-.189	.586	-.230	.623	-.305	.577	.1000	
.1500	-.158	.410	-.207	.508	-.198	.557	-.224	.605	-.242	.623	-.305	.577	.1500	
.2000	-.188	.414	-.241	.523	-.231	.553	-.251	.598	-.255	.594	-.310	.554	.2000	
.2500	-.187		-.241	.514	-.253	.538	-.269	.575	-.272	.601	-.312	.560	.2500	
.3000	-.177	.462	-.275	.486	-.268	.553	-.281	.563	-.279	.591	-.317	.560	.3000	
.3500	-.192		-.281	.501	-.283	.532	-.287	.556	-.288	.592	-.316	.566	.3500	
.4000	-.227		-.286	.507	-.288	.524	-.294	.547	-.286	.573	-.316	.561	.4000	
.4500	-.222		-.295	.509	-.298	.526	-.304	.547	-.287	.554	-.317	.545	.4500	
.5000	-.226	.458	-.298	.485	-.300	.504	-.308	.519	-.286	.549	-.317	.529	.5000	
.6000	-.242	.469	-.294	.462	-.311	.477	-.319	.501	-.278	.498	-.317	.499	.6000	
.7000	-.244	.444	-.289	.445	-.317	.443	-.315	.465	-.275	.489	-.317	.482	.7000	
.8000	-.230	.404	-.283	.418		.426	-.294	.444	-.275	.454	-.315	.475	.8000	
.9000	-.230	.409	-.279	.401	-.275	.398	-.267	.407	-.267	.447			.9000	

I-3000

TABLE VI
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING

[From reference 1]

(a) M = 1.61

X/C	FRACTION OF SEMISSPAN													X/C	
	0.050		0.200		0.350		0.500		0.700		0.825		0.950		
	U	L	U	L	U	L	U	L	U	L	U	L	U	L	
$\alpha = -20$															
.0125	+.686	-+.435	.698	-+.484	.638	-+.484	.647	-+.440	.638	-+.424	.612	-+.320	.459	-+.289	.0125
.0250	+.685	-+.434	.688	-+.484	.671	-+.485	.672	-+.440	.638	-+.424	.612	-+.320	.459	-+.289	.0250
.0500	+.619	-+.423	.670	-+.479	.667	-+.485	.690	-+.442	.638	-+.424	.612	-+.320	.555	-+.293	.0500
.0750	+.580	-+.429	.651	-+.470	.666	-+.482	.696	-+.440	.708	-+.422	.685	-+.320	.555	-+.293	.0750
.1000	+.580	-+.423	.629	-+.465	.654	-+.481	.694	-+.441	.724	-+.416	.724	-+.318	.656	-+.298	.1000
.1500	+.549	-+.316	.582	-+.460	.582	-+.476	.681	-+.441	.724	-+.416	.724	-+.318	.656	-+.298	.1500
.2000	+.531	-+.183	.551	-+.469	.613	-+.462	.687	-+.441	.724	-+.416	.724	-+.318	.666	-+.299	.2000
.2500	+.565	-+.214	.553	-+.474	.605	-+.457	.625	-+.441	.675	-+.407	.702	-+.316	.666	-+.299	.2500
.3000	+.499	-+.200	.536	-+.473	.583	-+.452	.611	-+.441	.660	-+.401	.657	-+.318	.672	-+.300	.3000
.3500	+.444	-+.185	.551	-+.469	.613	-+.449	.576	-+.440	.649	-+.397	.633	-+.321	.630	-+.299	.4000
.4000	+.512	-+.256	.500	-+.460	.564	-+.449	.576	-+.440	.621	-+.392	.580	-+.325	.578	-+.299	.5000
.4500	+.453	-+.250	.509	-+.462	.529	-+.448	.516	-+.439	.621	-+.392	.580	-+.325	.578	-+.299	.5500
.5000	+.446	-+.284	.503	-+.458	.501	-+.447	.646	-+.437	.595	-+.379	.552	-+.326	.501	-+.298	.6000
.6000	+.452	-+.307	.416	-+.399	.533	-+.444	.609	-+.432	.555	-+.369	.531	-+.325	.412	-+.296	.7500
.7000	+.452	-+.307	.416	-+.399	.533	-+.444	.609	-+.432	.555	-+.369	.531	-+.325	.412	-+.296	.8000
.8000	+.480	-+.265	.493	-+.338	.634	-+.439	.579	-+.414	.493	-+.364	.485	-+.321	.354	-+.296	.8500
.8500	+.417	-+.287	.552	-+.316	.570	-+.386	.591	-+.3929000
$\alpha = -18$															
.0125	+.628	-+.408	.657	-+.476	.621	-+.486	.603	-+.458	.582	-+.423	.622	-+.355	.485	-+.307	.0125
.0250	+.612	-+.414	.638	-+.476	.635	-+.488	.605	-+.457	.583	-+.418	.664	-+.351	.565	-+.308	.0250
.0500	+.549	-+.396	.588	-+.472	.614	-+.488	.598	-+.459	.583	-+.418	.664	-+.351	.565	-+.308	.0500
.0750	+.513	-+.415	.597	-+.470	.604	-+.488	.601	-+.458	.583	-+.418	.664	-+.351	.565	-+.308	.0750
.1000	+.504	-+.383	.577	-+.465	.595	-+.488	.604	-+.458	.585	-+.412	.699	-+.348	.650	-+.307	.1000
.1500	+.476	-+.195	.529	-+.462	.542	-+.488	.604	-+.460	.585	-+.412	.699	-+.348	.650	-+.307	.1500
.2000	+.465	-+.178	.482	-+.422	.517	-+.481	.558	-+.4602000
.2500	+.502	-+.194	.482	-+.394	.491	-+.472	.540	-+.455	.532	-+.405	.679	-+.343	.647	-+.304	.2500
.3000	+.437	-+.184	.471	-+.376	.458	-+.465	.525	-+.4543000
.3500	+.444	-+.238	.405	-+.387	.445	-+.464	.480	-+.450	.450	-+.393	.640	-+.337	.651	-+.301	.4000
.4000	+.499	-+.381	.381	-+.391	.436	-+.465	.450	-+.450	.606	-+.385	.616	-+.335	.617	-+.299	.4500
.5000	+.381	-+.238	.381	-+.391	.436	-+.465	.450	-+.450	.610	-+.379	.566	-+.332	.563	-+.298	.5500
.6000	+.340	-+.265	.382	-+.378	.424	-+.458	.428	-+.443	.583	-+.373	.536	-+.332	.485	-+.296	.6500
.7000	+.324	-+.290	.334	-+.345	.393	-+.454	.451	-+.426	.546	-+.366	.511	-+.328	.395	-+.295	.7500
.8000	+.369	-+.254	.385	-+.311	.420	-+.446	.537	-+.406	.487	-+.358	.480	-+.322	.334	-+.298	.8000
.9000	+.411	-+.262	.408	-+.297	.460	-+.328	.540	-+.3829000
$\alpha = -16$															
.0125	+.571	+.368	.615	-+.476	.594	-+.512	.582	-+.4930125
.0250	+.549	+.374	.595	-+.466	.594	-+.513	.581	-+.492	.558	-+.446	.553	-+.366	.500	-+.310	.0250
.0500	+.485	+.358	.526	-+.466	.552	-+.509	.558	-+.493	.505	-+.445	.535	-+.362	.566	-+.313	.0500
.0750	+.447	+.383	.523	-+.472	.539	-+.503	.540	-+.493	.505	-+.445	.535	-+.362	.566	-+.313	.0750
.1000	+.522	+.362	.510	-+.475	.528	-+.499	.544	-+.491	.506	-+.441	.548	-+.355	.629	-+.317	.1000
.1500	+.418	+.483	.483	-+.457	.547	-+.493	.527	-+.491	.506	-+.4412000
.2000	+.407	+.169	.423	-+.443	.459	-+.487	.471	-+.4923000
.2500	+.438	+.188	.419	-+.401	.432	-+.479	.429	-+.484	.419	-+.426	.504	-+.365	.628	-+.328	.3500
.3000	+.381	+.172	.410	-+.336	.396	-+.476	.429	-+.484	.371	-+.471	.612	-+.340	.591	-+.333	.4500
.3500	+.400	+.381	.222	-+.299	.361	-+.468	.371	-+.4445000
.4000	+.550	+.323	.323	-+.299	.318	-+.447	.359	-+.444	.394	-+.416	.531	-+.361	.540	-+.339	.5500
.5000	+.284	+.244	.284	-+.296	.307	-+.412	.335	-+.456	.401	-+.407	.526	-+.367	.465	-+.341	.6000
.6000	+.257	+.275	.218	-+.297	.304	-+.406	.328	-+.449	.447	-+.402	.502	-+.364	.375	-+.342	.7000
.7000	+.253	+.243	.270	-+.307	.306	-+.402	.345	-+.432	.488	-+.393	.477	-+.359	.318	-+.341	.8000
.8000	+.298	+.245	.301	-+.306	.336	-+.391	.367	-+.4449000
$\alpha = -14$															
.0125	+.523	+.329	.572	-+.456	.570	-+.503	.590	-+.5080125
.0250	+.488	+.334	.546	-+.457	.569	-+.506	.543	-+.508	.534	-+.482	.519	-+.465	.469	-+.443	.0250
.0500	+.415	+.313	.476	-+.439	.505	-+.496	.519	-+.5080500
.0750	+.388	+.315	.457	-+.432	.487	-+.489	.497	-+.503	.445	-+.488	.477	-+.469	.477	-+.441	.0750
.1000	+.397	+.211	.441	-+.432	.472	-+.482	.492	-+.5001000
.1500	+.365	+.155	.429	-+.420	.473	-+.476	.476	-+.495	.428	-+.485	.479	-+.466	.486	-+.437	.1500
.2000	+.350	+.153	.372	-+.403	.415	-+.466	.419	-+.4912000
.2500	+.378	+.164	.362	-+.369	.386	-+.455	.394	-+.483	.368	-+.486	.436	-+.465	.461	-+.431	.3000
.3000	+.330	+.160	.346	-+.312	.348	-+.457	.376	-+.482	.339	-+.475	.392	-+.466	.474	-+.430	.3500
.4000	+.323	+.205	.297	-+.248	.311	-+.450	.311	-+.470	.323	-+.474	.373	-+.465	.471	-+.428	.4000
.5000	+.273	+.205	.273	-+.255	.275	-+.400	.274	-+.471	.306	-+.471	.346	-+.465	.469	-+.420	.5000
.5500	+.227	+.234	.256	-+.256	.244	-+.382	.241	-+.466	.299	-+.463	.328	-+.454	.447	-+.424	.5500
.6000	+.233	+.227	.234	-+.256	.244	-+.382	.241	-+.466	.289	-+.403	.329	-+.440	.373	-+.420	.7500
.7000	+.206	+.256	.171	-+.256	.222	-+.374	.228	-+.4638000
.7500	+.197	+.230	.189	-+.268	.212	-+.368	.244	-+.4038500
.8000	+.219	+.228	.207	-+.268	.249	-+.352	.262	-+.3509000

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING

From reference [1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN														X/C								
	0.050		0.200		0.350		0.500		0.700		0.825		0.950										
	U	L	U	L	U	L	U	L	U	L	U	L	U	L									
$\alpha = -12^\circ$																							
.0125	.477	-.273	.520	-.413	.532	-.466	.522	-.513	.509	-.498	.488	-.504	.455	-.504	.0125								
.0250	.429	-.260	.498	-.426	.534	-.472	.508	-.510	.509	-.498	.488	-.504	.455	-.504	.0250								
.0500	.362	-.248	.423	-.413	.457	-.465	.494	-.498	.486	-.488	.488	-.504	.455	-.504	.0500								
.0750	.332	-.241	.403	-.397	.434	-.454	.456	-.489	.408	-.506	.436	-.505	.437	-.503	.0750								
.1000	.346	-.194	.384	-.390	.417	-.444	.449	-.480	.448	-.506	.436	-.505	.437	-.503	.1000								
.1500	.319	-.134	.370	-.376	.435	-.455	.427	-.469	.388	-.495	.428	-.500	.424	-.504	.1500								
.2000	.299	-.127	.326	-.346	.360	-.428	.372	-.460	.325	-.487	.380	-.495	.396	-.498	.2500								
.2500	.321	-.143	.311	-.248	.337	-.421	.346	-.452	.325	-.487	.325	-.493	.398	-.498	.3000								
.3000	.285	-.138	.293	-.240	.298	-.423	.330	-.450	.285	-.479	.300	-.489	.368	-.498	.3500								
.3500	.270	-.184	.252	-.229	.262	-.411	.267	-.442	.265	-.474	.238	-.474	.268	-.489	.4000								
.4000	.224	-.181	.225	-.240	.228	-.351	.228	-.446	.219	-.472	.244	-.487	.278	-.493	.4500								
.4500	.188	-.203	.193	-.241	.199	-.328	.188	-.444	.219	-.472	.244	-.487	.278	-.493	.5000								
.5000	.165	-.238	.126	-.239	.179	-.299	.177	-.445	.206	-.464	.246	-.440	.225	-.481	.7500								
.6000	.151	-.210	.141	-.251	.167	-.286	.176	-.391	.212	-.358	.257	-.389	.206	-.425	.8500								
.6500	.179	-.205	.156	-.254	.167	-.278	.175	-.319															.9000
$\alpha = -10^\circ$																							
.0125	.421	-.184	.466	-.362	.493	-.425	.483	-.491	.480	-.498	.461	-.514	.444	-.517	.0125								
.0250	.376	-.187	.442	-.383	.481	-.437	.455	-.486	.480	-.498	.488	-.504	.428	-.508	.0250								
.0500	.305	-.201	.364	-.374	.405	-.429	.453	-.468	.400	-.507	.416	-.515	.0750	-.500	.0500								
.0750	.276	-.210	.339	-.356	.379	-.417	.406	-.457	.377	-.495	.400	-.507	.416	-.515	.1000								
.1000	.283	-.173	.323	-.346	.359	-.406	.392	-.449	.370	-.437	.344	-.476	.379	-.495	.1500								
.1500	.269	-.102	.301	-.323	.337	-.437	.320	-.427	.327	-.463	.332	-.484	.351	-.504	.2000								
.2000	.246	-.100	.274	-.277	.302	-.386	.320	-.427	.272	-.420	.215	-.451	.253	-.476	.3500								
.2500	.263	-.117	.258	-.210	.282	-.376	.292	-.420	.272	-.463	.232	-.484	.311	-.504	.3000								
.3000	.234	-.126	.239	-.186	.246	-.377	.277	-.422															.3500
.4000	.217	-.157	.202	-.209	.210	-.327	.214	-.410	.215	-.451	.215	-.474	.315	-.499	.4000								
.4500	.179	-.161	.173	-.218	.178	-.289	.179	-.418	.189	-.452	.215	-.474	.276	-.493	.5000								
.5000	.140	-.184	.145	-.222	.149	-.285	.146	-.412	.170	-.452	.192	-.475	.213	-.488	.6000								
.6000	.118	-.221	.081	-.221	.129	-.271	.125	-.369	.157	-.452	.183	-.469	.196	-.485	.7000								
.7000	.103	-.192	.097	-.234	.120	-.264	.129	-.346	.152	-.343	.179	-.401	.130	-.471	.8000								
.8000	.089	-.168	.107	-.242	.117	-.256	.126	-.296															.9000
$\alpha = -8^\circ$																							
.0125	.361	-.096	.416	-.301	.448	-.374	.445	-.439	.430	-.507	.432	-.518	.0125	-.500	.0125								
.0250	.324	-.149	.374	-.315	.424	-.389	.407	-.440	.456	-.445	.430	-.507	.432	-.518	.0250								
.0500	.255	-.150	.315	-.346	.340	-.383	.403	-.423	.362	-.456	.366	-.480	.387	-.508	.0500								
.0750	.202	-.202	.297	-.319	.369	-.369	.355	-.412	.362	-.456	.366	-.480	.387	-.508	.0750								
.1000	.226	-.117	.259	-.282	.310	-.358	.340	-.403	.320	-.434	.335	-.457	.342	-.493	.1000								
.1500	.226	-.075	.247	-.267	.347	-.349	.318	-.390	.320	-.434	.335	-.457	.342	-.493	.1500								
.2000	.198	-.074	.223	-.139	.253	-.336	.270	-.380	.232	-.421	.287	-.443	.306	-.482	.2000								
.2500	.211	-.080	.211	-.129	.231	-.316	.247	-.376	.232	-.421	.287	-.443	.306	-.482	.2500								
.3000	.188	-.117	.190	-.136	.198	-.293	.228	-.377	.200	-.416	.236	-.441	.300	-.474	.3500								
.4000	.170	-.133	.155	-.192	.164	-.232	.168	-.366	.168	-.414	.206	-.437	.268	-.469	.4000								
.4500	.135	-.141	.128	-.190	.133	-.242	.134	-.375	.143	-.416	.173	-.439	.231	-.463	.5000								
.5000	.100	-.163	.107	-.199	.109	-.259	.104	-.321	.123	-.417	.145	-.443	.166	-.461	.6000								
.6000	.079	-.204	.042	-.198	.081	-.242	.081	-.296	.113	-.421	.138	-.446	.112	-.461	.7000								
.7000	.080	-.173	.059	-.214	.073	-.233	.083	-.285	.104	-.413	.131	-.443	.082	-.462	.8000								
.8000	.089	-.168	.065	-.220	.075	-.228	.081	-.268															.9000
$\alpha = -6^\circ$																							
.0125	.296	-.034	.358	-.228	.398	-.312	.399	-.382	.357	-.389	.420	-.392	.398	-.481	.421	-.511	.0125						
.0250	.265	-.115	.319	-.257	.370	-.332	.357	-.374	.344	-.374	.327	-.415	.327	-.444	.367	-.486	.0250						
.0500	.207	-.116	.262	-.251	.293	-.325	.344	-.362	.327	-.415	.327	-.444	.367	-.486	.0750								
.0750	.175	-.130	.228	-.231	.252	-.313	.295	-.362	.279	-.412	.327	-.444	.367	-.486	.1000								
.1000	.185	-.083	.208	-.201	.244	-.302	.279	-.342	.262	-.404	.276	-.396	.286	-.420	.310	-.462	.1500						
.1500	.177	-.046	.192	-.142	.204	-.268	.226	-.320	.220	-.329	.226	-.323	.242	-.407	.251	-.450	.2000						
.2000	.153	-.050	.166	-.111	.180	-.225	.185	-.323	.194	-.379	.224	-.407	.251	-.450	.2500								
.2500	.163	-.051	.170	-.107	.180	-.225	.185	-.323	.194	-.379	.224	-.407	.251	-.450	.3000								
.3000	.143	-.078	.142	-.108	.151	-.199	.179	-.325	.158	-.375	.191	-.403	.242	-.443	.3500								
.4000	.128	-.107	.108	-.166	.113	-.180	.122	-.314	.120	-.373	.159	-.405	.219	-.440	.4000								
.4500	.094	-.140	.084	-.164	.089	-.225	.090	-.260	.097	-.377	.121	-.411	.186	-.434	.5000								
.5000	.061	-.139	.067	-.176	.063	-.227	.058	-.235	.077	-.379	.096	-.415	.120	-.432	.6000								
.6000	.041	-.181	.002	-.178	.041	-.214	.043	-.248	.067	-.382	.090	-.419	.063	-.434	.7000								
.7000	.023	-.150	.023	-.193	.035	-.209	.037	-.250	.064	-.378	.081	-.424	.036	-.429	.8000								
.8000	.051	-.146	.024	-.201	.037	-.206	.039	-.239															.9000

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING

[From reference 1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN														X/C		
	0.050		0.200		0.350		0.500		0.700		0.825		0.950				
	U	L	U	L	U	L	U	L	U	L	U	L	U	L			
$\alpha = -04$																	
.0125	.226	.017	.289	-.122	.333	-.211	.359	-.300	.368	-.315	.340	-.419	.396	-.463	.0125		
.0250	.196	-.079	.248	-.183	.302	-.240	.288	-.313	.368	-.315	.340	-.419	.396	-.463	.0250		
.0500	.153	-.057	.201	-.177	.238	-.251	.273	-.299	.321	-.356	.291	-.383	.329	-.432	.0500		
.0750	.121	-.054	.152	-.145	.186	-.244	.234	-.286	.281	-.356	.291	-.383	.329	-.432	.0750		
.1000	.100	-.052	.141	-.120	.172	-.232	.210	-.278	.221	-.356	.223	-.361	.269	-.411	.1000		
.1500	.129	-.022	.133	-.094	.120	-.202	.202	-.272	.221	-.356	.190	-.349	.193	-.399	.1500		
.2000	.107	-.023	.108	-.076	.155	-.152	.163	-.266	.114	-.258	.137	-.325	.190	-.349	.2000		
.2500	.116	-.022	.121	-.075	.127	-.143	.138	-.258	.125	-.235	.125	-.325	.190	-.349	.2500		
.3000	.097	-.074	.093	-.081	.099	-.143	.125	-.235	.111	-.321	.142	-.354	.180	-.397	.3000		
.3500	.080	-.080	.061	-.136	.061	-.146	.068	-.178	.057	-.319	.108	-.353	.153	-.394	.3500		
.4000	.050	-.106	.042	-.137	.041	-.203	.038	-.194	.044	-.323	.067	-.365	.127	-.391	.4000		
.5000	.050	-.116	.025	-.149	.017	-.197	.009	-.203	.029	-.317	.050	-.368	.071	-.389	.5000		
.6000	.021	-.159	.036	-.154	.003	-.185	.005	-.223	.015	-.279	.037	-.376	.014	-.391	.6000		
.7000	-.001	-.159	.036	-.154	.003	-.185	.005	-.223	.015	-.279	.037	-.376	.014	-.391	.7000		
.7500	.018	-.132	-.017	-.171	-.008	-.182	-.007	-.212	.005	-.257	.027	-.384	-.015	-.386	.8500		
.9000	.010	-.124	-.019	-.173	-.006	-.179	-.008	-.208							.9000		
$\alpha = -02$																	
.0125	.145	.071	.214	-.033	.267	-.105	.276	-.171	.305	-.223	.285	-.342	.364	-.395	.0125		
.0250	.129	-.030	.183	-.099	.228	-.170	.225	-.199	.305	-.223	.285	-.342	.364	-.395	.0250		
.0500	.112	-.007	.145	-.065	.183	-.141	.199	-.207	.206	-.227	.280	-.245	-.305	.284	-.344	.0500	
.0750	.083	-.004	.088	-.046	.133	-.124	.164	-.206	.227	-.280	.245	-.305	.284	-.344	.0750		
.1000	.096	-.022	.096	-.041	.110	-.105	.135	-.193	.206	-.280	.245	-.305	.284	-.344	.1000		
.1500	.087	.013	.086	-.023	.103	-.070	.148	-.167	.160	-.260	.167	-.286	.227	-.346	.1500		
.2000	.073	.013	.064	-.036	.103	-.065	.118	-.139	.138	-.223	.148	-.279	.147	-.337	.2000		
.2500	.079	.010	.074	-.035	.090	-.074	.090	-.091	.092	-.241	.148	-.279	.147	-.337	.2500		
.3000	.062	-.030	.056	-.049	.053	-.092	.078	-.135	.075	-.220	.107	-.286	.143	-.339	.3000		
.3500	.045	-.044	.021	-.093	.018	-.119	.029	-.127	.022	-.194	.072	-.286	.109	-.339	.4000		
.4000	.015	-.073	.003	-.099	.001	-.158	-.011	-.161							.5000		
.5000	-.009	-.083	-.009	-.117	-.023	-.156	-.028	-.171	-.008	-.195	.017	-.307	.075	-.340	.5500		
.6000	-.027	-.130	-.070	-.121	-.035	-.148	-.044	-.179	-.017	-.195	.004	-.303	.019	-.333	.6500		
.7000	-.045	-.103	-.048	-.141	-.045	-.150	-.042	-.174	-.022	-.215	-.008	-.282	-.034	-.336	.7500		
.8000	-.018	-.093	-.054	-.142	-.042	-.148	-.046	-.171	-.032	-.226	-.015	-.272	-.059	-.329	.8500		
$\alpha = 00$																	
.0125	.086	.121	.125	.073	.175	-.008	.193	-.031	.227	-.086	.217	-.239	.312	-.309	.0125		
.0250	.047	.026	.105	.018	.139	-.051	.147	-.095	.227	-.086	.217	-.239	.312	-.309	.0250		
.0500	.065	.042	.090	.030	.111	-.009	.132	-.077	.206	-.227	.223	-.283	.284	-.344	.0500		
.0750	.045	.040	.038	.016	.073	-.011	.096	-.051	.153	-.167	.179	-.203	.223	-.283	.0750		
.1000	.054	.004	.040	.021	.044	-.021	.057	-.049	.202	-.124	.100	-.193	.172	-.249	.1000		
.1500	.050	.050	.038	.021	.021	-.026	.076	-.087	.093	-.124	.100	-.193	.172	-.249	.1500		
.2000	.042	.045	.021	-.004	.042	-.023	.059	-.061	.061	-.124	.076	-.178	.085	-.260	.2000		
.2500	.044	.024	.026	-.002	.042	-.034	.038	-.053	.055	-.124	.076	-.178	.085	-.260	.2500		
.3000	.014	.010	.014	-.015	.012	-.037	.026	-.085	.026	-.112	.061	-.177	.088	-.262	.3000		
.3500	.010	-.008	-.017	-.055	-.025	-.082	-.019	-.093	-.018	-.142	.030	-.180	.062	-.264	.4000		
.4000	-.017	-.041	-.031	-.067	-.046	-.119	-.063	-.124	-.054	-.158	-.028	-.188	.025	-.267	.5000		
.5000	-.043	-.056	-.043	-.088	-.062	-.126	-.074	-.124	-.068	-.167	-.045	-.199	-.027	-.252	.6500		
.6000	-.061	-.105	-.099	-.093	-.077	-.116	-.085	-.143	-.077	-.192	-.062	-.224	-.075	-.244	.7500		
.7000	-.071	-.088	-.079	-.116	-.081	-.120	-.085	-.143	-.075	-.188	-.067	-.243	-.101	-.239	.8500		
.8000	-.048	-.067	-.081	-.118	-.081	-.120	-.084	-.139							.9000		
$\alpha = 02$																	
.0125	.037	.170	.028	.161	.081	.110	.078	.093	.129	.040	.128	-.094	.246	-.179	.0125		
.0250	-.012	.090	.020	.103	.045	.069	.050	.038	.129	.040	.128	-.094	.246	-.179	.0250		
.0500	.018	.081	.030	.085	.040	.070	.064	.030	.129	.040	.128	-.094	.246	-.179	.0500		
.0750	.013	.048	-.001	.075	.016	.045	.048	.024	.072	-.027	.091	-.067	.155	-.142	.0750		
.1000	.020	.036	-.019	.080	-.015	.036	.026	.002	.001	-.007	.006	-.043	.029	-.055	.1000		
.1500	.018	.077	-.006	.044	-.015	.024	.010	-.009	.005	-.005	.005	-.028	.018	-.020	.1500		
.2000	.012	.072	-.017	.044	-.009	.022	.002	-.009	.004	-.007	.004	-.020	.007	-.020	.2000		
.2500	.032	.050	-.004	.036	-.003	.010	-.009	-.005	.005	-.005	.005	-.020	.006	-.022	.2500		
.3000	-.028	.054	-.016	.022	-.019	.017	-.019	-.019	.019	-.019	.019	-.021	.016	-.024	.3000		
.3500	-.023	.021	-.056	-.022	-.003	-.040	-.051	-.047	-.052	-.049	-.052	-.052	-.020	-.026	.4000		
.4000	-.048	-.017	-.067	-.035	-.083	-.085	-.106	-.089	-.089	-.090	-.090	-.090	-.010	-.018	.5000		
.5000	-.069	-.028	-.075	-.056	-.096	-.094	-.116	-.087	-.087	-.090	-.090	-.090	-.019	-.028	.6000		
.6000	-.069	-.087	-.082	-.124	-.045	-.107	-.086	-.125	-.108	-.108	-.108	-.108	-.081	-.163	-.193	.7000	
.7000	-.087	-.082	-.107	-.088	-.109	-.094	-.119	-.105	-.105	-.105	-.105	-.105	-.100	-.109	-.200	.7500	
.8000	-.087	-.079	-.107	-.088	-.109	-.094	-.119	-.105	-.105	-.105	-.105	-.105	-.100	-.109	-.200	.8000	
.8500	-.063	-.048	-.106	-.096	-.107	-.092	-.116	-.106	-.106	-.106	-.106	-.106	-.111	-.120	-.130	-.199	.8500
.9000	-.063	-.048	-.106	-.096	-.107	-.092	-.116	-.106	-.106	-.106	-.106	-.106	-.111	-.120	-.130	-.199	.9000

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING
[From reference 1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN														X/C	
	0.050		0.200		0.350		0.500		0.700		0.825		0.950			
	U	L	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = 0^\circ$																
.0125	-.021	.243	-.087	.255	-.042	.220	-.092	.209	-.013	.180	.009	.097	.135	.056	.0125	
.0250	-.065	.155	-.098	.183	-.102	.163	-.103	.155	-.013	.104	.006	.072	.037	.010	.0250	
.0500	-.041	.133	-.062	.153	-.081	.147	-.063	.109	-.012	.104	.006	.072	.037	.010	.0500	
.0750	-.023	.083	-.061	.145	-.056	.111	-.037	.111	-.012	.104	.006	.072	.037	.010	.0750	
.1000	-.016	.076	-.072	.136	-.062	.108	-.041	.085	-.012	.104	.006	.072	.037	.010	.1000	
.1500	-.013	.111	-.054	.115	-.056	.090	-.045	.080	-.012	.104	.006	.072	.037	.010	.1500	
.2000	-.020	.113	-.062	.092	-.072	.086	-.095	.076	-.012	.104	.006	.072	.037	.010	.2000	
.2500	-.013	.087	-.044	.089	-.057	.063	-.078	.039	-.012	.104	.006	.072	.037	.010	.2500	
.3000	-.067	.092	-.045	.063	-.055	.028	-.084	.021	-.012	.104	.006	.072	.037	.010	.3000	
.3500	-.057	.059	-.098	.019	-.092	.005	-.088	.004	-.012	.104	.006	.072	.037	.010	.3500	
.4000	-.057	.059	-.098	.019	-.092	.005	-.088	.004	-.012	.104	.006	.072	.037	.010	.4000	
.4500	-.080	.020	-.102	.004	-.132	.043	-.134	.039	-.110	-.039	-.096	-.042	-.094	-.108	.4500	
.5000	-.097	.012	-.108	.019	-.136	.054	-.161	.043	-.136	-.072	-.129	-.099	-.112	-.127	.5000	
.6000	-.097	.111	-.053	.152	-.032	.142	-.047	.175	-.067	-.132	-.082	-.135	-.114	-.136	.6000	
.7000	-.111	.053	-.152	.032	-.142	.047	-.175	.067	-.150	-.108	-.148	-.145	-.152	-.159	.7000	
.8000	-.106	.050	-.134	.058	-.141	.058	-.159	.069	-.162	-.108	-.146	-.163	-.154	-.165	.8000	
.8500	-.086	.019	-.128	.064	-.132	.057	-.153	.061	-.112	-.004	-.097	-.038	-.089	-.104	.8500	
.9000	-.086	.019	-.128	.064	-.132	.057	-.153	.061	-.112	-.004	-.097	-.038	-.089	-.104	.9000	
$\alpha = 0.6^\circ$																
.0125	-.077	.302	-.170	.331	-.139	.303	-.211	.290	-.159	.268	-.129	.221	-.009	.173	.0125	
.0250	-.114	.241	-.188	.256	-.188	.240	-.217	.241	-.159	.215	-.129	.221	-.009	.173	.0250	
.0500	-.097	.177	-.173	.208	-.199	.208	-.186	.201	-.158	.176	-.135	.155	-.008	.109	.0500	
.0750	-.049	.151	-.158	.197	-.184	.178	-.167	.177	-.158	.176	-.135	.155	-.008	.109	.0750	
.1000	-.052	.112	-.128	.191	-.178	.165	-.181	.146	-.167	.121	-.133	.100	-.108	.047	.1000	
.1500	-.040	.155	-.106	.166	-.158	.174	-.137	.167	-.167	.121	-.133	.100	-.108	.047	.1500	
.2000	-.052	.153	-.098	.139	-.154	.129	-.178	.128	-.177	.102	-.158	.066	-.126	-.000	.2000	
.2500	-.009	.130	-.086	.129	-.121	.106	-.183	.088	-.162	.070	-.158	.066	-.126	-.000	.2500	
.3000	-.099	.133	-.074	.102	-.112	.074	-.177	.071	-.180	.045	-.149	.032	-.147	-.023	.3000	
.3500	-.086	.098	-.139	.057	-.120	.047	-.149	.044	-.190	.002	-.176	.011	-.161	-.050	.4000	
.4000	-.108	.064	-.136	.037	-.166	.004	-.161	.005	-.217	-.020	-.197	.055	-.176	-.076	.5000	
.4500	-.120	.049	-.140	.015	-.179	.014	-.185	.006	-.217	-.037	-.210	.067	-.190	-.116	.6000	
.7000	-.136	.022	-.178	.000	-.180	.009	-.206	.025	-.207	-.069	-.218	.091	-.198	-.143	.7000	
.8000	-.123	.020	-.162	.031	-.177	.023	-.210	.030	-.188	-.072	-.225	.127	-.202	-.157	.8000	
.8500	-.113	.016	-.154	.036	-.169	.022	-.192	.034	-.112	-.004	-.097	-.038	-.154	-.165	.9000	
$\alpha = 0.8^\circ$																
.0125	-.142	.351	-.236	.385	-.224	.373	-.290	.348	-.248	.339	-.221	.295	-.152	.248	.0125	
.0250	-.163	.268	-.224	.324	-.265	.292	-.303	.311	-.248	.339	-.221	.295	-.152	.248	.0250	
.0500	-.125	.202	-.248	.274	-.248	.247	-.275	.255	-.248	.237	-.215	.227	-.174	.178	.0500	
.0750	-.125	.208	-.237	.242	-.254	.236	-.254	.223	-.240	.237	-.215	.227	-.174	.178	.0750	
.1000	-.098	.149	-.228	.226	-.248	.226	-.244	.203	-.238	.237	-.216	.217	-.190	.101	.1000	
.1500	-.071	.200	-.193	.208	-.208	.206	-.238	.188	-.237	.217	-.216	.217	-.190	.101	.1500	
.2000	-.082	.195	-.127	.189	-.248	.169	-.239	.177	-.238	.212	-.226	.215	-.202	.061	.2000	
.2500	-.035	.167	-.122	.164	-.234	.150	-.245	.131	-.238	.212	-.226	.215	-.202	.061	.2500	
.3000	-.123	.170	-.100	.143	-.165	.118	-.250	.121	-.242	.091	-.231	.087	-.215	.035	.3000	
.4000	-.115	.133	-.162	.094	-.155	.087	-.258	.086	-.252	.046	-.236	.052	-.225	-.009	.4000	
.4500	-.134	.090	-.167	.073	-.185	.028	-.264	.044	-.269	.023	-.259	.008	-.235	-.026	.5000	
.5000	-.146	.084	-.166	.052	-.215	.022	-.211	.032	-.276	.000	-.285	.029	-.235	-.079	.6000	
.6000	-.181	.009	-.199	.032	-.211	.022	-.224	.012	-.280	-.032	-.274	.059	-.243	-.114	.7000	
.7000	-.145	.010	-.186	.001	-.206	.007	-.234	.012	-.287	-.025	-.278	.086	-.240	-.135	.8000	
.8000	-.137	.050	-.177	.000	-.195	.012	-.235	.005	-.303	.023	-.289	.086	-.240	-.135	.9000	
$\alpha = 10^\circ$																
.0125	-.202	.410	-.303	.449	-.307	.433	-.305	.435	-.404	.404	-.323	.404	-.313	.361	.287	.322
.0250	-.223	.333	-.313	.393	-.326	.362	-.345	.375	-.323	.323	-.306	.305	-.295	.293	.281	.249
.0500	-.195	.283	-.305	.357	-.328	.364	-.344	.355	-.325	.325	-.306	.305	-.295	.293	.281	.249
.0750	-.179	.242	-.300	.302	-.318	.327	-.327	.327	-.321	.321	-.306	.305	-.295	.293	.281	.249
.1000	-.146	.216	-.277	.288	-.311	.287	-.318	.270	-.307	.307	-.290	.298	-.284	.284	.272	.193
.1500	-.092	.252	-.274	.247	-.323	.253	-.307	.250	-.329	.329	-.284	.284	-.272	.272	.213	.193
.2000	-.110	.241	-.261	.248	-.307	.222	-.300	.227	-.303	.303	-.287	.287	-.272	.272	.124	.2000
.2500	-.059	.214	-.184	.217	-.301	.198	-.301	.186	-.302	.317	-.289	.289	-.273	.273	.124	.2500
.3000	-.138	.216	-.136	.197	-.299	.171	-.303	.174	-.293	.314	-.291	.291	-.273	.273	.3000	.3000
.4000	-.143	.180	-.181	.142	-.243	.137	-.312	.140	-.293	.314	-.291	.291	-.273	.273	.4000	.4000
.4500	-.139	.149	-.195	.120	-.217	.065	-.333	.086	-.302	.095	-.291	.102	-.287	.044	.4500	.4500
.5000	-.159	.139	-.195	.120	-.217	.065	-.333	.086	-.317	.074	-.306	.052	-.288	.027	.5000	.5000
.6000	-.167	.124	-.192	.096	-.236	.068	-.339	.084	-.323	.056	-.313	.015	-.278	-.024	.6000	.6000
.7000	-.184	.045	-.227	.074	-.246	.062	-.305	.058	-.327	.018	-.318	.010	-.288	-.080	.7000	.7000
.8000	-.164	.050	-.209	.037	-.232	.049	-.249	.045	-.327	.018	-.318	.010	-.288	-.080	.8000	.8000
.8500	-.160	.090	-.201	.045	-.220	.053	-.263	.052	-.323	.016	-.323	.039	-.289	-.100	.8500	.8500

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING

[From reference 1]

(a) $M = 1.61$ - Continued

X/C	FRACTION OF SEMISPAN														X/C	
	0.050		0.200		0.350		0.500		0.700		0.825		0.950			
	U	L	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = 12^\circ$																
.0125	-0.259	.464	-0.361	.515	-0.385	.497	-0.417	.442	-0.400	.451	-0.389	.408	-0.372	.376	.0125	
.0250	-0.283	.395	-0.365	.449	-0.390	.426	-0.416	.429	-0.400	.451	-0.372	.397	-0.355	.397	.0250	
.0500	-0.250	.335	-0.356	.391	-0.381	.402	-0.398	.382	-0.372	.353	-0.362	.348	-0.355	.297	.0500	
.0750	-0.239	.314	-0.349	.359	-0.369	.340	-0.381	.339	-0.372	.353	-0.362	.348	-0.355	.297	.0750	
.1000	-0.169	.278	-0.345	.337	-0.361	.336	-0.374	.329	-0.362	.300	-0.354	.293	-0.346	.264	.1000	
.1500	-0.135	.293	-0.331	.317	-0.311	.275	-0.355	.284	-0.351	.241	-0.352	.219	-0.344	.229	.1500	
.2000	-0.131	.289	-0.306	.294	-0.349	.275	-0.355	.284	-0.351	.241	-0.352	.219	-0.344	.172	.2000	
.2500	-0.100	.267	-0.253	.284	-0.347	.245	-0.351	.241	-0.352	.219	-0.344	.229	-0.335	.172	.2500	
.3000	-0.153	.261	-0.227	.239	-0.345	.216	-0.351	.222	-0.346	.177	-0.344	.186	-0.330	.139	.3000	
.3500	-0.168	.222	-0.208	.188	-0.347	.186	-0.355	.188	-0.351	.142	-0.343	.149	-0.334	.103	.4000	
.4000	-0.180	.169	-0.226	.160	-0.281	.109	-0.375	.129	-0.381	.119	-0.359	.100	-0.337	.078	.5500	
.5000	-0.180	.169	-0.226	.160	-0.281	.109	-0.375	.129	-0.381	.119	-0.359	.100	-0.337	.078	.5000	
.5500	-0.186	.163	-0.217	.137	-0.270	.109	-0.381	.129	-0.385	.094	-0.359	.059	-0.323	.014	.6500	
.6000	-0.205	.085	-0.253	.110	-0.273	.102	-0.386	.097	-0.366	.054	-0.361	.033	-0.331	-.036	.7500	
.7500	-0.284	.089	-0.231	.077	-0.261	.086	-0.366	.087	-0.370	.050	-0.366	.001	-0.333	-.061	.8500	
.8500	-0.180	.130	-0.225	.087	-0.246	.102	-0.268	.088	-0.260	.050	-0.366	.001	-0.333	-.061	.9000	
$\alpha = 14^\circ$																
.0125	-0.309	.525	-0.418	.556	-0.445	.548	-0.446	.473	-0.450	.494	-0.438	.460	-0.434	.417	.0125	
.0250	-0.335	.454	-0.417	.507	-0.445	.470	-0.446	.487	-0.445	.494	-0.438	.460	-0.434	.417	.0250	
.0500	-0.311	.392	-0.401	.446	-0.433	.448	-0.442	.453	-0.427	.413	-0.415	.398	-0.417	.353	.0500	
.0750	-0.293	.370	-0.392	.414	-0.418	.392	-0.429	.396	-0.427	.413	-0.415	.398	-0.417	.353	.0750	
.1000	-0.200	.316	-0.388	.389	-0.411	.388	-0.421	.382	-0.412	.357	-0.407	.355	-0.396	.283	.1000	
.1500	-0.155	.342	-0.374	.382	-0.361	.412	-0.357	.407	-0.396	.334	-0.399	.283	-0.350	.1800		
.2000	-0.155	.339	-0.357	.339	-0.400	.324	-0.404	.332	-0.397	.272	-0.389	.295	-0.389	.237	.2000	
.2500	-0.126	.326	-0.305	.315	-0.393	.300	-0.396	.285	-0.392	.231	-0.386	.241	-0.381	.198	.3000	
.3000	-0.176	.310	-0.268	.286	-0.392	.269	-0.394	.277	-0.392	.186	-0.385	.202	-0.381	.189	.4000	
.3500	-0.191	.273	-0.264	.232	-0.396	.235	-0.395	.234	-0.393	.186	-0.385	.202	-0.381	.189	.5000	
.4000	-0.202	.219	-0.261	.205	-0.382	.160	-0.410	.178	-0.402	.170	-0.393	.155	-0.381	.145	.6500	
.5000	-0.227	.127	-0.273	.151	-0.327	.144	-0.422	.138	-0.405	.148	-0.396	.128	-0.365	.091	.7500	
.6000	-0.210	.209	-0.250	.175	-0.334	.147	-0.416	.166	-0.406	.116	-0.399	.106	-0.371	.061	.8500	
.7500	-0.226	.127	-0.273	.151	-0.327	.144	-0.422	.138	-0.406	.116	-0.399	.106	-0.371	.061	.9000	
.8500	-0.202	.133	-0.258	.118	-0.313	.134	-0.420	.135	-0.406	.126	-0.400	.085	-0.374	.092	.9500	
$\alpha = 16^\circ$																
.0125	-0.345	.578	-0.448	.589	-0.473	.584	-0.489	.504	-0.473	.523	-0.467	.508	-0.474	.479	.0125	
.0250	-0.360	.501	-0.442	.556	-0.471	.500	-0.488	.528	-0.473	.547	-0.460	.480	-0.459	.462	.0250	
.0500	-0.342	.444	-0.424	.494	-0.458	.491	-0.473	.479	-0.462	.513	-0.445	.460	-0.445	.462	.0500	
.0750	-0.324	.397	-0.420	.464	-0.446	.439	-0.459	.441	-0.452	.460	-0.445	.460	-0.445	.462	.0750	
.1000	-0.255	.352	-0.419	.443	-0.439	.432	-0.451	.425	-0.440	.401	-0.436	.416	-0.427	.398	.1000	
.1500	-0.172	.385	-0.412	.431	-0.408	.440	-0.440	.401	-0.436	.421	-0.427	.398	-0.438	.368	.1500	
.2000	-0.172	.384	-0.400	.379	-0.428	.363	-0.433	.373	-0.421	.321	-0.415	.302	-0.411	.310	.2000	
.2500	-0.151	.379	-0.365	.368	-0.420	.343	-0.426	.333	-0.423	.342	-0.417	.361	-0.425	.378	.2500	
.3000	-0.188	.352	-0.332	.331	-0.419	.310	-0.421	.321	-0.421	.321	-0.415	.302	-0.411	.310	.3000	
.3500	-0.208	.318	-0.284	.278	-0.422	.277	-0.424	.274	-0.421	.271	-0.413	.261	-0.409	.279	.3500	
.4000	-0.224	.265	-0.266	.245	-0.422	.199	-0.434	.219	-0.421	.242	-0.415	.246	-0.408	.283	.4000	
.5000	-0.220	.265	-0.266	.245	-0.422	.199	-0.434	.219	-0.421	.242	-0.415	.246	-0.408	.283	.5000	
.5500	-0.224	.248	-0.256	.215	-0.377	.181	-0.439	.237	-0.423	.223	-0.416	.245	-0.397	.242	.6500	
.6000	-0.241	.163	-0.279	.190	-0.363	.196	-0.442	.216	-0.425	.185	-0.418	.267	-0.400	.253	.7500	
.7500	-0.233	.201	-0.337	.293	-0.390	.300	-0.450	.323	-0.437	.264	-0.419	.355	-0.402	.212	.8500	
.8500	-0.234	.336	-0.306	.314	-0.370	.361	-0.331	.450	-0.453	.383	-0.459	.474	-0.474	.440	.9000	
$\alpha = 18^\circ$																
.0125	-0.384	.637	-0.462	.640	-0.498	.629	-0.510	.536	-0.502	.557	-0.498	.576	-0.500	.609	.0125	
.0250	-0.404	.562	-0.456	.606	-0.497	.534	-0.508	.573	-0.502	.549	-0.490	.556	-0.494	.470	.0250	
.0500	-0.381	.507	-0.454	.556	-0.487	.538	-0.497	.525	-0.488	.549	-0.490	.556	-0.494	.470	.0500	
.0750	-0.393	.468	-0.458	.528	-0.477	.486	-0.488	.497	-0.489	.549	-0.470	.556	-0.457	.440	.0750	
.1000	-0.379	.414	-0.449	.517	-0.472	.496	-0.481	.483	-0.474	.547	-0.474	.527	-0.488	.520	.1000	
.1500	-0.179	.447	-0.444	.486	-0.445	.465	-0.471	.499	-0.474	.547	-0.474	.527	-0.488	.520	.1500	
.2000	-0.187	.443	-0.391	.433	-0.459	.427	-0.464	.451	-0.464	.547	-0.474	.527	-0.488	.520	.2000	
.2500	-0.168	.448	-0.343	.430	-0.451	.400	-0.458	.427	-0.460	.547	-0.474	.533	-0.481	.476	.2500	
.3000	-0.203	.406	-0.336	.384	-0.448	.367	-0.453	.421	-0.453	.538	-0.459	.474	-0.474	.440	.3000	
.4000	-0.225	.477	-0.335	.335	-0.450	.338	-0.453	.380	-0.453	.364	-0.454	.463	-0.470	.368	.4000	
.5000	-0.240	.325	-0.333	.305	-0.418	.290	-0.461	.328	-0.461	.319	-0.456	.386	-0.455	.463	.5000	
.5500	-0.244	.299	-0.345	.279	-0.387	.295	-0.464	.319	-0.466	.327	-0.451	.327	-0.449	.457	.5500	
.6000	-0.260	.220	-0.362	.289	-0.387	.295	-0.467	.294	-0.467	.327	-0.451	.327	-0.449	.457	.6000	
.6500	-0.260	.220	-0.362	.289	-0.387	.295	-0.467	.294	-0.467	.327	-0.451	.327	-0.449	.457	.6500	
.7000	-0.233	.201	-0.337	.293	-0.390	.300	-0.450	.323	-0.451	.329	-0.449	.320	-0.433	.328	.7000	
.7500	-0.233	.175	-0.262	.169	-0.354	.205	-0.442	.209	-0.442	.378	-0.449	.355	-0.402	.212	.7500	
.8000	-0.234	.336	-0.306	.314	-0.370	.361	-0.331	.450	-0.453	.383	-0.459	.474	-0.474	.440	.8000	
.8500	-0.234	.336	-0.306	.314	-0.370	.361	-0.331	.450	-0.453	.383	-0.459	.474	-0.474	.440	.8500	
.9000	-0.234	.336	-0.306	.314												

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING
[From reference 1]

(a) $M = 1.61$ - Concluded

X/C	FRACTION OF SEMISPAN														X/C	
	0.050		0.200		0.350		0.500		0.700		0.825		0.950			
	U	L	U	L	U	L	U	L	U	L	U	L	U	L		
a: 20																
.0125	-+408	+685	-+467	+679	-+503	+651	-+516	+596	-+505	+583	-+498	+638	-+499	+595	.0125	
.0250	-+418	+621	-+467	+647	-+498	+584	-+515	+650	-+505	+617	-+498	+672	-+492	+621	.0250	
.0500	-+401	+561	-+467	+604	-+495	+582	-+508	+617	-+505	+614	-+496	+672	-+492	+621	.0500	
.0750	-+406	+565	-+467	+591	-+489	+543	-+503	+596	-+498	+614	-+496	+672	-+492	+621	.0750	
.1000	-+413	+473	-+456	+577	-+488	+547	-+499	+584	-+497	+564	-+490	+590	-+488	+651	.1000	
.1500	-+296	+504	-+428	+525	-	+522	-+487	+564	-+490	+590	-+488	+651	-+487	+475	.1500	
.2000	-+188	+500	-+369	+486	-+481	+487	-+483	+535	-	+500	-+484	+543	-+488	+645	-+481	.2000
.2500	-+189	+506	-+339	+476	-+475	+480	-+475	+500	-+484	+543	-+488	+645	-+481	+535	.2500	
.3000	-+209	+456	-+328	+431	-+467	+482	-+471	+494	-	+475	+537	-+484	+560	-+479	+490	.3000
.3500	-	+238	+429	-+343	+387	-+467	+452	-+467	+456	-	+475	+537	-+484	+560	-+479	.3500
.4000	-	+254	+382	-+335	+410	-+464	+363	-+475	+395	-	+473	+554	-+479	+527	-+468	.4000
.4500	-	+254	+382	-+335	+410	-+464	+363	-+475	+395	-	+473	+543	-+475	+466	-+456	.4500
.5000	-	+257	+374	-+322	+378	-+433	+368	-+477	+413	-	+473	+543	-+475	+466	-+456	.5000
.6000	-	+272	+319	-+331	+364	-+415	+364	-+479	+462	-	+386	+501	-+415	+422	-+437	.6000
.7000	-	+272	+319	-+331	+364	-+415	+364	-+479	+462	-	+347	+449	-+381	+372	-+415	.7000
.7500	-	+241	+379	-+342	+376	-+412	+408	-+377	+522	-	+355	+427	-+365	+370	-+404	.7500
.8000	-	+251	+423	-+340	+417	-+391	+495	-+373	+520	-	+355	+427	-+365	+370	-+404	.8000
.8500	-	+251	+423	-+340	+417	-+391	+495	-+373	+520	-	+355	+427	-+365	+370	-+404	.8500
.9000	-	+251	+423	-+340	+417	-+391	+495	-+373	+520	-	+355	+427	-+365	+370	-+404	.9000

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING
[From reference 1]

(b) $M = 2.01$

X/C	FRACTION OF SEMISPAN														X/C							
	0.050		0.200		0.350		0.500		0.700		0.825		0.950									
	U	L	U	L	U	L	U	L	U	L	U	L	U	L								
$\alpha = -20$																						
.0125	.671	-.254	.695	-.320	.675	-.322	.661	-.328	.675	-.322	.661	-.328	.675	-.322	.0125							
.0250	.642	-.258	.686	-.314	.697	-.322	.662	-.326	.626	-.277	.619	-.290	.519	-.283	.0250							
.0500	.561	-.240	.618	-.306	.649	-.322	.676	-.325	.524	-.277	.519	-.290	.554	-.287	.0500							
.0750	.531	-.244	.600	-.303	.638	-.320	.626	-.326	.589	-.318	.598	-.290	.554	-.287	.0750							
.1000	.521	-.245	.683	-.304	.634	-.321	.635	-.325	.577	-.309	.606	-.286	.578	-.290	.1000							
.1500	.491	-.245	.556	-.304	.523	-.321	.621	-.324	.519	-.311	.577	-.282	.572	-.291	.2000							
.2000	.467	-.237	.521	-.304	.546	-.326	.560	-.327	.490	-.312	.528	-.282	.584	-.291	.2500							
.2500	.481	-.172	.618	-.303	.524	-.329	.541	-.327	.519	-.311	.577	-.282	.572	-.291	.3000							
.3000	.470	-.121	.511	-.299	.488	-.328	.529	-.326	$\alpha = -18$													
.3500	.451	-.221	.433	-.289	.453	-.327	.464	-.327	.466	-.307	.499	-.283	.545	-.293	.3500							
.4000	.452	-.192	.406	-.294	.417	-.322	.425	-.329	.436	-.308	.465	-.292	.543	-.292	.4000							
.4500	.392	-	.372	-.293	.380	-.317	.383	-.328	.414	-.303	.442	-.294	.505	-.293	.5000							
.5000	.364	-.207	.372	-.293	.380	-.317	.383	-.328	.400	-.303	.433	-.294	.455	-.294	.5500							
.6000	.343	-.222	.293	-.295	.345	-.311	.363	-.312	.414	-.303	.442	-.294	.505	-.293	.6000							
.6500	.343	-.222	.293	-.295	.345	-.311	.363	-.312	.400	-.303	.433	-.294	.455	-.294	.6500							
.7000	.343	-.222	.293	-.295	.345	-.311	.363	-.312	.400	-.303	.433	-.294	.455	-.294	.7000							
.7500	.343	-.222	.293	-.295	.345	-.311	.363	-.312	.400	-.303	.433	-.294	.455	-.294	.7500							
.8000	.296	-.205	.308	-.299	.334	-.304	.360	-.301	.393	-.298	.437	-.294	.437	-.296	.8000							
.8500	.331	-.205	.300	-.300	.326	-.292	.351	-.289	$\alpha = -16$													
.9000	.331	-.205	.300	-.300	.326	-.292	.351	-.289	$\alpha = -14$													
.0125	.621	-.231	.649	-.298	.657	-.314	.613	-.334	.603	-.276	.597	-.301	.519	-.289	.0125							
.0250	.584	-.248	.638	-.303	.664	-.308	.615	-.331	.554	-.320	.565	-.307	.536	-.291	.0250							
.0500	.508	-.240	.569	-.288	.611	-.306	.629	-.330	.554	-.320	.565	-.307	.536	-.291	.0500							
.0750	.471	-.247	.546	-.283	.592	-.314	.586	-.330	.554	-.320	.565	-.307	.536	-.291	.0750							
.1000	.461	-.247	.525	-.280	.582	-.315	.588	-.329	.537	-.311	.568	-.303	.547	-.294	.1000							
.1500	.431	-.235	.497	-.280	.521	-.318	.569	-.329	.537	-.311	.544	-.300	.538	-.294	.2000							
.2000	.412	-.170	.467	-.281	.514	-.322	.516	-.329	.482	-.313	.534	-.301	.543	-.294	.2500							
.3000	.414	-.110	.446	-.279	.445	-.324	.479	-.327	$\alpha = -12$													
.3500	.396	-.215	.383	-.263	.410	-.320	.411	-.327	.446	-.313	.486	-.301	.543	-.294	.3500							
.4000	.341	-.180	.360	-.269	.373	-.317	.372	-.327	.421	-.308	.452	-.297	.502	-.294	.4000							
.4500	.341	-.180	.360	-.269	.373	-.317	.372	-.327	.387	-.310	.417	-.304	.499	-.294	.5000							
.5000	.313	-.193	.338	-.271	.345	-.319	.335	-.327	.367	-.304	.391	-.304	.457	-.294	.6000							
.6000	.294	-.210	.255	-.273	.306	-.308	.310	-.313	.347	-.301	.380	-.301	.406	-.294	.7000							
.7000	.294	-.210	.255	-.273	.306	-.308	.310	-.313	.319	-.306	.341	-.318	.407	-.310	.8000							
.7500	.245	-.193	.271	-.271	.296	-.297	.306	-.292	.342	-.294	.377	-.299	.387	-.294	.8500							
.8000	.282	-.192	.258	-.270	.288	-.286	.301	-.282	$\alpha = -10$													
.8500	.282	-.180	.202	-.219	.237	-.237	.237	-.237	.353	-.237	.373	-.300	.394	-.316	.9000							
.9000	.314	-.084	.376	-.237	.337	-.282	.373	-.300	$\alpha = -8$													
.0125	.564	-.205	.604	-.275	.618	-.310	.597	-.332	.576	-.329	.579	-.279	.576	-.307	.513	-.317	.0125					
.0250	.525	-.226	.576	-.284	.509	-.310	.581	-.332	.524	-.329	.524	-.327	.535	-.315	.510	-.318	.0250					
.0500	.452	-.220	.509	-.282	.551	-.309	.588	-.329	.519	-.321	.535	-.315	.510	-.318	.0500							
.0750	.403	-.224	.486	-.276	.524	-.308	.539	-.329	.507	-.321	.522	-.311	.507	-.319	.0750							
.1000	.403	-.223	.458	-.273	.516	-.305	.527	-.326	.489	-.309	.522	-.311	.507	-.319	.1000							
.1500	.372	-.211	.434	-.269	.502	-.302	.507	-.321	.521	-.309	.522	-.311	.507	-.319	.1500							
.2000	.355	-.113	.400	-.265	.468	-.299	.462	-.318	.432	-.308	.489	-.310	.494	-.316	.2000							
.3000	.356	-.139	.401	-.261	.422	-.295	.433	-.317	.432	-.308	.486	-.308	.498	-.318	.3000							
.3500	.361	-.088	.382	-.261	.386	-.294	.421	-.314	.396	-.313	.438	-.308	.498	-.318	.3500							
.4000	.336	-.213	.314	-.261	.346	-.296	.356	-.311	.368	-.308	.404	-.305	.454	-.318	.4000							
.4500	.290	-.166	.299	-.254	.316	-.298	.316	-.314	.339	-.309	.368	-.318	.448	-.318	.5000							
.5000	.248	-.181	.282	-.240	.288	-.300	.286	-.316	.319	-.306	.341	-.318	.407	-.310	.6000							
.6000	.248	-.202	.202	-.223	.254	-.293	.263	-.307	.301	-.308	.328	-.318	.355	-.316	.7000							
.7000	.243	-.191	.160	-.202	.211	-.280	.219	-.295	.291	-.302	.320	-.315	.332	-.317	.8000							
.7500	.198	-.191	.160	-.202	.211	-.280	.219	-.295	.274	-.300	.295	-.319	.356	-.320	.7500							
.8000	.149	-.176	.176	-.207	.193	-.278	.213	-.295	.260	-.302	.285	-.320	.306	-.322	.8000							
.8500	.185	-.169	.158	-.209	.191	-.272	.203	-.291	.249	-.301	.275	-.320	.285	-.320	.8500							
.9000	.211	-.171	.237	-.211	.238	-.283	.238	-.293	.291	-.303	.324	-.318	.401	-.323	.9000							

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING

[From reference 1]

(b) **M = 2.01 - Continued**

X/C	FRACTION OF SEMISPAN														X/C							
	0.050		0.200		0.350		0.500		0.700		0.825		0.950									
	U	L	U	L	U	L	U	L	U	L	U	L	U	L								
	a = -12																					
.0125	.448	-0.181	.512	-0.213	.543	-0.261	.531	-0.310	.538	-0.286	.528	-0.306	.498	-0.326	.0125							
.0250	.412	-0.176	.473	-0.234	.523	-0.272	.501	-0.308	.529	-0.266	.528	-0.306	.498	-0.326	.0250							
.0500	.355	-0.172	.416	-0.240	.458	-0.272	.501	-0.299	.458	-0.308	.477	-0.313	.472	-0.325	.0500							
.0750	.287	-0.165	.373	-0.232	.411	-0.268	.435	-0.297	.458	-0.308	.477	-0.313	.472	-0.325	.0750							
.1000	.294	-0.156	.351	-0.232	.394	-0.261	.435	-0.263	.408	-0.296	.440	-0.304	.443	-0.325	.1000							
.1500	.270	-0.120	.325	-0.230	.363	-0.261	.363	-0.263	.311	-0.288	.353	-0.302	.405	-0.322	.2000							
.2000	.256	-0.102	.292	-0.220	.358	-0.261	.363	-0.260	.346	-0.290	.405	-0.302	.411	-0.322	.2500							
.2500	.253	-0.119	.292	-0.227	.331	-0.260	.336	-0.279	.228	-0.290	.253	-0.311	.334	-0.321	.3000							
.3000	.267	-0.086	.279	-0.218	.280	-0.264	.325	-0.280	.217	-0.293	.240	-0.312	.263	-0.318	.3600							
.3500	.233	-0.184	.214	-0.180	.247	-0.267	.262	-0.277	.277	-0.286	.317	-0.298	.364	-0.318	.4000							
.4000	.205	-0.142	.203	-0.180	.218	-0.246	.222	-0.283	.250	-0.290	.281	-0.308	.361	-0.319	.5500							
.4500	.173	-0.157	.191	-0.198	.190	-0.245	.191	-0.286	.228	-0.290	.253	-0.311	.334	-0.321	.6500							
.5000	.154	-0.178	.123	-0.188	.169	-0.264	.173	-0.279	.217	-0.293	.240	-0.312	.263	-0.318	.7000							
.5500	.109	-0.165	.135	-0.196	.150	-0.265	.169	-0.279	.206	-0.294	.231	-0.314	.240	-0.315	.8000							
.6000	.149	-0.157	.110	-0.199	.147	-0.264	.161	-0.270	a = -10													
.0125	.393	-0.105	.461	-0.189	.501	-0.225	.490	-0.270	.506	-0.251	.487	-0.307	.492	-0.330	.0125							
.0250	.347	-0.121	.423	-0.198	.473	-0.237	.451	-0.272	.466	-0.287	.446	-0.301	.453	-0.326	.0250							
.0500	.308	-0.133	.369	-0.203	.408	-0.239	.441	-0.266	.435	-0.287	.446	-0.301	.453	-0.326	.0500							
.0750	.258	-0.133	.313	-0.198	.356	-0.233	.386	-0.259	.435	-0.287	.446	-0.301	.453	-0.326	.0750							
.1000	.245	-0.121	.299	-0.197	.335	-0.229	.376	-0.255	.375	-0.274	.391	-0.288	.418	-0.317	.1000							
.1500	.230	-0.100	.273	-0.196	.329	-0.229	.367	-0.251	.322	-0.248	.360	-0.283	.442	-0.312	.2000							
.2000	.211	-0.077	.239	-0.195	.300	-0.229	.325	-0.271	.269	-0.269	.308	-0.281	.349	-0.307	.3000							
.2500	.207	-0.095	.234	-0.191	.280	-0.227	.291	-0.247	.297	-0.269	.360	-0.283	.342	-0.312	.3600							
.3000	.227	-0.051	.236	-0.174	.239	-0.233	.275	-0.251	.224	-0.270	.267	-0.281	.309	-0.298	.4500							
.3500	.189	-0.147	.166	-0.145	.201	-0.238	.219	-0.250	.271	-0.269	.308	-0.281	.349	-0.307	.5500							
.4000	.159	-0.120	.154	-0.162	.170	-0.247	.175	-0.259	.200	-0.273	.227	-0.288	.316	-0.294	.6000							
.4500	.140	-0.140	.143	-0.179	.143	-0.240	.148	-0.261	.178	-0.275	.204	-0.293	.270	-0.289	.6500							
.5000	.118	-0.159	.085	-0.172	.123	-0.217	.130	-0.262	.166	-0.278	.191	-0.289	.218	-0.287	.7500							
.5500	.069	-0.148	.095	-0.180	.113	-0.207	.124	-0.264	.156	-0.276	.180	-0.285	.193	-0.284	.8500							
.6000	.105	-0.141	.079	-0.183	.108	-0.209	.123	-0.263	a = -8													
.0125	.347	-0.050	.412	-0.124	.460	-0.182	.496	-0.223	.474	-0.216	.451	-0.295	.478	-0.322	.0125							
.0250	.286	-0.059	.373	-0.157	.423	-0.201	.405	-0.239	.442	-0.247	.438	-0.295	.445	-0.308	.0250							
.0500	.252	-0.112	.322	-0.166	.363	-0.204	.385	-0.231	.402	-0.256	.418	-0.277	.445	-0.308	.0500							
.0750	.198	-0.105	.251	-0.162	.307	-0.198	.332	-0.226	.402	-0.256	.418	-0.277	.445	-0.308	.0750							
.1000	.195	-0.098	.249	-0.162	.286	-0.193	.320	-0.222	.340	-0.244	.348	-0.262	.385	-0.296	.1000							
.1500	.190	-0.063	.227	-0.163	.227	-0.197	.311	-0.218	.340	-0.244	.348	-0.262	.385	-0.296	.2000							
.2000	.171	-0.060	.195	-0.163	.249	-0.196	.275	-0.216	.254	-0.240	.320	-0.256	.323	-0.288	.3500							
.2500	.167	-0.076	.189	-0.159	.229	-0.198	.252	-0.217	.254	-0.244	.271	-0.258	.313	-0.285	.3500							
.3000	.189	-0.044	.191	-0.137	.205	-0.204	.235	-0.222	.234	-0.244	.271	-0.258	.313	-0.285	.3500							
.3500	.149	-0.116	.127	-0.120	.156	-0.208	.180	-0.222	.186	-0.244	.229	-0.259	.269	-0.283	.4000							
.4000	.122	-0.106	.113	-0.147	.128	-0.217	.131	-0.233	.157	-0.251	.186	-0.266	.266	-0.274	.4500							
.4500	.105	-0.124	.102	-0.163	.102	-0.192	.107	-0.239	.139	-0.254	.164	-0.272	.226	-0.266	.5000							
.5000	.084	-0.146	.053	-0.155	.084	-0.181	.089	-0.238	.126	-0.256	.149	-0.275	.175	-0.263	.7500							
.5500	.037	-0.135	.062	-0.166	.077	-0.189	.085	-0.238	.116	-0.256	.189	-0.271	.151	-0.258	.8500							
.6000	.071	-0.127	.049	-0.170	.072	-0.193	.084	-0.233	a = -6													
.0125	.294	.036	.367	-.057	.402	-.120	.402	-.172	.422	-.166	.406	-.258	.457	-.299	.0125							
.0250	.226	.058	.261	-.117	.305	-.155	.324	-.184	.422	-.166	.406	-.258	.457	-.299	.0250							
.0500	.193	.068	.261	-.117	.305	-.155	.324	-.184	.422	-.166	.406	-.258	.457	-.299	.0500							
.0750	.154	-.076	.197	-.116	.252	-.182	.271	-.180	.349	-.217	.374	-.235	.499	-.277	.0750							
.1000	.148	-.064	.190	-.120	.225	-.148	.250	-.177	.349	-.217	.374	-.235	.499	-.277	.1000							
.1500	.122	-.040	.171	-.126	.171	-.157	.157	-.176	.287	-.206	.292	-.225	.434	-.263	.1500							
.2000	.119	-.054	.135	-.124	.193	-.159	.182	-.176	.211	-.204	.266	-.221	.273	-.257	.2000							
.2500	.119	-.081	.135	-.114	.176	-.159	.195	-.178	.211	-.204	.266	-.221	.273	-.257	.3000							
.3000	.147	-.094	.137	-.102	.156	-.167	.184	-.184	.184	-.218	.228	-.227	.263	-.259	.3500							
.3500	.105	-.095	.087	-.097	.112	-.173	.133	-.196	.144	-.211	.188	-.232	.220	-.260	.4000							
.4000	.078	-.086	.070	-.133	.083	-.163	.084	-.204	.110	-.221	.140	-.242	.216	-.260	.4500							
.4500	.033	-.082	.061	-.141	.058	-.153	.063	-.207	.093	-.224	.121	-.248	.176	-.260	.5000							
.5000	.062	-.106	.061	-.136	.041	-.160	.044	-.209	.081	-.233	.104	-.255	.125	-.253	.5500							
.5500	.041	-.130	.018	-.136	.041	-.160	.044	-.209	.084	-.236	.098	-.261	.101	-.246	.6000							
.6000	.003	-.122	.023	-.149	.036	-.175	.040	-.205	.081	-.233	.104	-.255	.125	-.253	.6500							
.6500	.032	-.112	.016	-.155	.032	-.171	.038	-.184	.084	-.236	.098	-.261	.101	-.246	.7000							
.7000	.000	-.122	.023	-.149	.036	-.175	.040	-.205	.081	-.233	.104	-.255	.125	-.253	.7500							
.7500	.000	-.135	.016	-.155	.032	-.171	.038	-.184	.084	-.236	.098	-.261	.101	-.246	.8000							
.8000	.000	-.112	.016	-.155	.032	-.171	.038	-.184	.084	-.236	.098	-.261	.101	-.246	.8500							
.8500	.000	-.112	.016	-.155	.032	-.171	.038	-.184	.084	-.236	.098	-.261	.101	-.246	.9000							

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING
[From reference 1]
(b) M = 2.01 - Continued

X/C	FRACTION OF SEMISPAN														X/C	
	0.050		0.200		0.350		0.500		0.700		0.825		0.950			
	U	L	U	L	U	L	U	L	U	L	U	L	U	L		
$\alpha = 0^\circ$																
.0125	+.256	+.098	+.294	+.014	+.353	-.042	+.359	-.014	+.378	-.112	+.363	-.215	+.424	-.257	.0125	
.0250	+.190	+.018	+.262	-.049	+.313	-.086	+.299	-.122	+.317	-.153	+.300	-.174	+.326	-.193	.0250	
.0500	.148	-.058	+.222	-.071	+.263	-.107	+.279	-.153	+.299	-.174	+.300	-.193	+.356	-.237	.0500	
.0750	.124	-.042	+.164	-.073	+.214	-.111	+.228	-.150	+.237	-.165	+.245	-.183	+.302	-.225	.0750	
.1000	.113	-.030	+.142	-.078	+.182	-.109	+.196	-.130	+.203	-.135	+.237	-.165	+.302	-.225	.1000	
.1500	.110	-.012	+.131	-.084	+.182	-.118	+.203	-.135	+.237	-.165	+.245	-.183	+.302	-.225	.1500	
.2000	.101	-.012	+.111	-.074	+.155	-.121	+.171	-.138	+.153	-.173	+.190	-.193	+.224	-.225	.2000	
.2500	.092	-.022	+.102	-.064	+.138	-.119	+.154	-.143	+.166	-.163	+.217	-.184	+.228	-.221	.2500	
.3000	.118	-.019	+.101	-.065	+.117	-.131	+.135	-.152	+.153	-.173	+.190	-.193	+.224	-.225	.3000	
.3500																
.4000	.075	-.069	+.058	-.073	+.088	-.125	+.099	-.154	+.114	-.176	+.154	-.201	+.186	-.229	.4000	
.4500																
.5000	.051	-.064	+.040	-.109	+.053	-.128	+.046	-.172	+.073	-.190	+.104	-.214	+.177	-.234	.5000	
.5500																
.6000	.039	-.085	+.030	-.110	+.028	-.131	+.030	-.174	+.055	-.195	+.083	-.221	+.135	-.238	.6000	
.6500																
.7000	.018	-.111	-.007	-.108	.011	-.145	.010	-.172	.049	-.206	.062	-.229	+.088	-.237	.7000	
.7500																
.8000	-.019	-.106	-.001	-.125	.007	-.157	.006	-.154	.037	-.207	.058	-.237	+.061	-.229	.8000	
.8500																
.9000	.009	-.094	-.007	-.130	.006	-.148	.005	-.158								
$\alpha = -2^\circ$																
.0125	+.210	+.142	+.219	+.073	+.290	+.019	+.298	+.019	+.326	-.041	+.308	-.154	+.389	-.204	.0125	
.0250	.144	-.052	+.188	+.001	+.246	-.040	+.252	-.058	+.261	-.041	+.308	-.154	+.389	-.204	.0250	
.0500	.088	-.024	+.165	-.026	+.207	-.061	+.237	-.074	+.242	-.120	+.269	-.133	+.309	-.188	.0500	
.0750	.065	-.004	+.156	-.030	+.164	-.059	+.193	-.081	+.212	-.172	+.269	-.133	+.309	-.188	.0750	
.1000	.077	-.002	+.162	-.036	+.119	-.053	+.146	-.080	+.212	-.172	+.269	-.133	+.309	-.188	.1000	
.1500	.073	.011	+.080	-.036	+.096	-.063	+.145	-.088	+.212	-.172	+.269	-.133	+.309	-.188	.1500	
.2000	.067	.012	+.057	-.033	+.097	-.055	+.121	-.094	+.212	-.112	+.200	-.131	+.256	-.182	.2000	
.2500	.060	.005	+.042	-.034	+.092	-.070	+.109	-.095	+.212	-.115	+.187	-.135	+.176	-.160	.2500	
.3000	.079	-.003	+.050	-.042	+.068	-.078	+.092	-.105	+.212	-.129	+.148	-.150	+.178	-.184	.3000	
.3500																
.4000	.040	-.043	+.027	-.055	+.046	-.081	+.061	-.110	+.212	-.129	+.148	-.150	+.178	-.184	.4000	
.4500																
.5000	.018	-.041	+.006	-.089	+.013	-.100	+.016	-.124	+.212	-.141	+.115	-.161	+.143	-.193	.5000	
.5500																
.6000	.008	-.065	-.002	-.092	-.010	-.109	-.009	-.122	+.044	-.154	+.071	-.179	+.139	-.201	.6000	
.6500																
.7000	-.012	-.091	-.035	-.090	-.028	-.125	-.021	-.122	+.008	-.170	+.033	-.195	+.055	-.207	.7000	
.7500																
.8000	-.043	-.091	-.029	-.111	-.030	-.131	-.027	-.126	-.005	-.171	+.021	-.206	+.027	-.199	.8000	
.8500																
.9000	.018	-.073	-.035	-.119	-.028	-.123	-.029	-.137								
$\alpha = 0^\circ$																
.0125	.163	.188	.154	.127	.223	.074	.230	.063	.270	.039	.297	.079	.339	.137	.0125	
.0250	.108	.087	.121	.050	.176	.013	.191	.002	.270	.039	.297	.079	.339	.137	.0250	
.0500	.044	.014	.107	.027	.151	.002	.191	-.024	.270	.039	.297	.079	.339	.137	.0500	
.0750	.045	.031	.079	.024	.124	.012	.164	-.022	.273	.062	.213	-.067	.253	.124	.0750	
.1000	.036	.015	.048	.016	.088	.007	.113	-.017	.273	.062	.213	-.067	.253	.124	.1000	
.1500	.045	.017	.035	.012	.088	.015	.088	-.031	.273	.099	.054	.147	.077	.208	.1500	
.2000	.038	.037	.057	.016	.045	.022	.066	-.034	.273	.099	.054	.147	.077	.208	.2000	
.2500	.039	.021	.031	-.003	.047	-.017	.059	-.040	.273	.099	.054	.147	.077	.208	.2500	
.3000	.041	.021	.023	-.017	.032	-.036	.044	-.058							.3000	
.3500																
.4000	.011	-.013	-.005	-.033	.012	-.047	.023	-.058							.4000	
.4500																
.5000	-.011	-.016	-.028	-.054	-.026	-.077	-.016	-.086							.5000	
.5500																
.6000	-.019	-.041	-.035	-.062	-.039	-.083	-.041	-.091							.6000	
.6500																
.7000	-.037	-.070	-.062	-.065	-.058	-.098	-.056	-.100							.7000	
.7500																
.8000	-.061	-.075	-.057	-.085	-.058	-.099	-.055	-.109							.8000	
.8500																
.9000	-.038	-.053	-.062	-.095	-.054	-.096	-.061	-.116							.9000	
$\alpha = 2^\circ$																
.0125	.113	.233	.093	.181	.154	.142	.141	.136							.0125	
.0250	.065	.124	.060	.110	.097	.090	.105	.072							.0250	
.0500	.007	.073	.043	.106	.076	.091	.111	.064							.0500	
.0750	.006	.059	.034	.085	.074	.085	.119	.074							.0750	
.1000	.013	.042	.000	.059	.051	.056	.078	.063							.1000	
.1500	.018	.066	-.011	.058	.027	.037	.014	.058							.1500	
.2000	.014	.067	-.008	.048	-.005	.028	.017	.009							.2000	
.2500	.041	.038	-.001	.029	-.006	.025	-.007	.013							.2500	
.3000	.005	.058	-.004	.015	-.008	-.001	-.010	.014							.3000	
.3500																
.4000	-.030	.019	-.031	-.001	-.023	-.016	-.028	-.020							.4000	
.4500																
.5000	-.039	.013	-.065	-.020	-.047	-.051	-.055	-.057							.5000	
.5500																
.6000	-.045	-.013	-.063	-.032	-.065	-.058	-.089	-.056							.6000	
.6500																
.7000	-.062	-.048	-.085	-.040	-.092	-.069	-.086	-.072							.7000	
.7500																
.8000	-.074	-.059	-.078	-.062	-.095	-.073	-.083	-.084							.8000	
.8500																
.9000	-.051	-.037	-.082	-.073	-.085	-.072	-.091	-.01							.9000	

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING
[From reference 1]

(b) M = 2.01 - Continued

ACTION OF SEMISPAN

X/C	FRACTION OF SEMISPAN														X/C	
	0.050		0.200		0.350		0.500		0.700		0.825		0.950			
	U	L	U	L	U	L	U	L	U	L	U	L	U	L		
$a = 0.4$																
.0125	.057	.277	.030	.268	.089	.224	.050	.224	.162	.096	.213	.107	.125	.221	.104	.0125
.0250	.057	.267	.004	.196	.028	.172	.026	.162	.144	.096	.213	.107	.125	.221	.104	.0250
.0500	.024	.130	-.015	.170	.005	.162	.027	.144	.139	.059	.122	.084	.103	.128	.066	.0500
.0750	.034	.089	-.022	.133	.006	.135	.033	.139	.059	.122	.084	.103	.128	.066	.0750	
.1000	.021	.072	-.041	.116	-.003	.104	.022	.111	.070	.024	.076	.062	.081	.108	.033	.1000
.1500	.006	.104	-.060	.114	.085	.015	.070	.024	.076	.062	.081	.108	.033	.1500	.0200	
.2000	.012	.099	-.055	.091	-.045	.076	.030	.074	.033	.033	.022	-.007	.003	.020	-.030	.0200
.2500	.019	.071	-.044	.066	-.053	.069	-.034	.067	.001	.043	.010	.022	.056	.021	.0250	.0300
.3000	.019	.100	-.034	.057	-.064	.038	-.049	.033	.033	.033	.003	.020	-.030	.0350	.0350	
.3500	.004	.104	-.049	.034	-.063	.027	-.071	.021	.033	.022	-.007	.003	.020	-.030	.0400	.0400
.4000	-.061	.051	-.049	.034	-.063	.027	-.071	.021	.048	-.005	.020	-.005	.002	-.053	.0500	.0500
.4500	-.066	.042	-.089	.015	-.071	-.019	-.107	-.012	.048	-.005	.020	-.005	.002	-.053	.0500	.0500
.5000	-.069	.017	-.089	.000	-.087	-.026	-.118	-.020	.078	-.027	-.049	-.035	.011	-.069	.0550	.0550
.6000	-.089	.022	-.108	-.009	-.113	-.038	-.115	-.048	.092	-.039	-.045	-.033	-.037	-.081	.0700	.0700
.7500	-.083	.022	-.108	-.009	-.113	-.038	-.115	-.048	.102	-.064	-.080	-.076	-.063	-.090	.0800	.0800
.8000	-.091	.042	-.098	-.033	-.125	-.044	-.119	-.055	.111	-.070	-.089	-.099	-.078	-.087	.0850	.0850
.8500	-.069	.014	-.101	-.052	-.112	-.042	-.114	-.055	.111	-.070	-.089	-.099	-.078	-.087	.0900	.0900
$a = 0.6$																
.0125	.002	.315	-.026	.324	-.026	.291	-.027	.294	.015	.284	.030	.218	.138	.192	.0125	.0125
.0250	-.023	.208	-.053	.248	-.030	.239	-.037	.206	.030	.186	.014	.178	.054	.135	.0500	.0500
.0500	-.058	.187	-.065	.216	-.054	.216	-.037	.193	-.007	.186	.005	.195	.094	.155	.0750	.0750
.0750	-.068	.126	-.069	.176	-.057	.181	-.030	.156	-.027	.132	.005	.195	.023	.093	.1000	.1000
.1000	-.054	.104	-.081	.168	-.061	.153	-.034	.127	-.057	.130	-.032	.132	.005	.195	.1500	.2000
.1500	-.039	.137	-.094	.145	-.072	.125	-.072	.126	-.027	.106	-.043	.104	-.037	.003	.064	.2500
.2000	-.037	.130	-.103	.126	-.091	.107	-.077	.106	-.089	.073	-.070	.058	-.052	.049	-.023	.3000
.2500	-.004	.100	-.089	.099	-.099	.107	-.077	.106	-.089	.073	-.070	.058	-.052	.049	-.023	.3500
.3000	-.040	.132	-.069	.088	-.107	.079	-.089	.073	-.089	.073	-.070	.058	-.052	.049	-.023	.4000
.3500	-.080	.080	-.079	.060	-.120	.065	-.106	.061	-.083	.038	-.064	.035	-.048	.005	.0450	.4000
.4000	-.087	.070	-.114	.040	-.113	.012	-.138	.029	-.108	.009	-.087	.005	-.052	-.018	.0500	.5000
.4500	-.087	.070	-.114	.040	-.113	.012	-.138	.029	-.108	.009	-.087	.005	-.052	-.018	.0500	.5000
.5000	-.090	.044	-.122	.023	-.119	.002	-.152	.005	-.122	-.005	-.099	-.025	-.077	-.043	-.060	.6000
.6000	-.103	.001	-.133	.013	-.138	-.005	-.161	-.016	-.130	-.031	-.111	-.049	-.094	-.066	-.066	.7000
.7500	-.107	.025	-.124	-.012	-.149	-.015	-.159	-.023	-.139	-.041	-.120	-.071	-.099	-.085	-.085	.8000
.8000	-.091	.008	-.126	-.028	-.148	-.015	-.143	-.025	-.139	-.041	-.120	-.071	-.099	-.085	-.085	.9000
$a = 0.8$																
.0125	-.057	.358	-.078	.382	-.045	.358	-.099	.358	-.058	.355	-.040	.305	-.044	.273	.0125	.0125
.0250	-.079	.259	-.101	.311	-.086	.304	-.110	.308	-.058	.355	-.040	.305	-.044	.273	.0250	.0250
.0500	-.090	.233	-.113	.267	-.106	.274	-.097	.280	-.072	.254	-.051	.245	-.012	.210	.0750	.0750
.0750	-.098	.164	-.117	.230	-.106	.226	-.088	.251	-.072	.254	-.051	.245	-.012	.210	.1000	.1000
.1000	-.084	.156	-.124	.218	-.109	.206	-.088	.208	-.072	.254	-.051	.245	-.012	.210	.1500	.2000
.1500	-.070	.176	-.139	.191	-.131	.184	-.100	.189	-.084	.192	-.055	.188	-.038	.153	.1500	.2000
.2000	-.058	.171	-.141	.164	-.131	.169	-.110	.177	-.095	.149	-.082	.143	-.055	.104	.2500	.2500
.2500	-.025	.138	-.132	.143	-.136	.151	-.119	.152	-.095	.149	-.082	.143	-.055	.104	.3000	.3000
.3000	-.055	.166	-.109	.132	-.146	.124	-.126	.121	-.112	.118	-.095	.098	-.073	.070	.3500	.3500
.3500	-.105	.116	-.100	.095	-.155	.099	-.142	.104	-.123	.082	-.107	.088	-.091	.040	.4000	.4000
.4000	-.105	.075	-.149	.104	-.130	.074	-.171	.049	-.169	.057	-.142	.047	-.126	.037	-.101	.5000
.4500	-.105	.105	-.130	.074	-.171	.049	-.169	.057	-.142	.047	-.126	.037	-.101	.022	.5500	.5500
.5000	-.105	.104	-.130	.074	-.171	.049	-.169	.057	-.142	.047	-.126	.037	-.101	.022	.5500	.5500
.6000	-.108	.077	-.149	.056	-.147	.036	-.181	.037	-.123	.030	-.136	.016	-.114	-.001	.6500	.6500
.7500	-.122	.030	-.153	.043	-.155	.027	-.193	.020	-.159	-.002	-.146	-.016	-.121	-.030	.7500	.7500
.8000	-.121	.001	-.145	.016	-.165	.018	-.195	.013	-.167	-.006	-.153	-.040	-.122	-.059	.8500	.8500
.9000	-.108	.037	-.144	-.001	-.168	-.015	-.195	.008	-.148	-.015	-.153	-.016	-.121	-.031	.9000	.9000
$a = 10$																
.0125	-.051	.403	-.122	.439	-.098	.421	-.148	.409	-.118	.412	-.103	.371	-.059	.334	.0125	.0125
.0250	-.129	.328	-.143	.372	-.137	.364	-.163	.363	-.118	.421	-.103	.371	-.059	.334	.0250	.0250
.0500	-.124	.270	-.150	.318	-.151	.321	-.151	.321	-.123	.309	-.108	.302	-.078	.269	.0750	.0750
.0750	-.121	.204	-.156	.278	-.149	.275	-.138	.298	-.123	.309	-.108	.302	-.078	.269	.0750	.0750
.1000	-.111	.209	-.160	.270	-.149	.259	-.137	.259	-.123	.238	-.105	.240	-.090	.200	.1500	.2000
.1500	-.090	.219	-.169	.239	-.142	.234	-.142	.241	-.129	.238	-.105	.240	-.090	.200	.2000	.2000
.2000	-.076	.216	-.174	.207	-.167	.216	-.149	.231	-.123	.238	-.105	.240	-.090	.200	.2500	.2500
.2500	-.063	.176	-.167	.187	-.168	.195	-.155	.194	-.136	.194	-.128	.209	-.105	.148	.3000	.3000
.3000	-.066	.202	-.151	.174	-.176	.171	-.160	.174	-.148	.153	-.136	.156	-.116	.121	.3500	.3500
.3500	-.125	.152	-.123	.133	-.187	.137	-.172	.147	-.159	.113	-.144	.125	-.131	.085	.4000	.4000
.4000	-.125	.152	-.123	.133	-.187	.137	-.172	.147	-.159	.113	-.144	.125	-.131	.085	.4500	.4500
.4500	-.121	.140	-.144	.112	-.200	.085	-.195	.090	-.174	.079	-.161	.071	-.137	.085	.5500	.5500
.5000	-.121	.140	-.144	.112	-.200	.085	-.195	.090	-.174	.079	-.161	.071	-.137	.085	.5500	.5500
.6000	-.124	.119	-.164	.090	-.203	.068	-.206	.075	-.184	.058	-.169	.046	-.147	.035	.6500	.6500
.6500	-.121	.070	-.157	.023	-.183	.048	-.221	.040	-.188	.033	-.178	.019	-.147	.001	.7000	.7000
.7500	-.136	.061	-.168	.076	-.186	.059	-.215	.054	-.187	.033	-.178	.019	-.147	.001	.8000	.8000
.8000	-.130	.027	-.159	.046	-.182	.053	-.218	.047	-.187	.033	-.178	.019	-.147	.001	.8500	.8500
.9000	-.121	.070	-.157	.023	-.183	.048	-.221	.040	-.188	.033	-.178	.019	-.147	.001	.9000	.9000

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR LINEAR TWIST WING
[From reference 1]

(b) $M = 2.01$ - Continued

X/C	FRACTION OF SEMISPAN														X/C		
	0.050		0.200		0.350		0.500		0.700		0.825		0.950				
	U	L	U	L	U	L	U	L	U	L	U	L	U	L			
	a: 12																
.0125	-129	.464	-168	.500	-157	.488	-197	.458	-208	.424	-180	.469	-172	.429	-151	.390	.0125
.0250	-165	.401	-183	.436	-179	.426	-208	.424	-200	.385	-190	.351	-189	.351	-154	.324	.0250
.0500	-168	.317	-187	.381	-190	.381	-200	.385	-200	.351	-189	.317	-163	.357	-154	.324	.0500
.0750	-156	.256	-191	.338	-186	.327	-186	.327	-189	.351	-176	.369	-163	.357	-154	.324	.0750
.1000	-153	.252	-194	.323	-184	.314	-184	.314	-185	.302	-176	.297	-152	.288	-151	.246	.1000
.1500	-105	.265	-202	.288	-196	.245	-196	.245	-197	.282	-176	.297	-152	.288	-151	.246	.1500
.2000	-099	.254	-208	.260	-198	.268	-188	.283	-188	.283	-176	.232	-172	.243	-155	.200	.2000
.2500	-088	.223	-202	.241	-196	.245	-191	.245	-197	.242	-179	.232	-172	.243	-155	.250	.2500
.3000	-072	.245	-192	.223	-202	.219	-197	.223	-197	.223	-179	.232	-172	.243	-155	.300	.3000
.3500																	
.4000	-149	.196	-159	.177	-212	.178	-206	.190	-186	.197	-177	.213	-162	.170	-170	.3500	
.4500																	
.5000	-142	.184	-160	.157	-224	.126	-224	.131	-195	.196	-183	.173	-174	.127	-127	.4500	
.5500																	
.6000	-142	.162	-179	.138	-234	.109	-233	.116	-206	.123	-194	.113	-179	.112	-179	.5500	
.6500																	
.7000	-154	.094	-186	.120	-234	.101	-241	.096	-214	.106	-203	.090	-182	.080	-182	.6500	
.7500																	
.8000	-145	.064	-175	.086	-215	.090	-241	.087	-218	.072	-207	.063	-182	.041	-182	.8000	
.8500																	
.9000	-138	.107	-172	.062	-203	.082	-245	.082	-224	.067	-213	.032	-186	.012	-186	.8500	.9000
a: 14																	
.0125	-157	.521	-208	.545	-216	.534	-241	.497	-229	.413	-221	.474	-211	.433	-211	.433	.0125
.0250	-192	.461	-218	.486	-225	.465	-234	.487	-229	.413	-221	.474	-211	.433	-211	.433	.0250
.0500	-190	.366	-218	.434	-227	.432	-235	.442	-230	.371	-225	.420	-210	.404	-207	.376	.0500
.0750	-183	.312	-220	.387	-223	.371	-225	.398	-215	.420	-210	.404	-207	.376	-207	.376	.0750
.1000	-187	.304	-224	.377	-221	.362	-222	.369	-221	.354	-210	.344	-190	.336	-198	.290	.1000
.1500	-180	.313	-227	.338	-221	.344	-219	.331	-220	.344	-190	.336	-198	.290	-198	.1500	
.2000	-100	.300	-232	.305	-227	.322	-219	.331	-220	.325	-208	.375	-204	.294	-198	.246	.2000
.2500	-126	.267	-221	.293	-227	.290	-220	.290	-220	.290	-208	.275	-204	.294	-198	.246	.2500
.3000	-074	.289	-210	.261	-233	.267	-222	.272	-215	.242	-209	.256	-198	.210	-198	.210	.3000
.3500																	
.4000	-181	.239	-198	.219	-236	.221	-230	.232	-223	.221	-194	-215	-204	-204	-167	-167	.4000
.4500																	
.5000	-159	.225	-189	.196	-248	.167	-248	.173	-231	.170	-225	.156	-211	.154	-211	.154	.5000
.6000	-157	.201	-198	.175	-256	.149	-254	.157	-235	.143	-228	.129	-212	.123	-212	.123	.6000
.6500																	
.7000	-148	.130	-205	.154	-282	.141	-261	.137	-237	.111	-231	.098	-210	.080	-210	.080	.7000
.7500																	
.8000	-157	.100	-192	.119	-255	.127	-259	.127	-241	.108	-236	.067	-215	.046	-215	.046	.8000
.8500																	
.9000	-151	.144	-189	.093	-239	.120	-263	.119	-244	.120	-239	.120	-235	.109	-235	.109	.9000
a: 16																	
.0125	-181	.578	-240	.539	-257	.578	-272	.533	-269	.548	-256	.510	-249	.477	-226	.477	.0125
.0250	-214	.507	-245	.533	-257	.567	-272	.533	-269	.548	-256	.510	-249	.477	-226	.477	.0250
.0500	-208	.412	-243	.484	-255	.483	-243	.484	-247	.443	-234	.467	-247	.453	-242	.417	.0500
.0750	-205	.373	-243	.458	-250	.458	-246	.458	-246	.443	-234	.467	-247	.453	-242	.417	.0750
.1000	-209	.347	-244	.428	-248	.413	-249	.424	-246	.424	-236	.404	-246	.397	-231	.336	.1000
.1500	-230	.362	-246	.382	-248	.400	-246	.404	-246	.394	-220	.397	-231	.336	-231	.336	.1500
.2000	-094	.349	-244	.351	-248	.366	-245	.380	-239	.325	-235	.346	-228	.292	-226	.292	.2000
.2500	-131	.316	-237	.337	-249	.336	-244	.337	-239	.325	-235	.346	-228	.292	-226	.292	.2500
.3000	-073	.333	-227	.304	-253	.308	-244	.320	-244	.281	-238	.293	-228	.254	-228	.254	.3000
.3500																	
.4000	-202	.283	-224	.259	-258	.265	-249	.272	-244	.281	-238	.293	-228	.254	-228	.254	.4000
.4500																	
.5000	-163	.266	-218	.235	-266	.200	-263	.217	-248	.237	-241	.252	-231	.216	-231	.216	.5000
.5500																	
.6000	-166	.242	-221	.211	-273	.189	-269	.200	-256	.211	-249	.201	-234	.200	-234	.200	.6000
.6500																	
.7000	-178	.169	-225	.190	-278	.182	-274	.177	-259	.152	-255	.138	-235	.115	-234	.115	.7000
.7500																	
.8000	-167	.140	-217	.153	-277	.162	-274	.166	-261	.147	-245	.109	-235	.081	-235	.081	.8000
.8500																	
.9000	-161	.186	-217	.127	-258	.155	-276	.153	-263	.137	-261	.143	-244	.104	-233	.104	.9000
a: 18																	
.0125	-215	.634	-276	.634	-287	.616	-305	.564	-284	.582	-286	.550	-296	.520	-250	.520	.0125
.0250	-233	.544	-276	.529	-289	.558	-303	.575	-284	.582	-286	.550	-296	.520	-250	.520	.0250
.0500	-232	.457	-271	.451	-285	.534	-295	.551	-284	.549	-286	.519	-276	.515	-286	.515	.0500
.0750	-228	.431	-268	.510	-278	.466	-286	.495	-278	.519	-276	.515	-286	.452	-276	.452	.0750
.1000	-230	.393	-265	.477	-274	.472	-282	.472	-274	.486	-274	.448	-262	.452	-273	.438	.1000
.1500	-239	.414	-264	.436	-267	.451	-279	.450	-269	.448	-262	.437	-262	.437	-260	.438	.1500
.2000	-121	.399	-262	.406	-273	.417	-275	.428	-274	.428	-274	.426	-262	.437	-262	.438	.2000
.2500	-144	.346	-260	.386	-271	.433	-272	.390	-271	.436	-271	.368	-262	.397	-260	.338	.2500
.3000	-089	.380	-254	.357	-274	.355	-271	.368	-263	.332	-261	.343	-264	.303	-260	.303	.3000
.3500																	
.4000	-216	.330	-248	.311	-278	.317	-275	.318	-263	.332	-261	.343	-264	.303	-260	.303	.4000
.4500					</												

TABLE VI.- Concluded
 PRESSURE COEFFICIENTS FOR LINEAR TWIST WING
 [From reference 1]
 (b) $M = 2.01$ - Concluded

X/C	FRACTION OF SEMISPAN														X/C	
	0.050		0.200		0.350		0.500		0.700		0.825		0.950			
	U	L	U	L	U	L	U	L	U	L	U	L	U	L		
	d = 20															
.0125	+.230	.883	-.293	.676	+.305	.654	-.316	.589	-.294	.610	-.294	.582	-.306	.534	.0125	
.0250	+.264	.587	-.294	.644	+.303	.593	-.294	.585	-.307	.595	-.294	.568	-.291	.556	-.299	
.0750	+.233	.491	-.282	.582	-.295	.522	-.302	.541	-.290	.568	-.291	.556	-.299	.498	.0750	
.1000	+.244	.445	-.279	.528	-.292	.525	-.298	.520	-.294	.498	-.285	.498	-.257	.497	-.289	
.1500	+.258	.470	-.276	.491	-.279	.510	-.294	.520	-.285	.498	-.257	.497	-.289	.397	.1500	
.2000	+.198	.457	-.275	.460	-.288	.468	-.290	.483	-.287	.419	-.274	.443	-.283	.387	.2000	
.2500	+.149	.421	-.274	.434	-.288	.436	-.286	.439	-.280	.419	-.274	.400	-.279	.350	.2500	
.3000	+.109	.429	-.277	.403	-.290	.405	-.287	.414	-.277	.374	-.277	.388	-.280	.348	.3000	
.3500															.3500	
.4000	+.223	.381	-.275	.356	-.293	.372	-.288	.367	-.277	.374	-.277	.388	-.280	.348	.4000	
.4500															.4500	
.5000	+.190	.358	-.269	.327	-.296	.278	-.297	.302	-.280	.332	-.280	.350	-.279	.322	.5000	
.5500															.5500	
.6000	+.198	.331	-.266	.294	-.297	.278	-.300	.296	-.283	.306	-.282	.291	-.279	.288	.6000	
.6500															.6500	
.7000	+.204	.252	-.273	.271	-.296	.264	-.303	.268	-.287	.277	-.280	.252	-.277	.243	.7000	
.7500															.7500	
.8000	+.191	.226	-.270	.229	-.292	.248	-.299	.248	-.285	.242	-.284	.221	-.275	.199	.8000	
.8500															.8500	
.9000	+.187	.274	-.266	.215	-.271	.238	-.242	.241	-.286	.234	-.281	.191	-.273	.167	.9000	

TABLE VII
SECTION AERODYNAMIC CHARACTERISTICS FOR CAMBERED AND TWISTED WING
(a) $M = 1.61$

α , deg	y/b $\frac{2}{2}$						
	.05	.20	.35	.50	.70	.825	.95
C_n							
-20	-0.991	-1.018	-0.921	-0.768	-0.506	-0.438	-0.276
-18	-0.878	-0.910	-0.852	-0.705	-0.435	-0.382	-0.224
-16	-0.778	-0.803	-0.796	-0.646	-0.393	-0.351	-0.194
-14	-0.673	-0.695	-0.734	-0.603	-0.355	-0.312	-0.174
-12	-0.566	-0.590	-0.651	-0.538	-0.319	-0.282	-0.155
-10	-0.468	-0.487	-0.545	-0.482	-0.283	-0.252	-0.133
-08	-0.372	-0.392	-0.435	-0.416	-0.244	-0.221	-0.111
-06	-0.275	-0.275	-0.319	-0.334	-0.194	-0.182	-0.089
-04	-0.184	-0.183	-0.210	-0.250	-0.146	-0.143	-0.076
-02	-0.090	-0.093	-0.112	-0.162	-0.098	-0.104	-0.067
00	.001	.012	-0.011	-0.057	-0.043	-0.061	-0.045
02	.102	.119	.093	.050	.019	.010	.016
04	.200	.223	.196	.157	.090	.049	.019
06	.292	.319	.299	.262	.158	.113	.063
08	.398	.429	.412	.364	.228	.189	.117
10	.493	.527	.498	.451	.272	.246	.155
12	.592	.626	.589	.516	.322	.291	.192
14	.692	.732	.686	.590	.370	.343	.231
16	.811	.850	.798	.681	.424	.417	.294
18	.947	.990	.912	.785	.509	.476	.315
20	1.086	1.139	1.036	.902	.552	.494	.319
C_m							
-20	-0.393	-0.116	.139	.303	.403	.401	.302
-18	-0.369	-0.133	.121	.272	.344	.345	.246
-16	-0.338	-0.139	.109	.244	.309	.315	.211
-14	-0.307	-0.141	.095	.227	.280	.277	.188
-12	-0.273	-0.140	.074	.198	.252	.248	.167
-10	-0.242	-0.135	.039	.174	.224	.220	.142
-08	-0.213	-0.129	.006	.144	.192	.192	.117
-06	-0.181	-0.117	-.019	.107	.152	.155	.092
-04	-0.147	-0.110	-.042	.070	.113	.120	.077
-02	-0.116	-.099	-.059	.030	.075	.084	.068
00	-.086	-.085	-.073	-.017	.030	.044	.044
02	-.052	-.072	-.086	-.064	-.021	-.001	.012
04	-.019	-.058	-.097	-.109	-.079	-.056	-.026
06	.009	-.045	-.107	-.148	-.137	-.115	-.074
08	.044	-.030	-.119	-.182	-.199	-.183	-.133
10	.074	-.017	-.122	-.211	-.237	-.232	-.173
12	.104	-.003	-.126	-.227	-.276	-.271	-.213
14	.137	.010	-.139	-.252	-.312	-.317	-.255
16	.172	.019	-.160	-.285	-.350	-.383	-.322
18	.209	.030	-.173	-.325	-.418	-.430	-.344
20	.254	.045	-.195	-.374	-.454	-.446	-.347

TABLE VII.- Concluded

SECTION AERODYNAMIC CHARACTERISTICS FOR CAMBERED AND TWISTED WING

(b) $M = 2.01$

α , deg	$y/b/2$						
	.05	.20	.35	.50	.70	.825	.95
C_n							
-20	-.836	-.786	-.690	-.574	-.427	-.328	-.223
-18	-.755	-.740	-.657	-.544	-.404	-.307	-.208
-16	-.670	-.670	-.605	-.504	-.375	-.281	-.192
-14	-.581	-.581	-.550	-.460	-.344	-.260	-.175
-12	-.489	-.493	-.490	-.412	-.313	-.239	-.160
-10	-.411	-.414	-.423	-.363	-.282	-.222	-.148
-08	-.331	-.332	-.351	-.311	-.245	-.200	-.138
-06	-.247	-.251	-.284	-.243	-.209	-.177	-.128
-04	-.165	-.164	-.192	-.184	-.171	-.148	-.113
-02	-.049	-.043	-.070	-.087	-.102	-.102	-.084
00	-.011	-.005	-.028	-.047	-.078	-.085	-.073
02	.062	.070	.059	.028	-.025	-.055	-.052
04	.148	.158	.134	.097	.035	-.000	-.024
06	.221	.233	.205	.170	.091	.041	.013
08	.305	.318	.285	.237	.149	.097	.052
10	.387	.394	.363	.307	.198	.148	.094
12	.476	.486	.429	.378	.257	.196	.045
14	.566	.576	.509	.447	.309	.246	.058
16	.658	.668	.582	.509	.358	.293	.069
18	.747	.754	.653	.568	.404	.323	.081
20	.843	.847	.734	.632	.445	.359	.087
C_m							
-20	-.335	-.104	.083	.209	.293	.294	.243
-18	-.314	-.103	.077	.196	.276	.274	.226
-16	-.290	-.107	.067	.178	.254	.249	.208
-14	-.262	-.110	.055	.159	.230	.229	.189
-12	-.234	-.112	.041	.138	.207	.210	.172
-10	-.208	-.110	.025	.116	.184	.193	.158
-08	-.183	-.106	.010	.092	.157	.172	.147
-06	-.156	-.099	-.006	.061	.130	.150	.137
-04	-.127	-.092	-.026	.035	.103	.123	.119
-02	-.087	-.078	-.049	-.006	.051	.081	.086
00	-.075	-.074	-.056	-.024	.032	.065	.074
02	-.050	-.064	-.068	-.058	-.008	.036	.050
04	-.023	-.055	-.081	-.085	-.052	-.015	.019
06	.001	-.046	-.088	-.112	-.091	-.053	-.023
08	.028	-.036	-.100	-.134	-.129	-.103	-.063
10	.054	-.024	-.108	-.158	-.160	-.146	-.108
12	.081	-.014	-.112	-.181	-.196	-.186	-.054
14	.112	-.003	-.115	-.200	-.230	-.229	-.067
16	.142	.006	-.120	-.218	-.261	-.270	-.079
18	.169	.015	-.128	-.238	-.291	-.296	-.091
20	.200	.023	-.137	-.261	-.317	-.327	-.097

TABLE VIII
SECTION AERODYNAMIC CHARACTERISTICS FOR REFLEX CAMBERED WING

(a) $M = 1.61$

α , deg	y/b						
	.05	.20	.35	.50	.70	.825	.95
C_n							
-20	-.972	-1.036	-.947	-.796	-.609	-.484	-.302
-18	-.862	-.932	-.882	-.738	-.563	-.442	-.274
-16	-.748	-.826	-.806	-.676	-.511	-.400	-.260
-14	-.638	-.713	-.711	-.601	-.453	-.352	-.234
-12	-.538	-.604	-.614	-.528	-.395	-.296	-.192
-10	-.436	-.498	-.495	-.449	-.329	-.244	-.168
-08	-.340	-.396	-.380	-.355	-.273	-.197	-.134
-06	-.250	-.292	-.266	-.239	-.199	-.163	-.108
-04	-.155	-.193	-.168	-.142	-.112	-.095	-.064
-02	-.060	-.091	-.007	.027	-.026	-.015	-.009
00	.036	.013	.044	.055	.064	.063	.041
02	.121	.110	.130	.145	.149	.113	.081
04	.206	.203	.231	.232	.204	.163	.116
06	.297	.298	.324	.310	.261	.211	.150
08	.384	.395	.412	.389	.318	.255	.184
10	.481	.499	.507	.468	.380	.298	.211
12	.580	.596	.603	.541	.431	.335	.235
14	.706	.729	.728	.632	.477	.377	.252
16	.844	.865	.848	.716	.497	.356	.228
18	.950	.955	.937	.765	.512	.355	.222
20	1.071	1.124	1.008	.811	.541	.378	.233
C_m							
-20	-.295	-.060	.193	.345	.444	.444	.331
-18	-.266	-.063	.178	.321	.411	.406	.301
-16	-.232	-.062	.161	.295	.375	.368	.286
-14	-.199	-.056	.138	.262	.333	.324	.259
-12	-.168	-.049	.114	.231	.289	.272	.210
-10	-.135	-.037	.085	.195	.240	.222	.185
-08	-.105	-.026	.062	.151	.199	.179	.148
-06	-.076	-.012	.044	.095	.136	.146	.118
-04	-.047	-.001	.031	.055	.073	.081	.067
-02	-.017	.009	.019	-.003	.013	.011	.009
00	.017	.025	.009	-.015	-.046	-.056	-.043
02	.044	.040	.001	-.044	-.098	-.098	-.086
04	.073	.052	-.009	-.070	-.135	-.141	-.122
06	.105	.065	-.017	-.096	-.172	-.182	-.159
08	.134	.078	-.026	-.123	-.208	-.220	-.196
10	.168	.092	-.036	-.148	-.252	-.258	-.226
12	.201	.104	-.047	-.175	-.288	-.293	-.252
14	.235	.114	-.068	-.210	-.319	-.332	-.271
16	.281	.134	-.088	-.249	-.334	-.312	-.246
18	.322	.157	-.107	-.273	-.344	-.311	-.239
20	.370	.149	-.123	-.292	-.366	-.333	-.251

TABLE VIII.- Concluded
 SECTION AERODYNAMIC CHARACTERISTICS FOR REFLEX CAMBERED WING
 (b) $M = 2.01$

α , deg	$y/b/2$						
	.05	.20	.35	.50	.70	.825	.95
C_n							
-20	-.823	-.815	-.713	-.593	-.447	-.356	-.263
-18	-.732	-.747	-.659	-.550	-.412	-.320	-.236
-16	-.650	-.675	-.601	-.499	-.375	-.295	-.210
-14	-.559	-.588	-.521	-.448	-.337	-.259	-.184
-12	-.477	-.497	-.451	-.395	-.297	-.227	-.159
-10	-.395	-.401	-.381	-.336	-.251	-.194	-.134
-08	-.311	-.316	-.291	-.269	-.204	-.156	-.108
-06	-.223	-.229	-.201	-.181	-.150	-.121	-.081
-04	-.154	-.159	-.140	-.117	-.103	-.087	-.057
-02	-.069	-.075	-.059	-.041	-.043	-.044	-.027
00	-.003	.004	.017	.031	.015	.002	.006
02	.079	.086	.100	.107	.071	.055	.046
04	.152	.162	.170	.177	.132	.101	.085
06	.226	.235	.250	.237	.192	.147	.113
08	.304	.321	.326	.311	.243	.183	.145
10	.396	.403	.387	.377	.288	.220	.168
12	.478	.485	.472	.434	.330	.247	.188
14	.565	.569	.544	.488	.370	.272	.204
16	.651	.647	.606	.539	.404	.294	.219
18	.747	.739	.687	.591	.434	.313	.231
20	.840	.832	.737	.639	.462	.327	.240
C_m							
-20	-.260	-.047	.132	.249	.326	.328	.292
-18	-.234	-.047	.119	.229	.299	.294	.262
-16	-.210	-.046	.105	.204	.269	.270	.233
-14	-.182	-.045	.087	.180	.240	.236	.204
-12	-.155	-.040	.071	.156	.209	.205	.174
-10	-.128	-.034	.055	.131	.175	.174	.147
-08	-.102	-.028	.037	.100	.139	.139	.118
-06	-.075	-.024	.022	.060	.100	.106	.087
-04	-.051	-.013	.013	.033	.066	.075	.061
-12	-.023	-.002	.003	.004	.022	.036	.028
00	.000	.007	-.006	-.022	-.018	-.005	-.009
02	.027	.019	-.014	-.048	-.056	-.052	-.052
04	.054	.027	-.023	-.071	-.094	-.089	-.092
06	.081	.039	-.029	-.088	-.131	-.128	-.121
08	.106	.049	-.036	-.112	-.164	-.157	-.155
10	.139	.062	-.040	-.133	-.192	-.189	-.180
12	.164	.072	-.049	-.151	-.220	-.212	-.202
14	.197	.082	-.056	-.170	-.248	-.235	-.219
16	.224	.091	-.062	-.186	-.271	-.254	-.235
18	.255	.100	-.075	-.206	-.291	-.271	-.249
20	.286	.102	-.080	-.224	-.311	-.284	-.259

TABLE IX
SECTION AERODYNAMIC CHARACTERISTICS FOR FLAT WING
[From reference 4]

(a) $M = 1.61$

α , deg	$y/b/2$					
	.05	.20	.35	.50	.70	.90
C_n						
-20	-1.033	-1.087	-1.019	-0.888	-0.652	-0.346
-18	-.889	-.935	-.903	-.791	-.611	-.388
-16	-.770	-.802	-.794	-.686	-.532	-.375
-14	-.653	-.678	-.684	-.611	-.463	-.323
-12	-.554	-.581	-.588	-.533	-.411	-.274
-10	-.465	-.489	-.494	-.468	-.365	-.238
-08	-.367	-.388	-.392	-.376	-.309	-.202
-06	-.272	-.286	-.295	-.283	-.241	-.163
-04	-.166	-.179	-.187	-.179	-.151	-.113
-02	-.084	-.094	-.096	-.091	-.077	-.054
00	.000	.000	.000	.000	.000	.000
02	.084	.094	.096	.091	.077	.054
04	.166	.179	.187	.179	.151	.113
06	.272	.286	.295	.283	.241	.163
08	.367	.388	.392	.376	.309	.202
10	.465	.489	.494	.468	.365	.238
12	.554	.581	.588	.533	.411	.274
14	.653	.678	.684	.611	.463	.323
16	.770	.802	.794	.686	.532	.375
18	.889	.935	.903	.791	.611	.388
20	1.033	1.087	1.019	0.888	0.652	0.346
C_m						
-20	-.319	-.079	.174	.356	.458	-.354
-18	-.284	-.082	.143	.313	.432	-.394
-16	-.254	-.086	.115	.262	.375	-.382
-14	-.220	-.085	.085	.230	.323	-.330
-12	-.187	-.076	.066	.194	.284	-.278
-10	-.158	-.066	.050	.169	.253	-.241
-08	-.125	-.052	.037	.130	.213	-.204
-06	-.094	-.039	.027	.095	.164	-.164
-04	-.059	-.025	.015	.058	.100	-.113
-02	-.031	-.013	.009	.029	.051	-.054
00	.000	.000	.000	.000	.000	.000
02	.031	.013	-.009	-.029	-.051	.054
04	.059	.025	-.015	-.058	-.100	.113
06	.094	.039	-.027	-.095	-.164	.164
08	.125	.052	-.037	-.130	-.213	.204
10	.158	.066	-.050	-.169	-.253	.241
12	.187	.076	-.066	-.194	-.284	.278
14	.220	.085	-.085	-.230	-.323	.330
16	.254	.086	-.115	-.262	-.375	.382
18	.284	.082	-.143	-.313	-.432	.394
20	.319	.079	-.174	-.356	-.458	.354

TABLE IX.- Concluded
 SECTION AERODYNAMIC CHARACTERISTICS FOR FLAT WING
 [From reference 4]

(b) M = 2.01

L-3000

α , deg	$y/b/2$					
	.05	.20	.35	.50	.70	.90
C_n						
-20	-0.830	-0.814	-0.732	-0.615	-0.463	-0.301
-18	-0.741	-0.742	-0.668	-0.564	-0.427	-0.279
-16	-0.648	-0.661	-0.604	-0.519	-0.392	-0.254
-14	-0.558	-0.567	-0.538	-0.464	-0.349	-0.226
-12	-0.475	-0.485	-0.464	-0.406	-0.309	-0.200
-10	-0.394	-0.407	-0.396	-0.351	-0.265	-0.174
-08	-0.310	-0.321	-0.312	-0.282	-0.216	-0.143
-06	-0.232	-0.240	-0.232	-0.213	-0.166	-0.109
-04	-0.151	-0.159	-0.156	-0.143	-0.114	-0.076
-02	-0.074	-0.078	-0.073	-0.069	-0.056	-0.038
00	.000	.000	.000	.000	.000	.000
02	.074	.078	.073	.069	.056	.038
04	.151	.159	.156	.143	.114	.076
06	.232	.240	.232	.213	.166	.109
08	.310	.321	.312	.282	.216	.143
10	.394	.407	.396	.351	.265	.174
12	.475	.485	.464	.406	.309	.200
14	.558	.567	.538	.464	.349	.226
16	.648	.661	.604	.519	.392	.254
18	.741	.742	.668	.564	.427	.279
20	.830	.814	.732	.615	.463	.301
C_m						
-20	-0.276	-0.075	.101	.231	.322	-0.306
-18	-0.248	-0.073	.090	.210	.297	-0.283
-16	-0.217	-0.068	.083	.196	.272	-0.257
-14	-0.189	-0.064	.072	.174	.241	-0.228
-12	-0.162	-0.058	.059	.151	.213	-0.202
-10	-0.135	-0.050	.048	.130	.182	-0.175
-08	-0.108	-0.041	.035	.103	.147	-0.144
-06	-0.081	-0.031	.026	.076	.112	-0.110
-04	-0.054	-0.020	.016	.050	.077	-0.076
-02	-0.028	-0.010	.007	.024	.038	-0.038
00	.000	.000	.000	.000	.000	.000
02	.028	.010	-.007	-.024	-.038	.038
04	.054	.020	-.016	-.050	-.077	.076
06	.081	.031	-.026	-.076	-.112	.110
08	.108	.041	-.035	-.103	-.147	.144
10	.135	.050	-.048	-.130	-.182	.175
12	.162	.058	-.059	-.151	-.213	.202
14	.189	.064	-.072	-.174	-.241	.228
16	.217	.068	-.083	-.196	-.272	.257
18	.248	.073	-.090	-.210	-.297	.283
20	.276	.075	-.101	-.231	-.322	.306

TABLE X
SECTION AERODYNAMIC CHARACTERISTICS FOR CAMBERED WING

(a) $M = 1.61$

α , deg	y/b					
	.05	.20	.35	.50	.70	.90
C_n						
-20	-.996	-1.036	-.934	-.783	-.576	-.323
-18	-.877	-.900	-.838	-.682	-.505	-.283
-16	-.762	-.767	-.752	-.600	-.421	-.232
-14						
-12	-.557	-.543	-.586	-.457	-.310	-.150
-10	-.446	-.430	-.483	-.382	-.256	-.129
-08	-.350	-.329	-.361	-.317	-.206	-.115
-06	-.252	-.228	-.249	-.231	-.150	-.087
-04	-.159	-.128	-.149	-.144	-.094	-.055
-02	-.068	-.033	-.046	-.054	-.034	-.021
00	.029	.071	.061	.055	.040	.023
02	.116	.166	.159	.152	.118	.074
04	.214	.271	.266	.256	.200	.139
06	.308	.363	.360	.341	.271	.189
08	.421	.468	.464	.439	.345	.238
10	.518	.565	.558	.505	.402	.276
12	.622	.666	.648	.579	.461	.326
14	.726	.766	.739	.655	.518	.384
16	.841	.883	.842	.743	.607	.424
18	.965	1.009	.955	.859	.674	.420
20	1.125	1.175	1.105	.967	.692	.431
C_m						
-20	-.394	-.106	.148	.313	.407	.328
-18	-.370	-.121	.124	.265	.357	.288
-16	-.341	-.135	.103	.226	.291	.236
-14						
-12	-.276	-.139	.059	.157	.205	.146
-10	-.240	-.135	.030	.124	.165	.125
-08	-.211	-.127	-.007	.098	.129	.111
-06	-.178	-.118	-.034	.060	.089	.081
-04	-.143	-.106	-.053	.022	.048	.049
-02	-.114	-.095	-.069	-.018	.006	.015
00	-.084	-.082	-.084	-.066	-.049	-.032
02	-.057	-.070	-.096	-.106	-.106	-.084
04	-.025	-.056	-.107	-.145	-.164	-.151
06	.005	-.043	-.116	-.174	-.212	-.200
08	.042	-.030	-.125	-.208	-.260	-.248
10	.078	-.017	-.133	-.223	-.297	-.286
12	.107	-.002	-.138	-.247	-.336	-.336
14	.143	.010	-.151	-.276	-.372	-.397
16	.176	.017	-.170	-.310	-.438	-.433
18	.206	.019	-.192	-.359	-.481	-.428
20	.243	.018	-.218	-.394	-.489	-.413

TABLE X.- Concluded

SECTION AERODYNAMIC CHARACTERISTICS FOR CAMBERED WING

(b) M = 2.01

α , deg	y/b					
	.05	.20	.35	.50	.70	.90
	C_n					
-20	-.795	-.774	-.674	-.537	-.390	-.234
-18	-.736	-.705	-.618	-.495	-.360	-.215
-16	-.648	-.624	-.566	-.452	-.330	.000
-14	-.552	-.533	-.501	-.398	-.294	-.178
-12	-.469	-.445	-.434	-.347	-.257	-.166
-10	-.376	-.365	-.366	-.294	-.222	-.151
-08	-.298	-.281	-.301	-.238	-.188	-.129
-06	-.222	-.207	-.219	-.178	-.147	-.103
-04	-.143	-.123	-.136	-.109	-.101	-.074
-02	-.073	-.044	-.056	-.041	-.053	-.045
00	.009	.034	.025	.030	-.001	-.012
02	.090	.115	.104	.102	.056	.022
04	.173	.198	.187	.173	.117	.071
06	.250	.270	.257	.238	.172	.113
08	.337	.355	.337	.308	.231	.158
10	.421	.435	.415	.374	.288	.199
12	.521	.525	.494	.447	.343	.236
14	.611	.612	.561	.505	.391	.267
16	.697	.701	.636	.565	.435	.296
18	.803	.794	.714	.629	.481	.322
20	.895	.872	.784	.685	.515	.342
	C_m					
-20	-.339	-.096	.079	.190	.266	.235
-18	-.316	-.100	.066	.172	.243	.214
-16	-.274	-.108	.055	.153	.221	.000
-14	-.258	-.112	.041	.130	.193	.176
-12	-.232	-.111	.027	.109	.166	.163
-10	-.203	-.107	.013	.086	.141	.148
-08	-.177	-.102	-.002	.061	.116	.125
-06	-.150	-.095	-.019	.035	.086	.097
-04	-.123	-.086	-.037	.005	.052	.068
-02	-.100	-.078	-.052	-.025	.016	.037
00	-.073	-.069	-.066	-.056	-.022	.003
02	-.044	-.058	-.077	-.085	-.065	-.033
04	-.018	-.049	-.088	-.112	-.108	-.083
06	.007	-.040	-.095	-.134	-.144	-.124
08	.036	-.032	-.106	-.158	-.181	-.167
10	.063	-.023	-.114	-.179	-.218	-.207
12	.101	-.013	-.119	-.200	-.253	-.244
14	.123	-.003	-.121	-.216	-.284	-.275
16	.150	.008	-.128	-.237	-.312	-.304
18	.183	.017	-.137	-.259	-.343	-.330
20	.211	.024	-.147	-.279	-.366	-.351

TABLE XI
SECTION AERODYNAMIC CHARACTERISTICS FOR LINEAR TWIST WING
[From reference 4]

(a) $M = 1.61$

α , deg	y/b						
	.05	.20	.35	.50	.70	.825	.95
	C_n						
-20	-1.106	-1.165	-1.071	-0.906	-0.641	-0.454	-0.288
-18	-0.949	-0.989	-0.956	-0.827	-0.601	-0.453	-0.285
-16	-0.823	-0.852	-0.859	-0.756	-0.557	-0.430	-0.289
-14	-0.707	-0.730	-0.764	-0.678	-0.506	-0.420	-0.298
-12	-0.605	-0.632	-0.653	-0.610	-0.471	-0.387	-0.282
-10	-0.502	-0.534	-0.556		-0.428	-0.359	-0.267
-08	-0.405	-0.431	-0.458	-0.451	-0.392	-0.329	-0.245
-06	-0.313	-0.334	-0.364	-0.361	-0.339	-0.291	-0.220
-04	-0.212	-0.231	-0.261	-0.263	-0.257	-0.241	-0.186
-02	-0.116	-0.130	-0.160	-0.166	-0.174	-0.180	-0.150
00	-0.024	-0.038	-0.066	-0.075	-0.097	-0.111	-0.104
02	.049	.049	.017	.006	-0.024	-0.051	-0.057
04	.142	.147	.114	.102	.051	.018	-0.006
06	.236	.242	.217	.195	.132	.087	.035
08	.326	.332	.307	.285	.209	.143	.073
10	.428	.441	.410	.384	.277	.197	.113
12	.522	.541	.507	.467	.333	.245	.148
14	.625	.643	.617	.547	.398	.298	.195
16	.712	.729	.707	.623	.458	.370	.254
18	.844	.879	.822	.722	.545	.428	.259
20	.966	.969	.939	.827	.607	.467	.297
	C_m						
-20	-0.323	-0.079	.189	.368	.454	.410	.316
-18	-0.288	-0.084	.154	.336	.430	.409	.312
-16	-0.263	-0.089	.128	.298	.401	.392	.317
-14	-0.228	-0.090	.102	.255	.355	.380	.327
-12	-0.195	-0.079	.073	.224	.327	.346	.308
-10	-0.161	-0.068	.058		.296	.321	.291
-08	-0.129	-0.054	.044	.156	.272	.294	.267
-06	-0.100	-0.041	.034	.121	.234	.260	.238
-04	-0.067	-0.027	.024	.085	.172	.214	.201
-02	-0.034	-0.015	.015	.053	.115	.158	.161
00	-0.004	-0.001	.005	.023	.064	.097	.110
02	.023	.010	-0.001	-0.004	.015	.045	.060
04	.055	.024	-0.010	-0.037	-0.033	-0.014	.007
06	.086	.038	-0.020	-0.067	-0.087	-0.075	-0.038
08	.117	.050	-0.028	-0.099	-0.143	-0.125	-0.078
10	.152	.065	-0.039	-0.138	-0.190	-0.173	-0.121
12	.183	.076	-0.053	-0.170	-0.229	-0.216	-0.160
14	.215	.085	-0.074	-0.205	-0.277	-0.265	-0.213
16	.241	.090	-0.097	-0.240	-0.321	-0.336	-0.277
18	.273	.079	-0.127	-0.284	-0.385	-0.382	-0.278
20	.302	.072	-0.156	-0.333	-0.427	-0.416	-0.323

TABLE XI.- Concluded

SECTION AERODYNAMIC CHARACTERISTICS FOR LINEAR TWIST WING

[From reference 4]

(b) $M = 2.01$

α , deg	$y/b/2$						
	.05	.20	.35	.50	.70	.825	.95
	C_n						
-20	-.869	-.879	-.792	-.676	-.559	-.391	-.282
-18	-.771	-.792	-.745	-.628	-.466	-.373	-.269
-16	-.675	-.686	-.671	-.575	-.437	-.356	-.261
-14	-.587	-.602	-.605	-.527	-.408	-.335	-.248
-12	-.500	-.518	-.538	-.476	-.374	-.312	-.234
-10	-.410	-.431	-.450	-.416	-.334	-.279	-.192
-08	-.331	-.355	-.375	-.354	-.294	-.249	-.168
-06	-.241	-.268	-.290	-.278	-.244	-.214	-.145
-04	-.172	-.190	-.222	-.212	-.197	-.179	-.118
-02	-.095	-.116	-.139	-.140	-.143	-.139	-.086
00	-.023	-.035	-.063	-.072	-.086	-.095	-.051
02	.048	.042	.012	.010	-.026	-.046	-.015
04	.125	.124	.092	.074	.032	.004	
06	.196	.199	.173	.144	.084	.046	.014
08	.274	.278	.249	.215	.139	.091	.045
10	.351	.352	.324	.278	.186	.132	.074
12	.438	.440	.402	.346	.239	.174	.107
14	.524	.517	.479	.405	.283	.210	.133
16	.601	.594	.544	.459	.327	.243	.157
18	.688	.677	.609	.514	.370	.277	.184
20	.778	.762	.671	.561	.405	.305	.204
	C_m						
-20	-.284	-.079	.114	.258	.389	.353	.311
-18	-.253	-.074	.105	.236	.328	.336	.295
-16	-.222	-.076	.092	.213	.307	.320	.287
-14	-.193	-.069	.082	.195	.285	.301	.271
-12	-.165	-.062	.071	.176	.261	.280	.256
-10	-.136	-.053	.053	.153	.231	.249	
-08	-.109	-.044	.042	.128	.202	.222	.208
-06	-.078	-.032	.031	.098	.167	.190	.182
-04	-.055	-.023	.023	.072	.135	.158	.157
-02	-.029	-.012	.013	.046	.097	.123	.128
00	-.005	-.003	.004	.022	.058	.084	.093
02	.021	.007	-.003	-.010	.017	.041	.055
04	.048	.020	-.011	-.027	-.022	-.003	.016
06	.073	.029	-.021	-.053	-.057	-.040	-.015
08	.100	.039	-.029	-.080	-.094	-.079	-.048
10	.127	.049	-.039	-.103	-.126	-.115	-.079
12	.155	.059	-.050	-.129	-.163	-.152	-.115
14	.185	.066	-.063	-.151	-.193	-.184	-.144
16	.209	.070	-.074	-.172	-.225	-.214	-.170
18	.236	.077	-.082	-.192	-.255	-.245	-.200
20	.265	.078	-.092	-.209	-.281	-.271	-.221

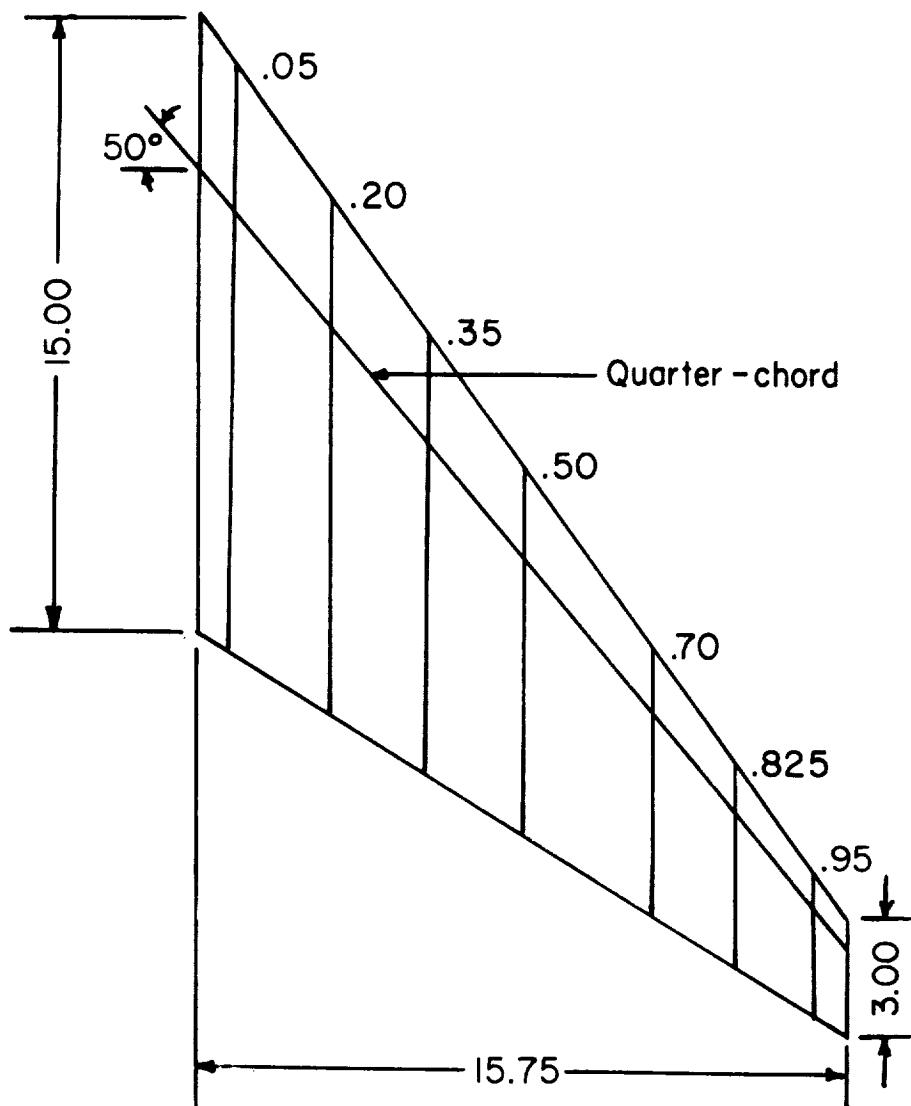


Figure 1.- Plan view of wings showing orifice stations. (Lengths are given in inches; stations are given in fractions of semispan.)



